



Urban Air Mobility Vehicle Architecture Trends

ICAS

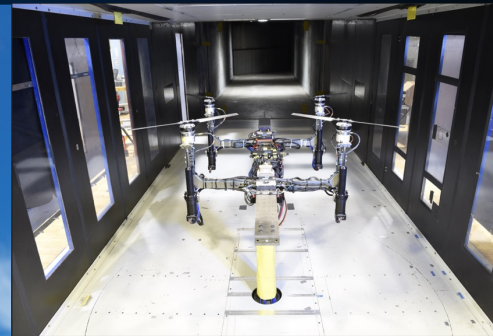
EMERGING TECHNOLOGY FORUM 2023

FUTURE AIR MOBILITY

– BARRIERS AND ENABLERS –

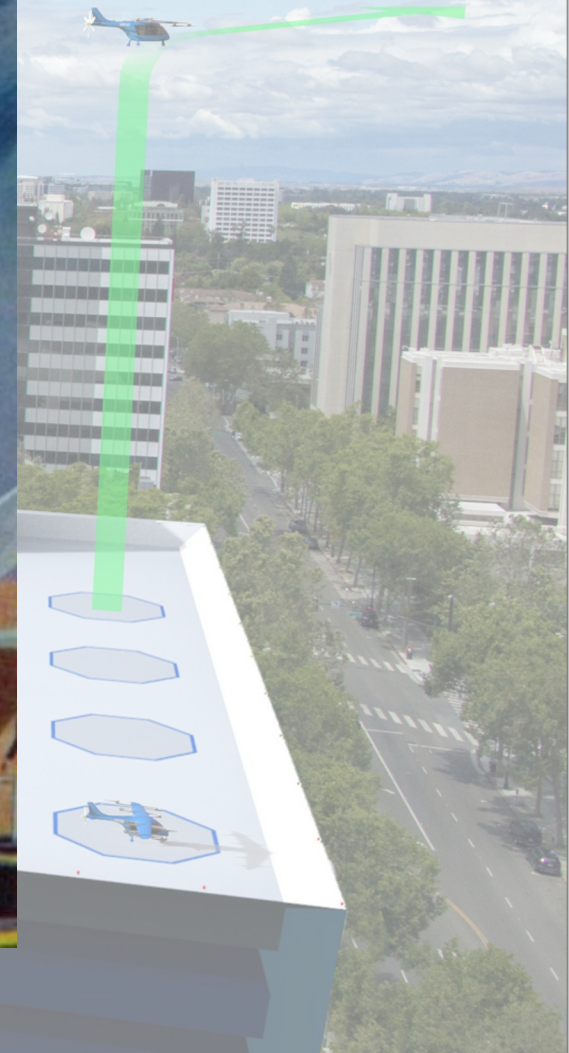
NASA Revolutionary Vertical Lift Technology Project

Presented by: Chris Silva



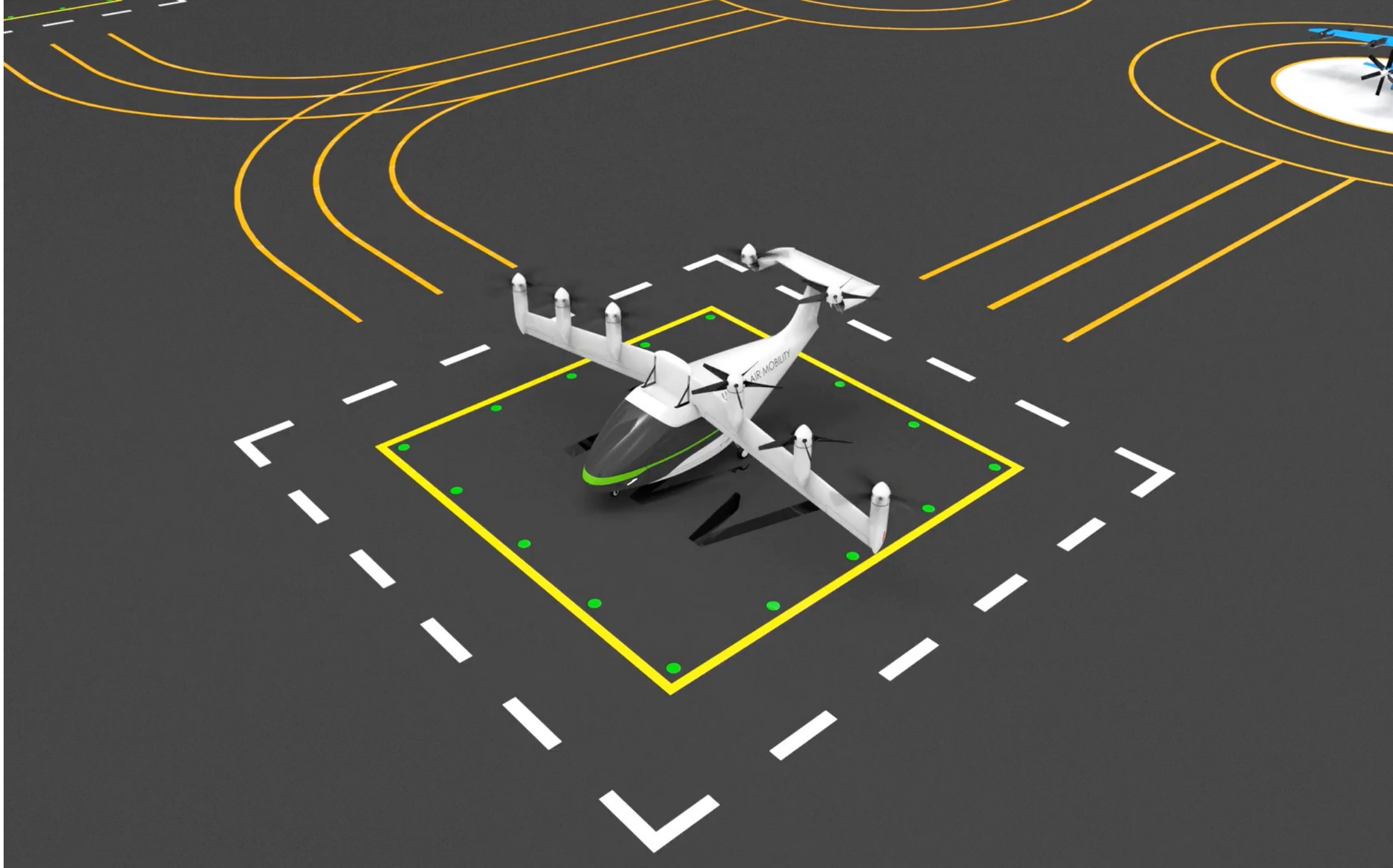


What is different? Why haven't we been doing this with helicopters?

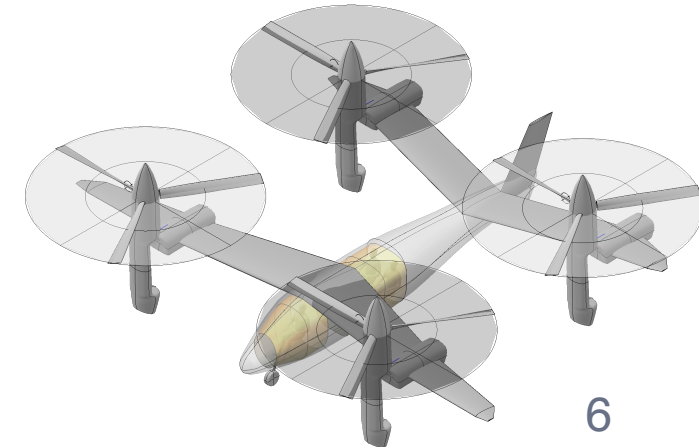
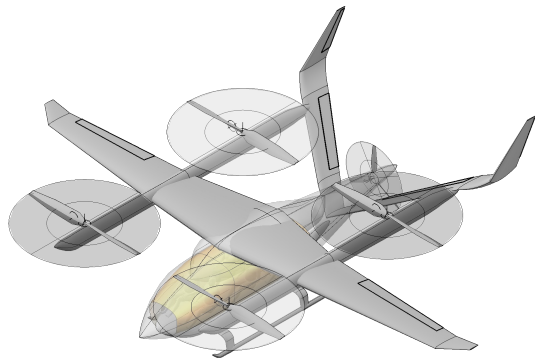
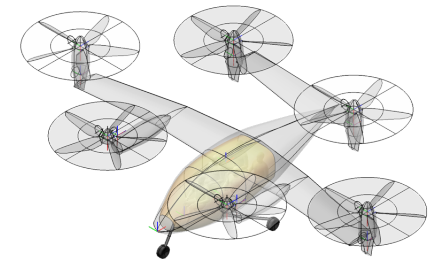
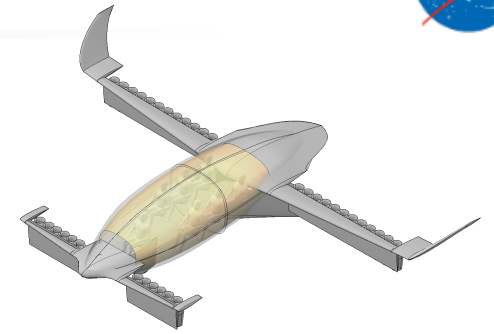
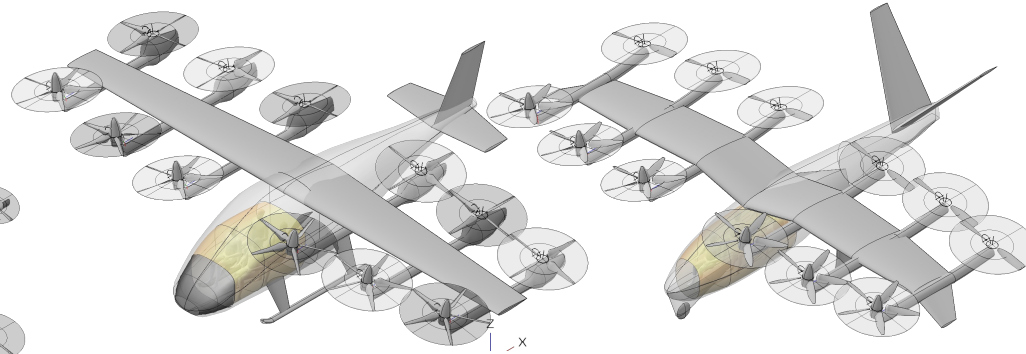
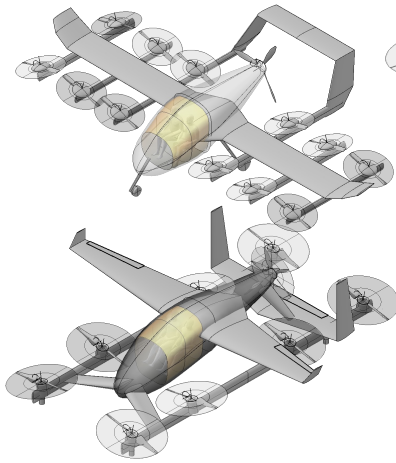
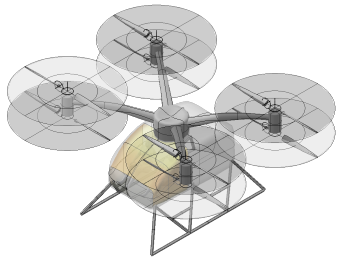
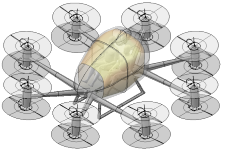
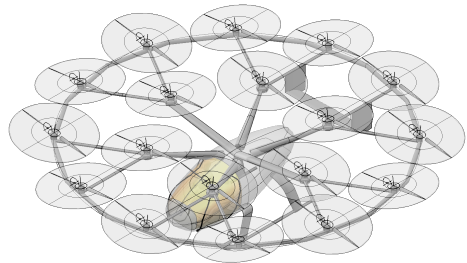


What is different? Why haven't we been doing this with helicopters?

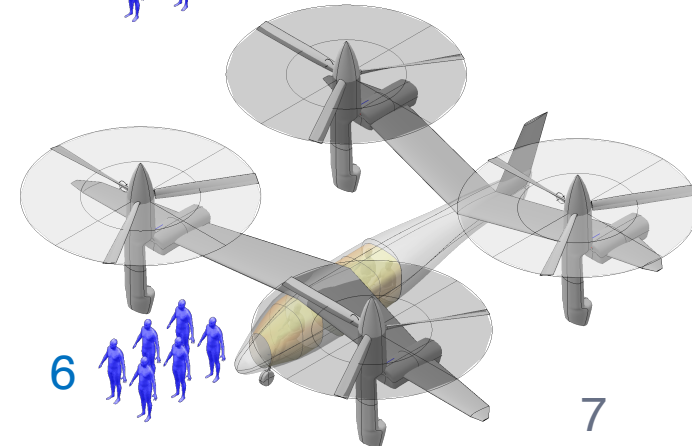
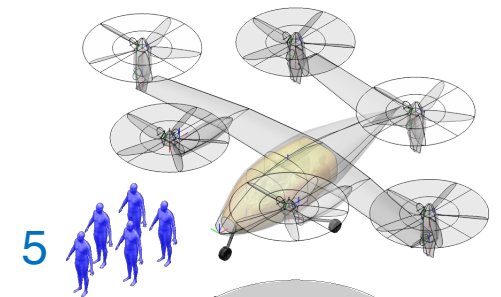
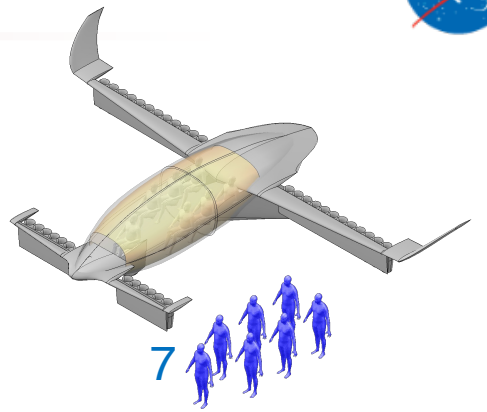
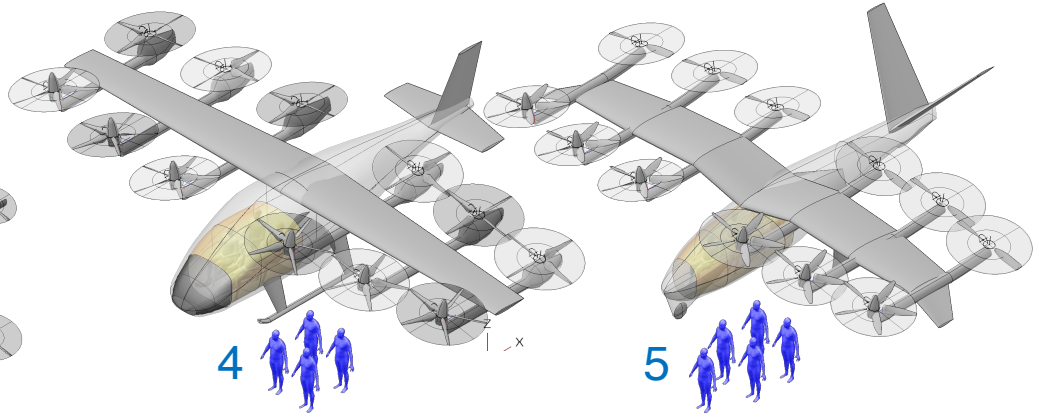
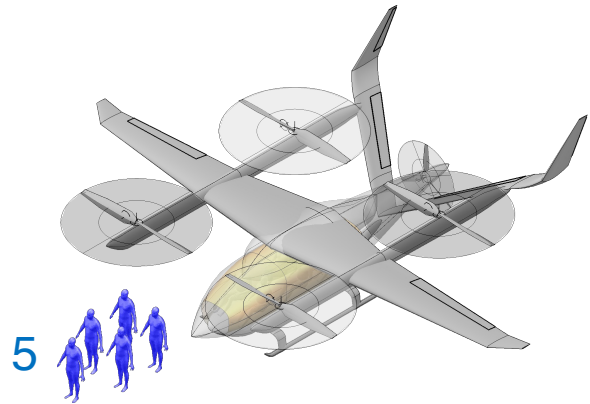
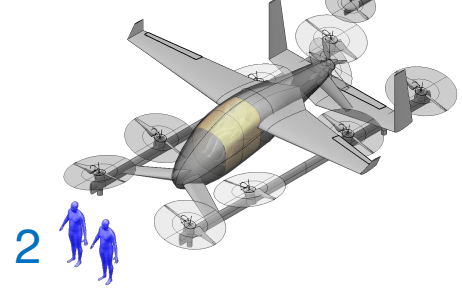
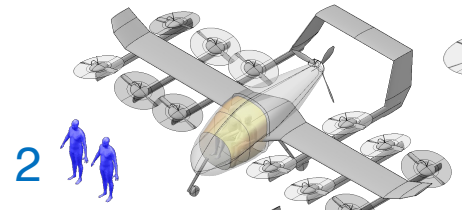
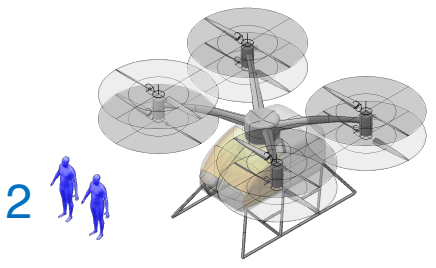
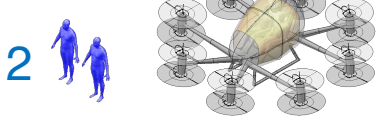
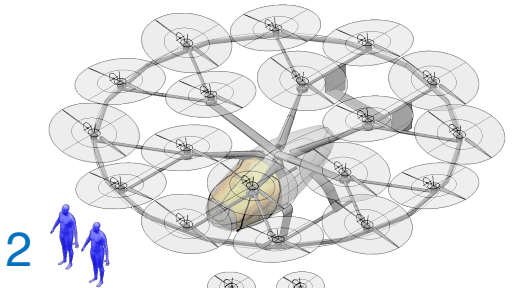
- U.S.A. had large-scale commuter helicopter service 40-60 years ago, and Canada has it today
- Helicopters have been **too loud**, **too expensive**, and **too dangerous** for everyday travel
- Alternative (electric) propulsion may drastically cut energy cost and reduce emissions
- Ridesharing taxi service may solve the first/last mile travel inconvenience
- Autonomy may drastically reduce flight crew cost
- Quiet rotorcraft technology now exists
- We have better predictive tools



UAM Requirement: Convenience

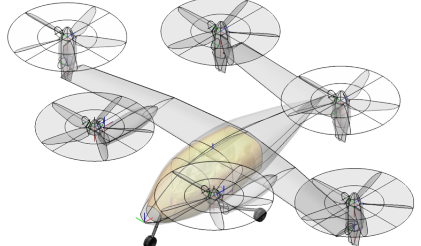
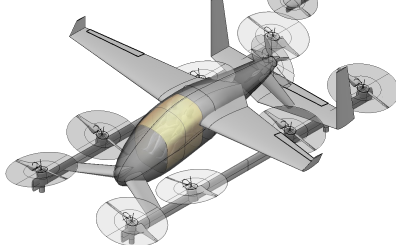
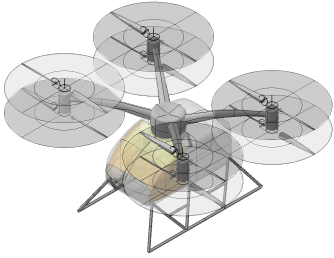
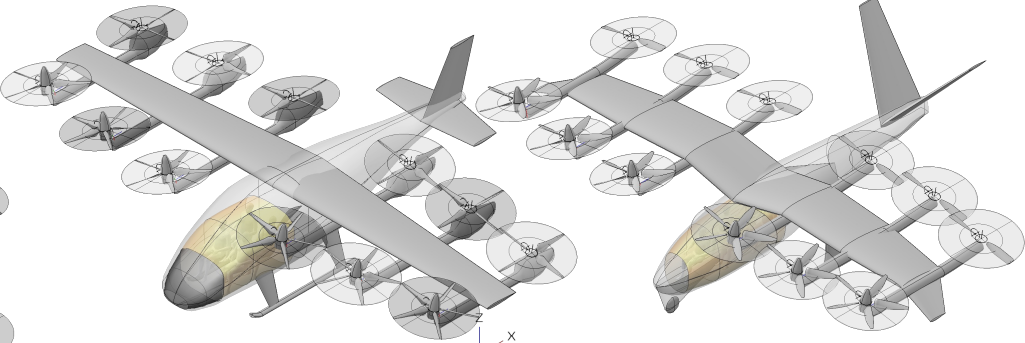
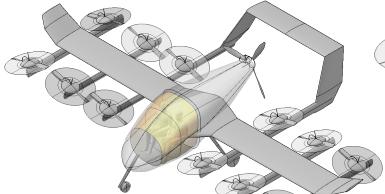
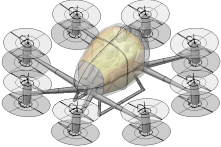
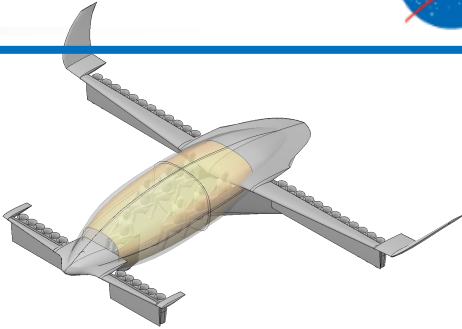
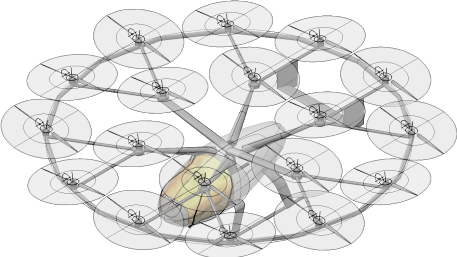


UAM Requirement: Convenience

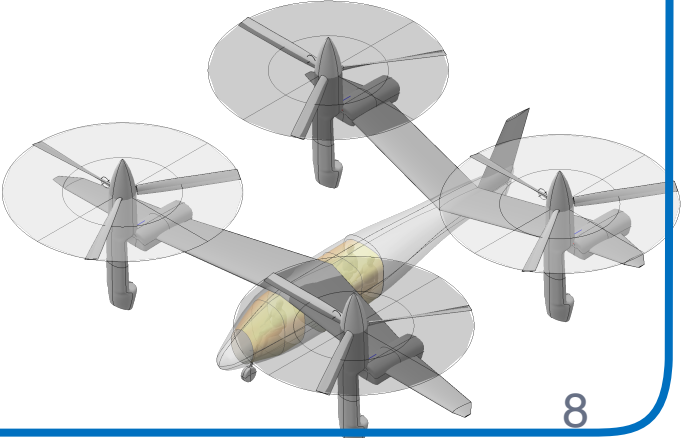
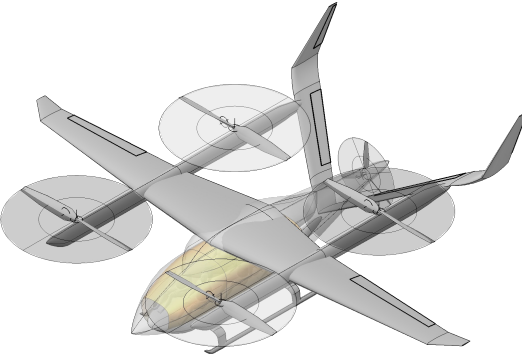


Between 2 and 7 occupants

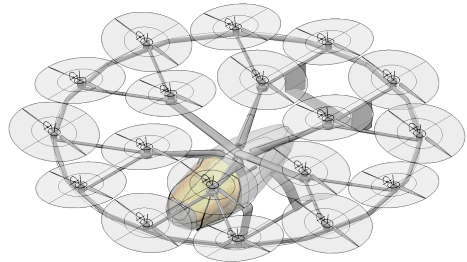
UAM Requirement: Convenience



Vertical Takeoff and Landing

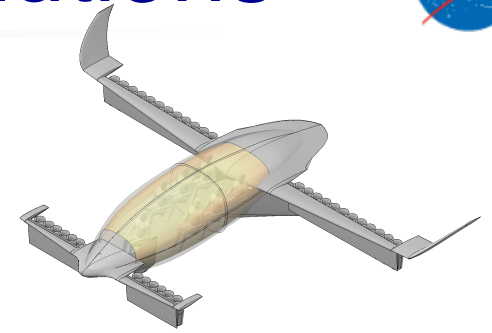


We observe very diverse VTOL vehicle solutions



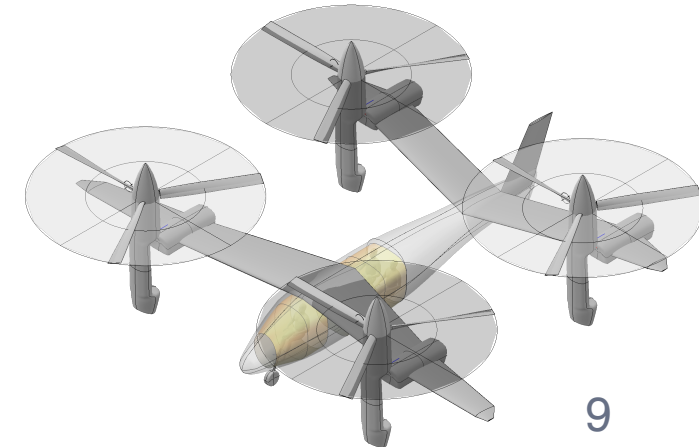
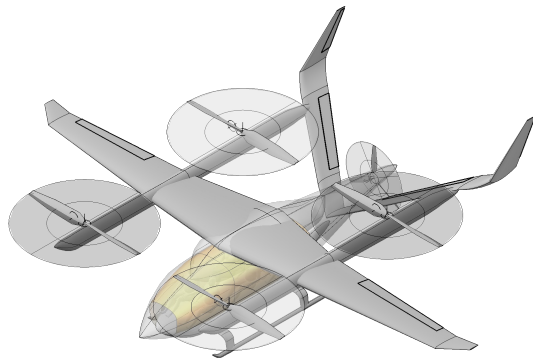
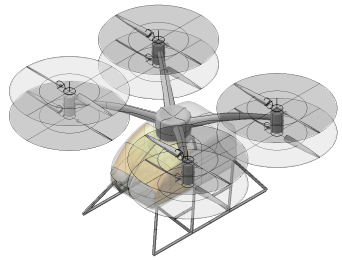
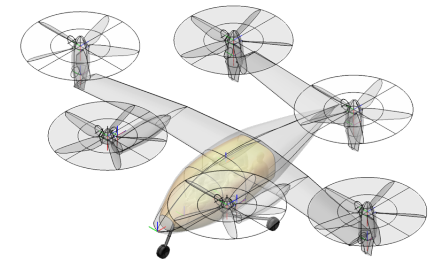
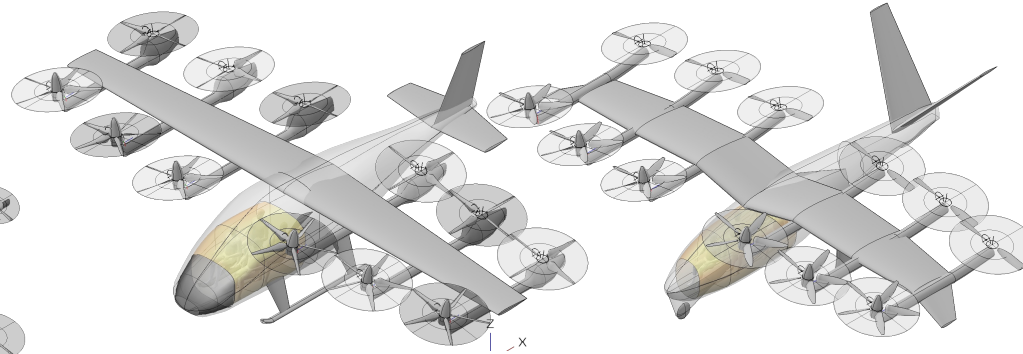
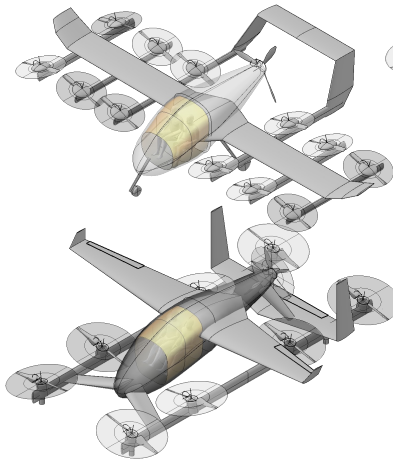
Stopping rotor
Lift+cruise

Stop some rotors,
tilt some rotors

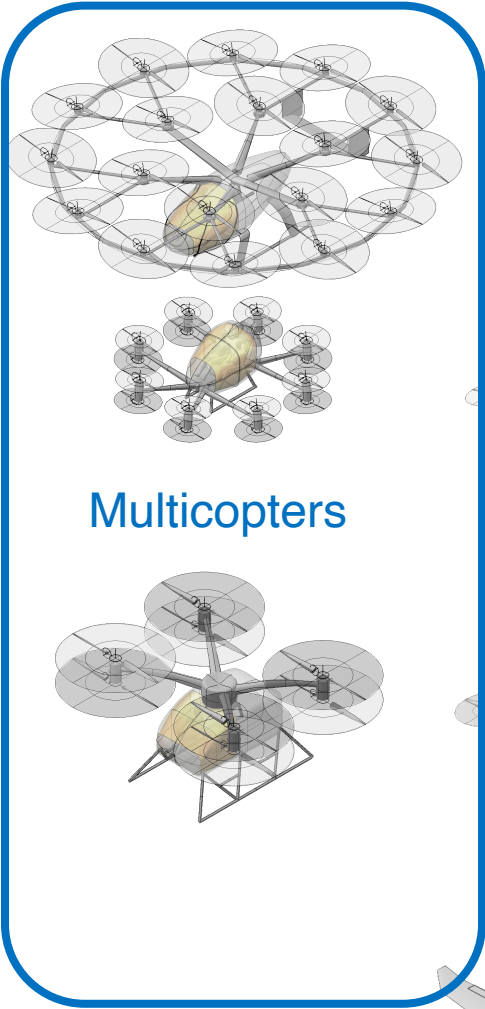


Tiltrotors and tiltwings

Multicopters



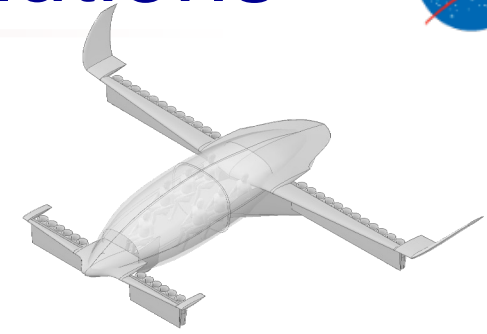
We observe very diverse VTOL vehicle solutions



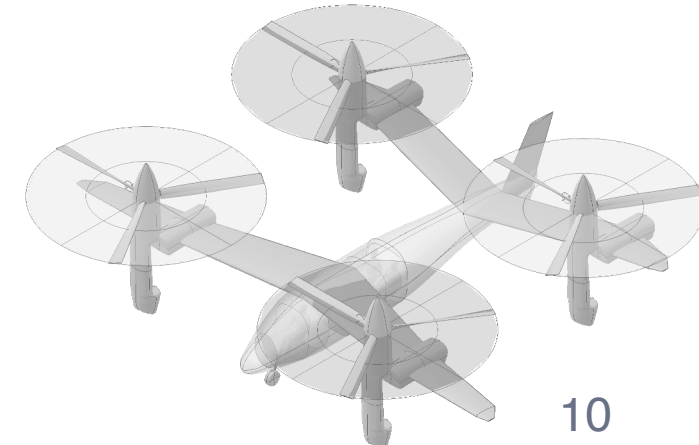
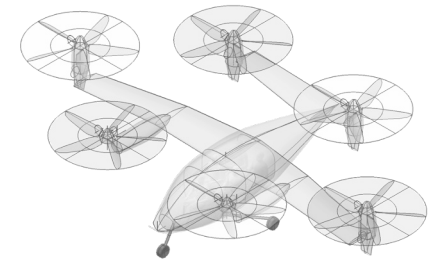
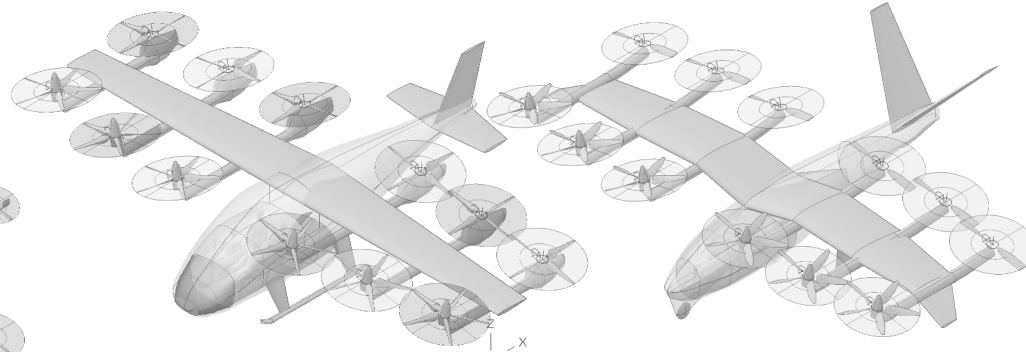
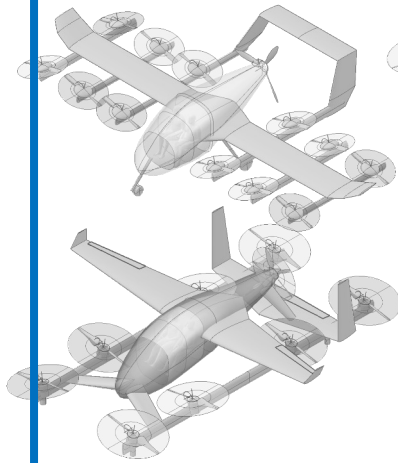
Multicopters

Stopping rotor
Lift+cruise

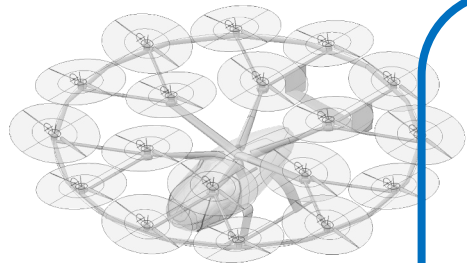
Stop some rotors,
tilt some rotors



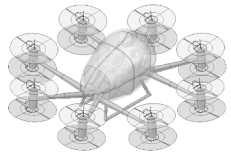
Tiltrotors and tiltwings



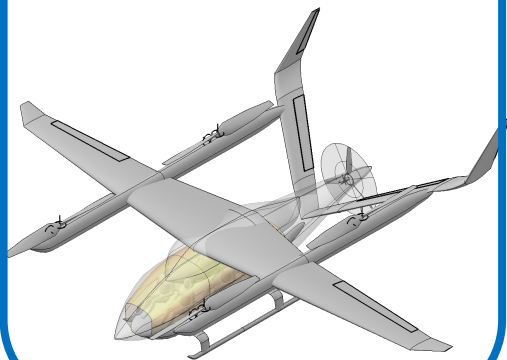
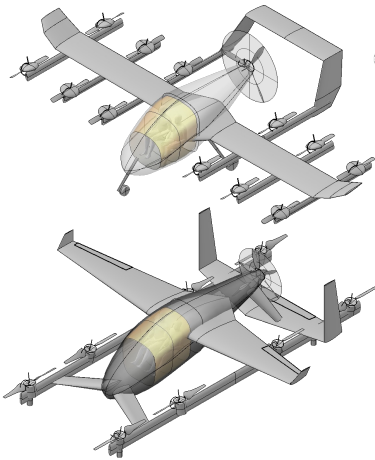
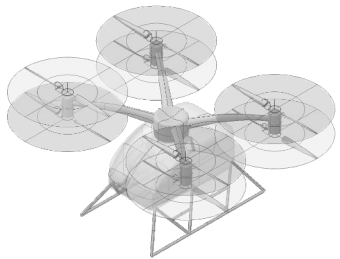
We observe very diverse VTOL vehicle solutions



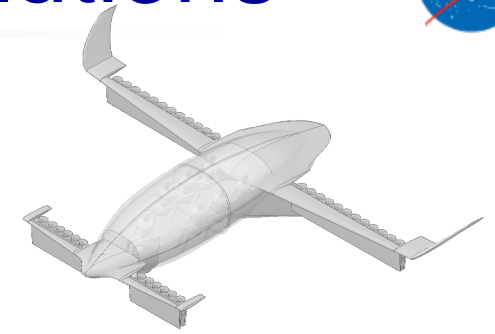
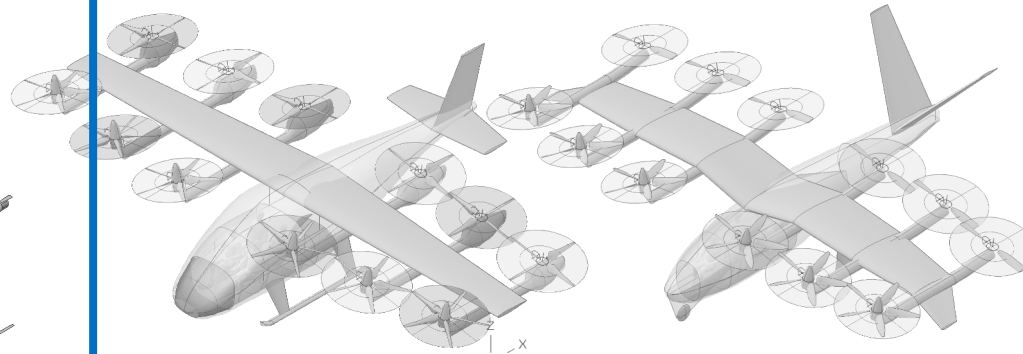
Stopping rotor
Lift+cruise



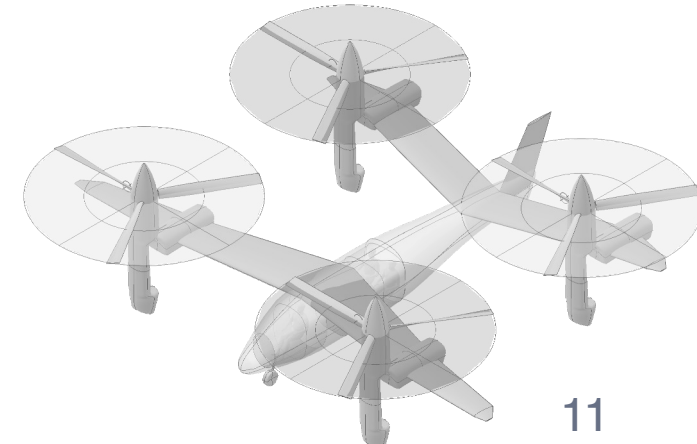
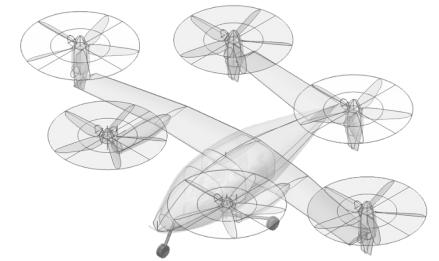
Multicopters



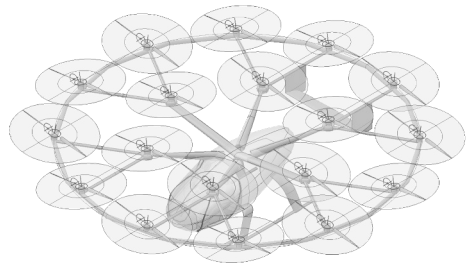
Stop some rotors,
tilt some rotors



Tiltrotors and tiltwings

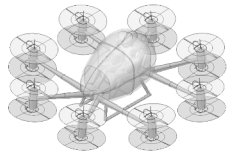


We observe very diverse VTOL vehicle solutions

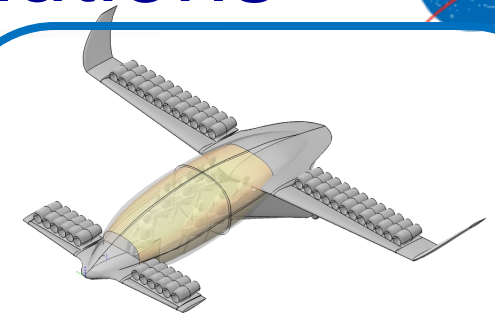
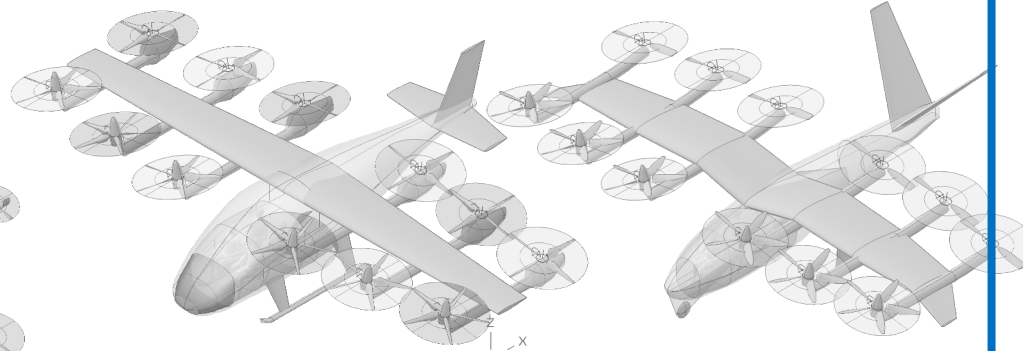
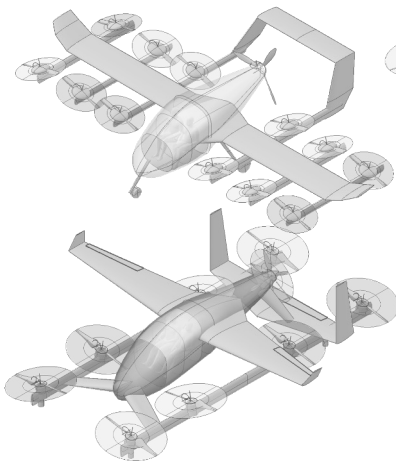
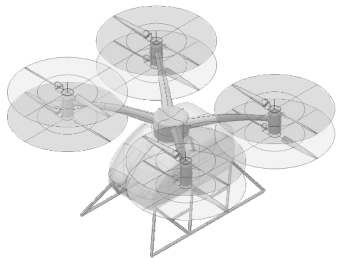


Stopping rotor
Lift+cruise

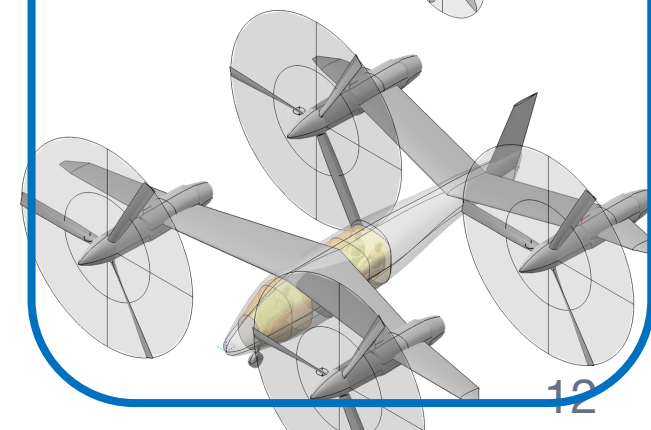
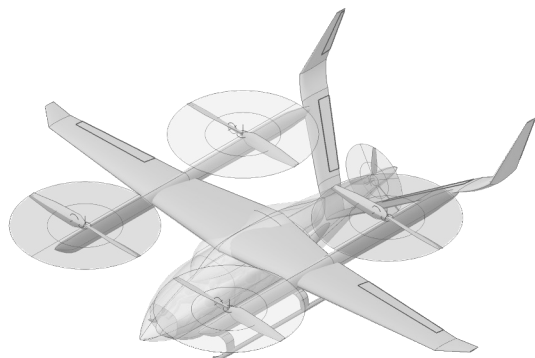
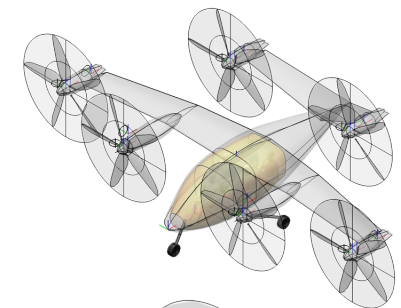
Stop some rotors,
tilt some rotors



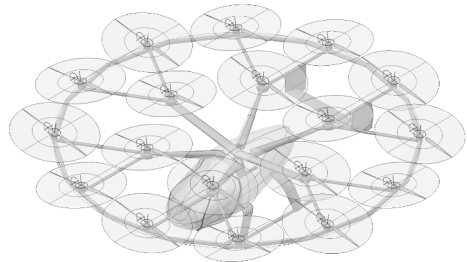
Multicopters



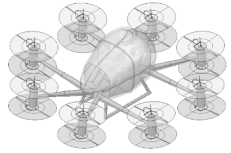
Tiltrotors and tiltwings



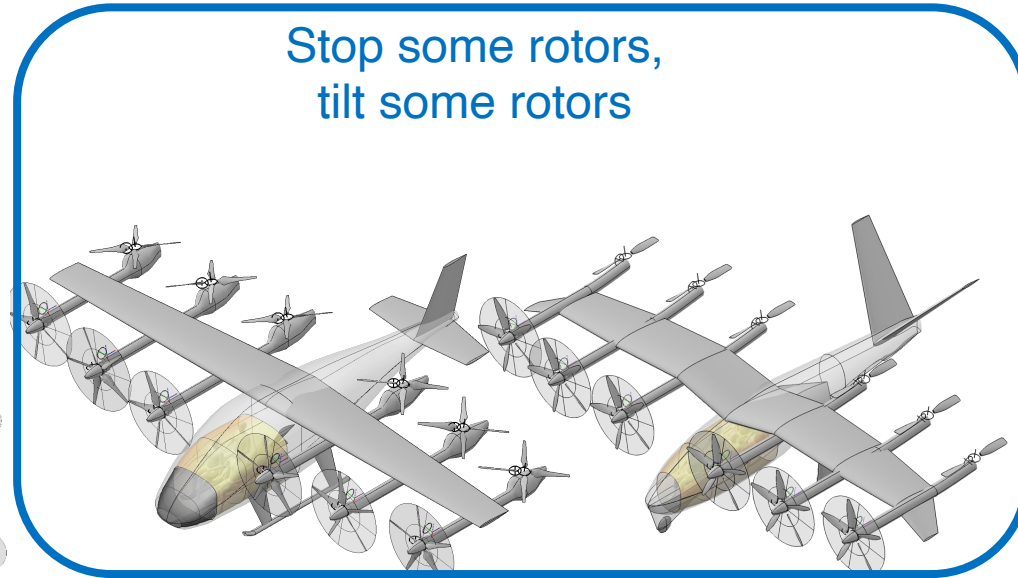
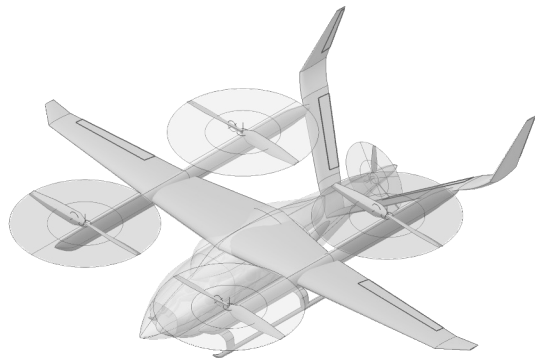
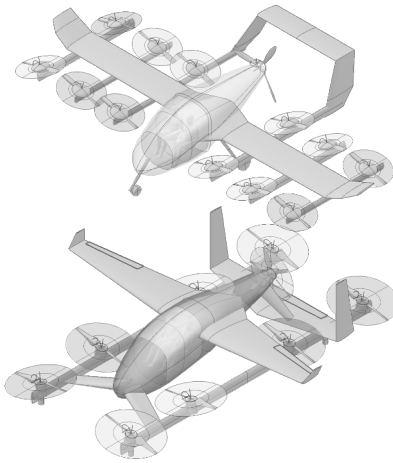
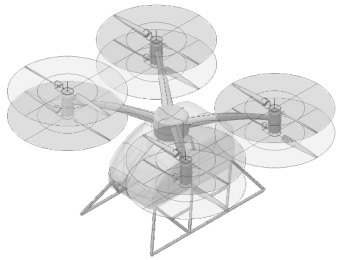
We observe very diverse VTOL vehicle solutions



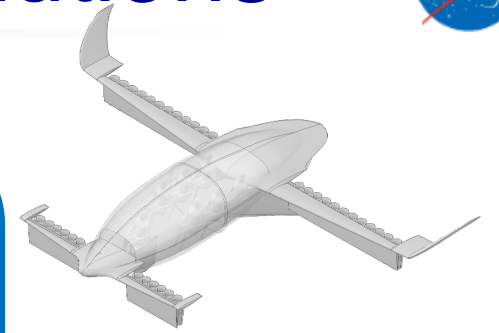
Stopping rotor
Lift+cruise



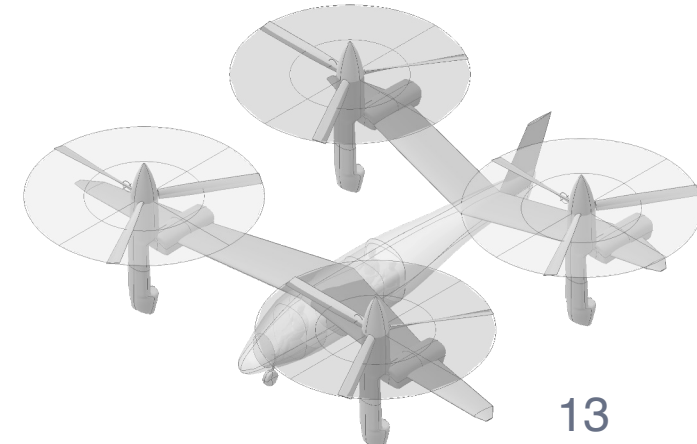
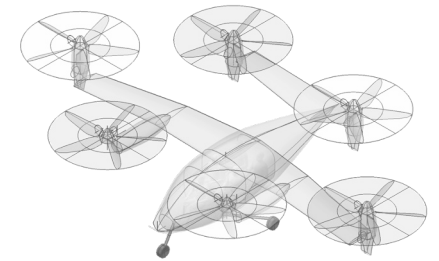
Multicopters



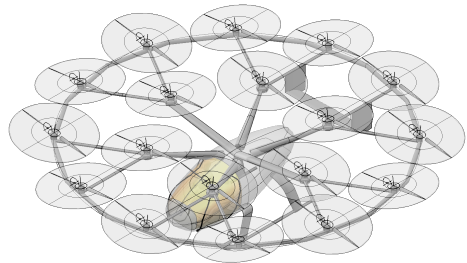
Stop some rotors,
tilt some rotors



Tiltrotors and tiltwings

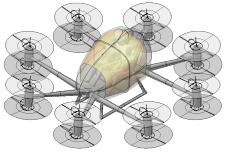
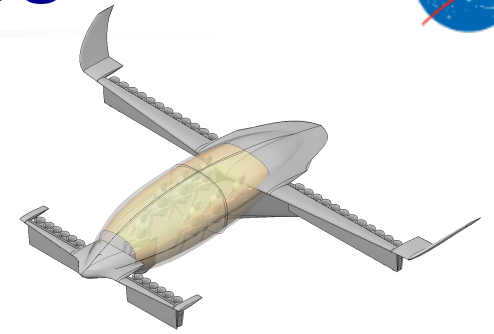


The arrangement of rotors indicate priorities

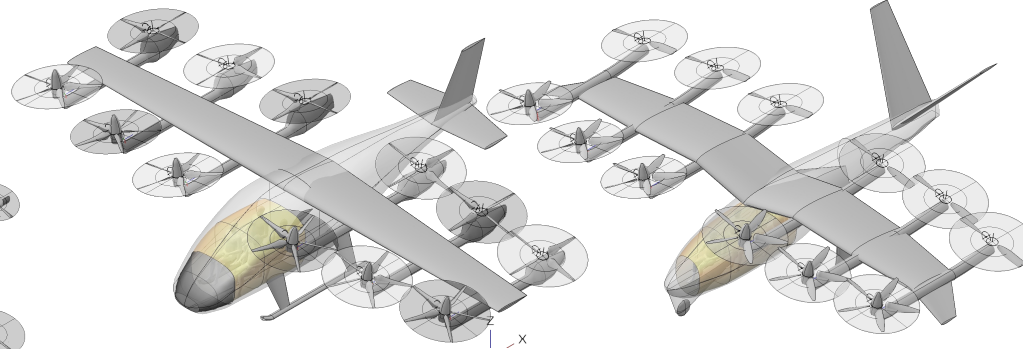
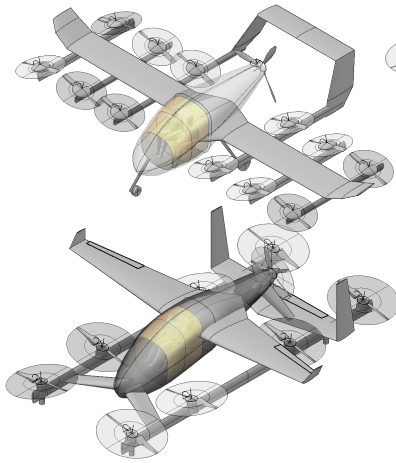


Stopping rotor
Lift+cruise

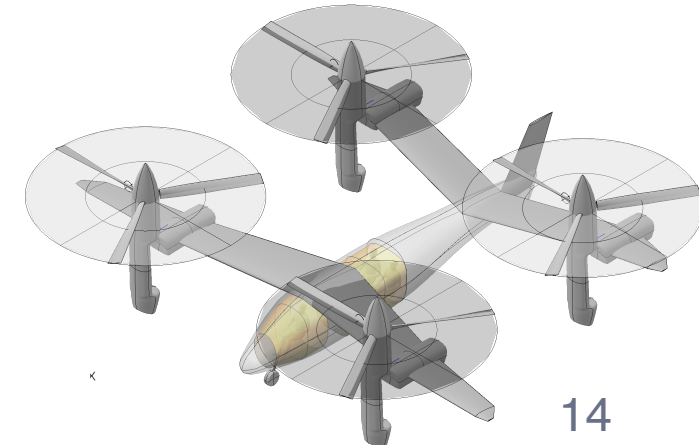
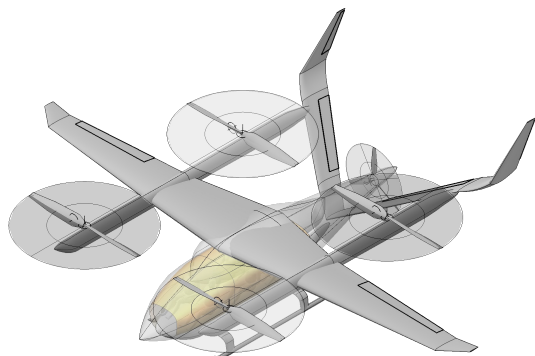
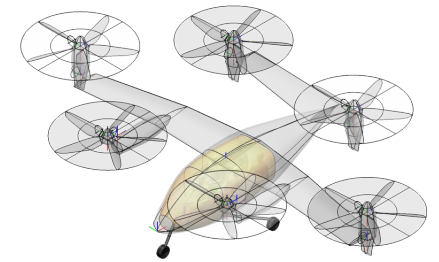
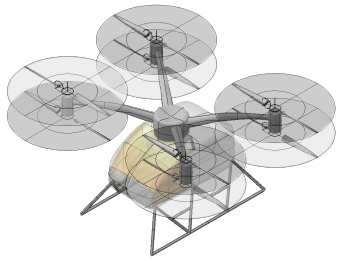
Stop some rotors,
tilt some rotors



Multicopters

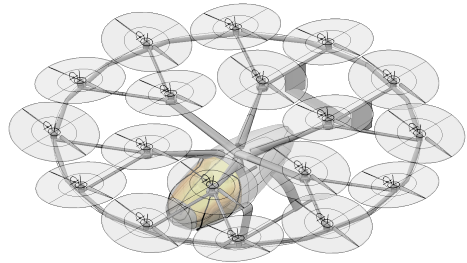


Tiltrotors and tiltwings



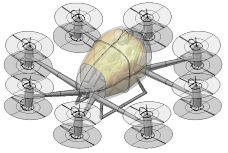
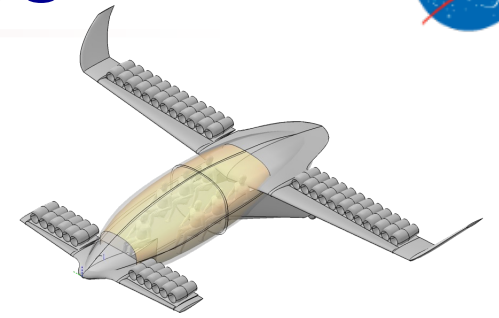
More like an airplane in cruise 

The arrangement of rotors indicate priorities

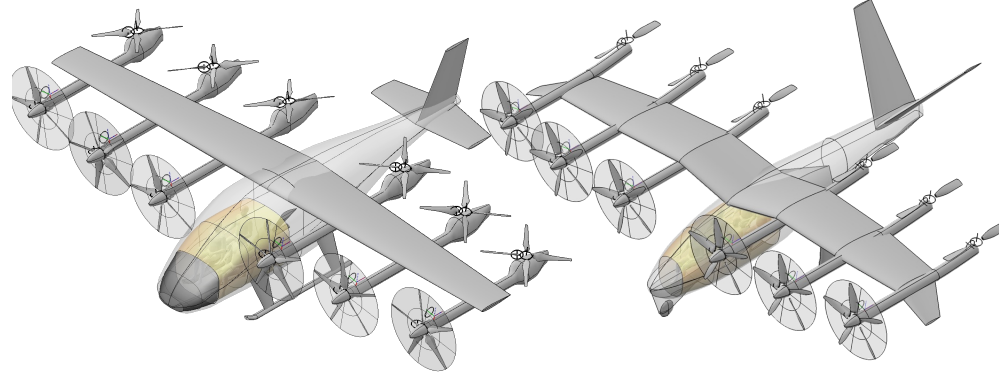
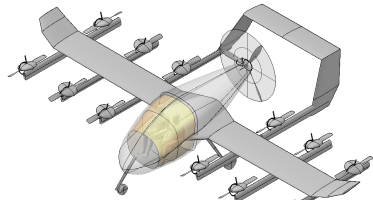


Stopping rotor
Lift+cruise

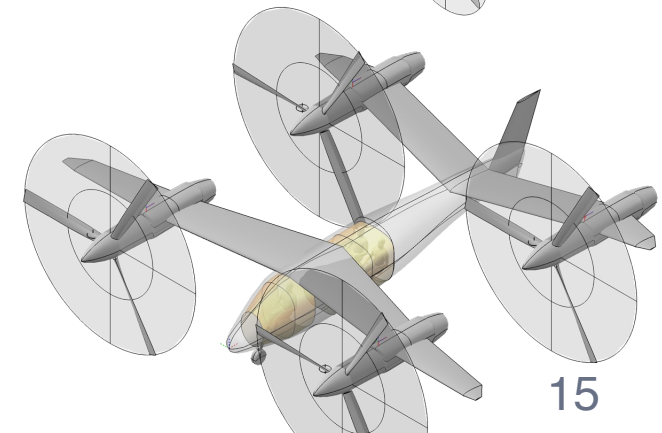
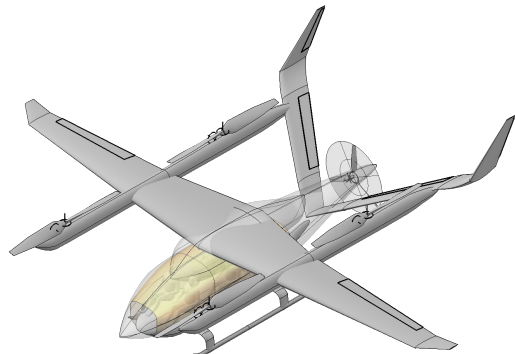
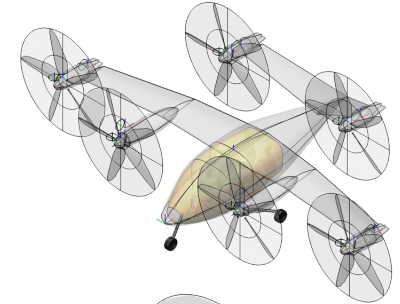
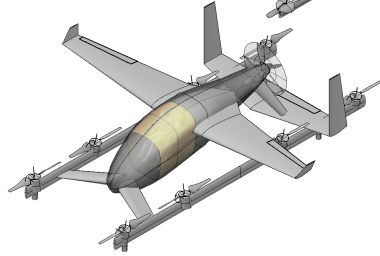
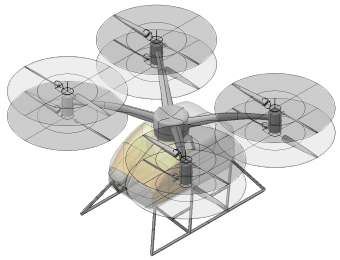
Stop some rotors,
tilt some rotors



Multicopters



Tiltrotors and tiltwings



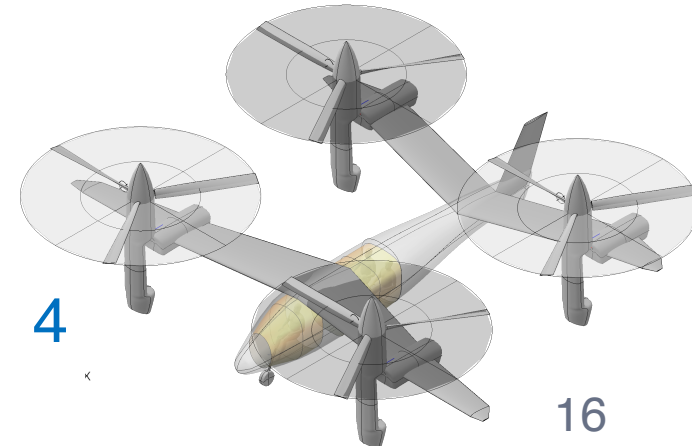
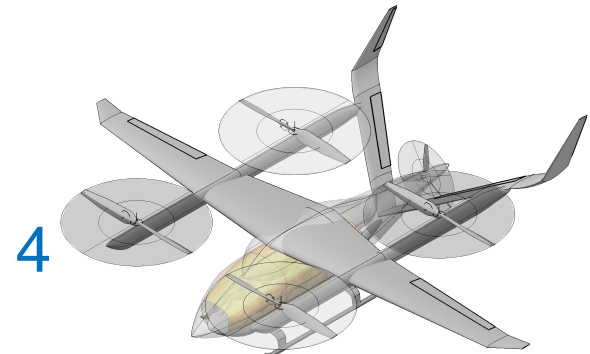
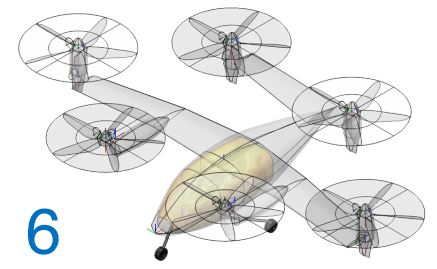
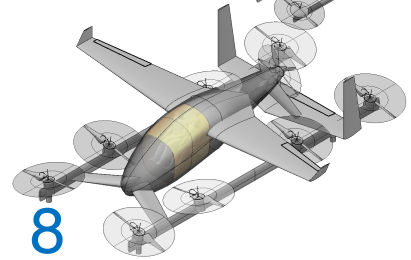
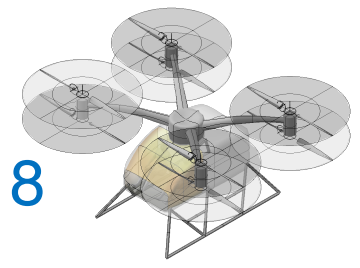
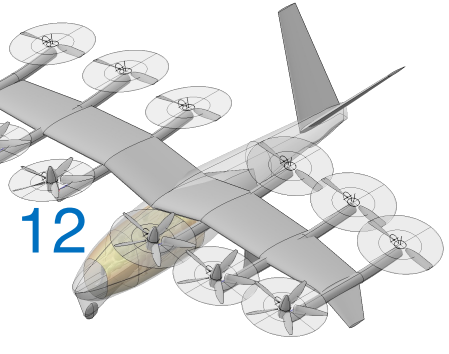
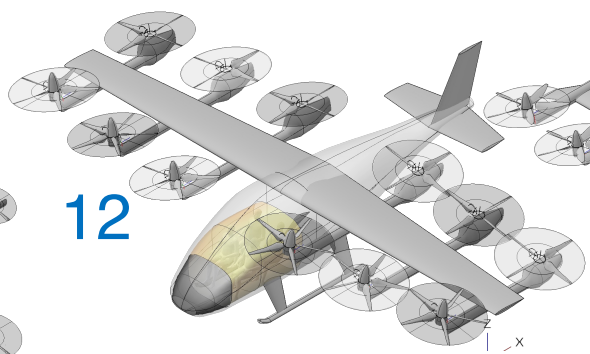
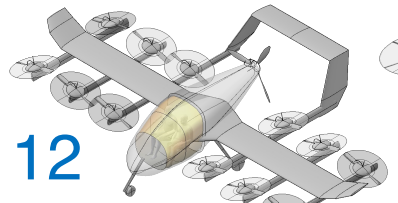
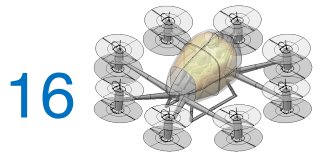
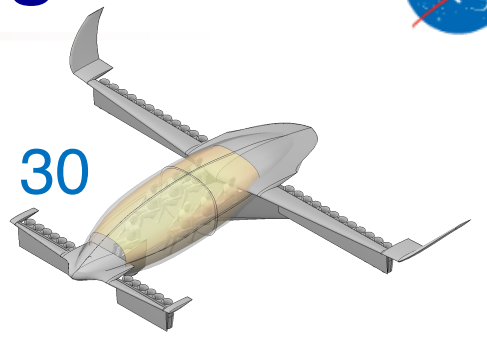
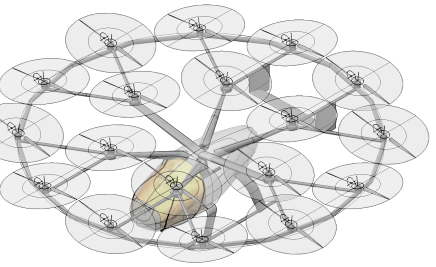
More like an airplane in cruise 

The arrangement of rotors indicate priorities



18
↑
16
8

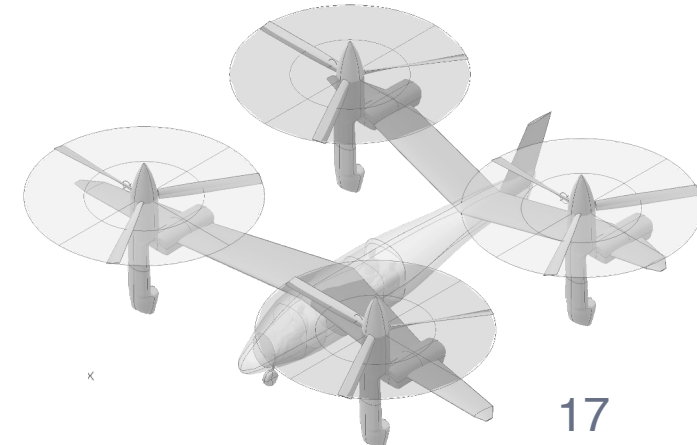
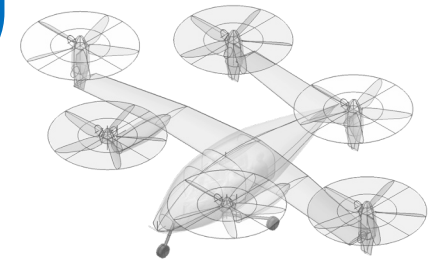
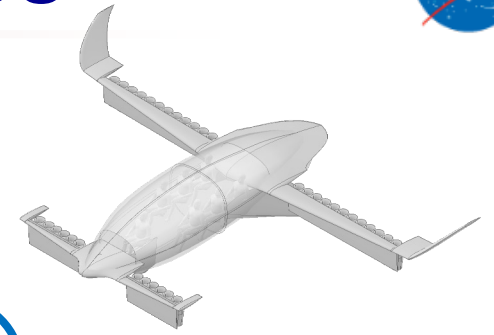
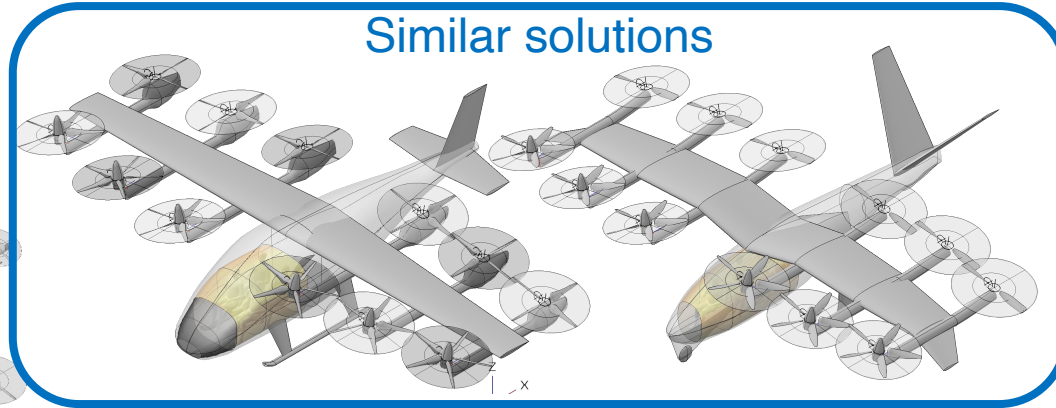
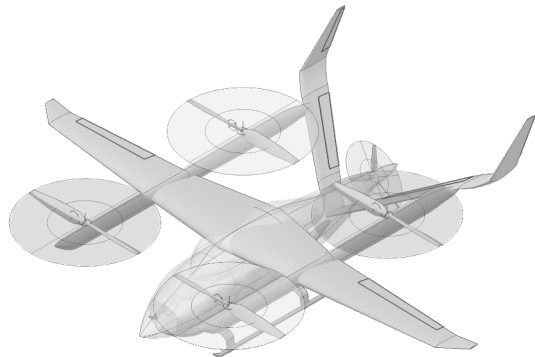
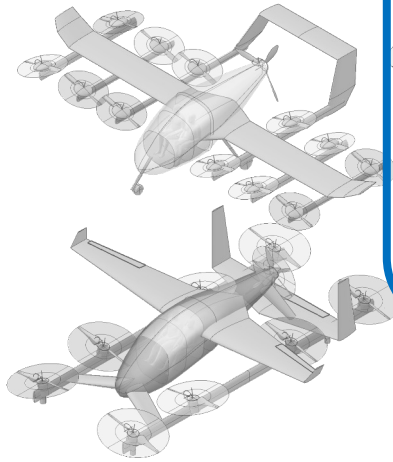
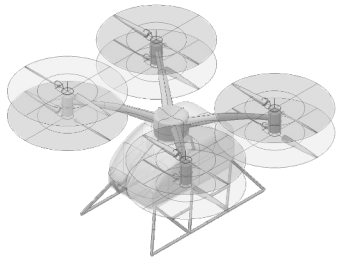
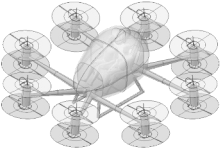
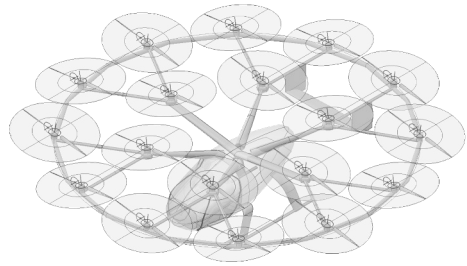
More lifting rotors for redundancy



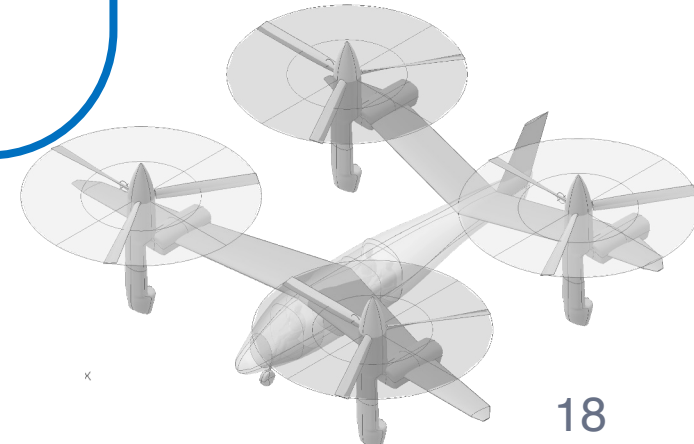
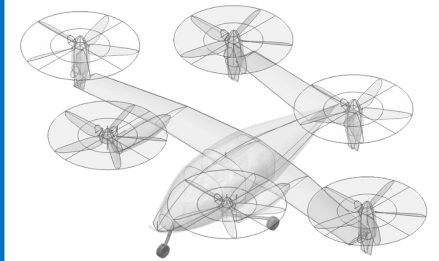
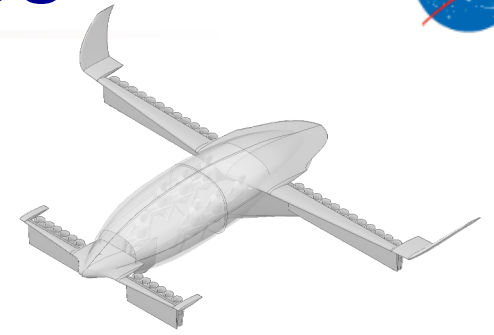
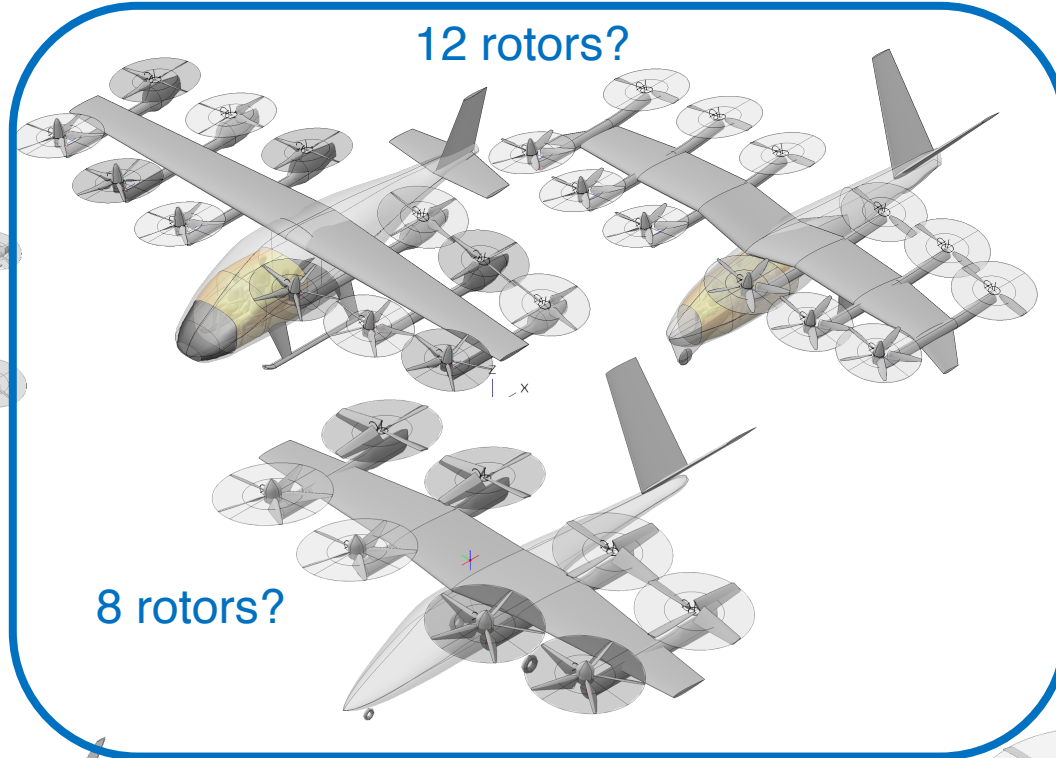
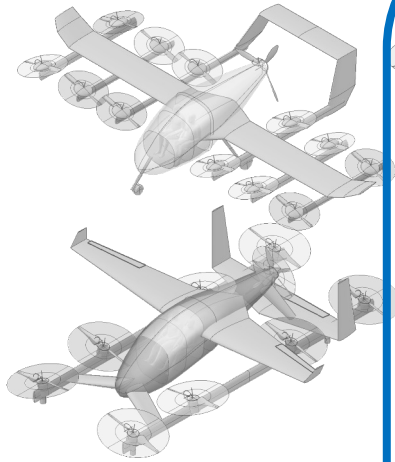
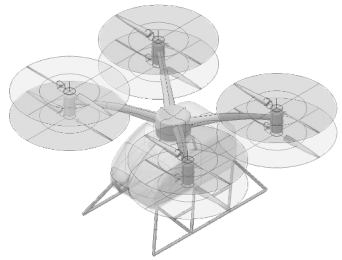
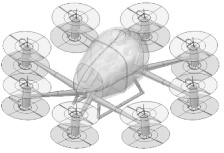
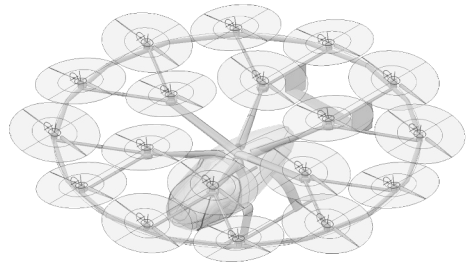
More like an airplane in cruise →

16

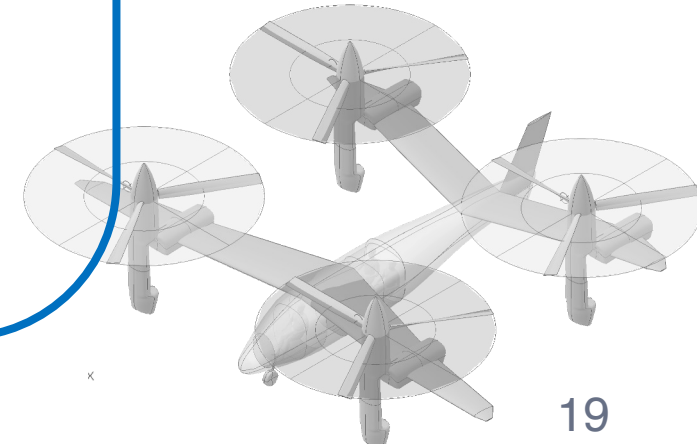
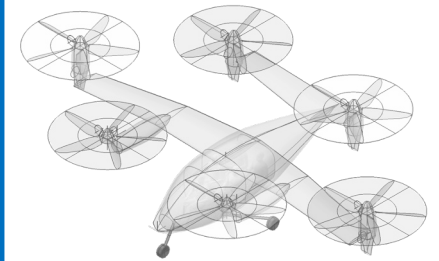
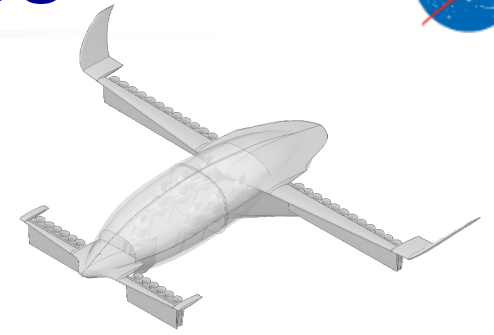
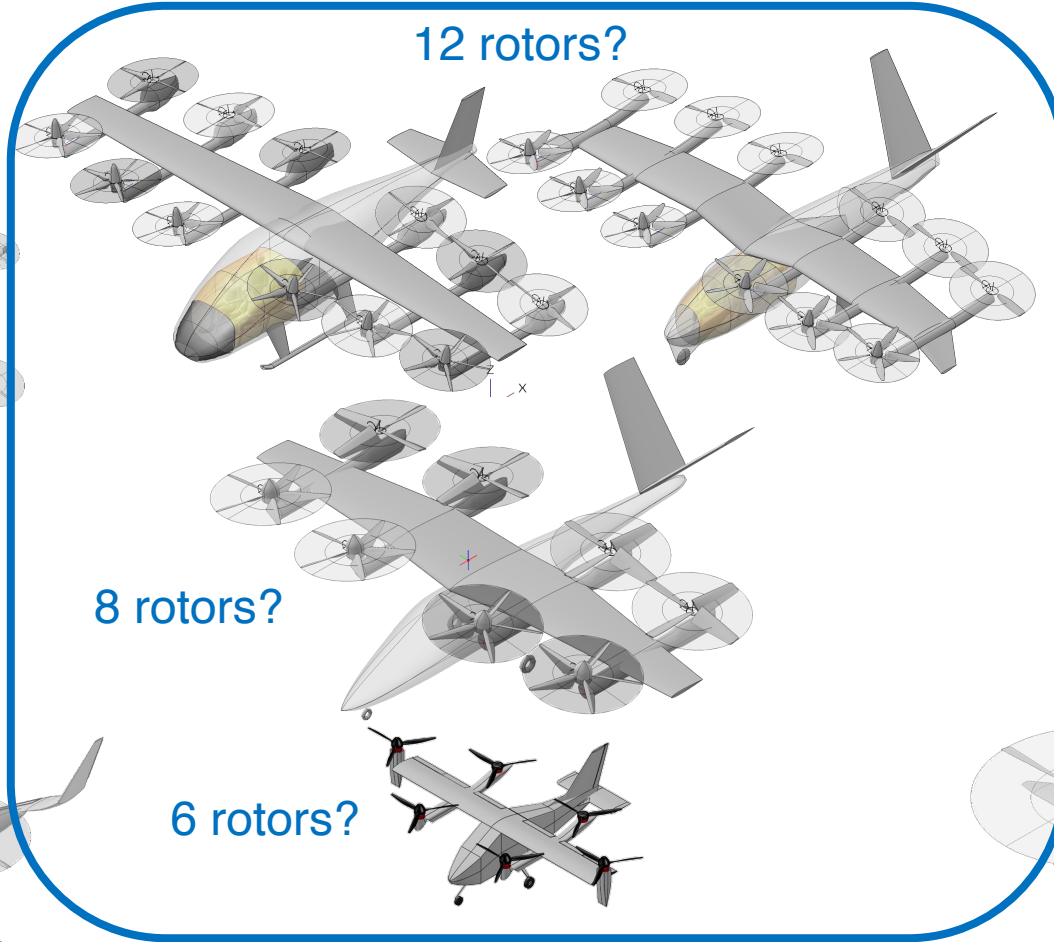
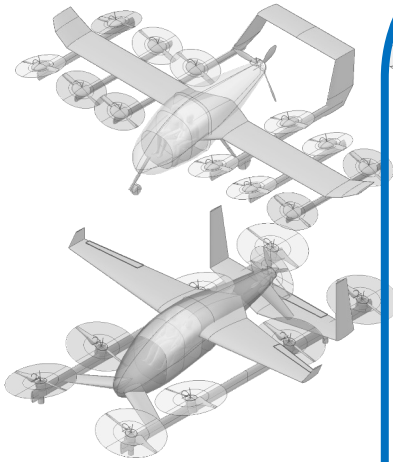
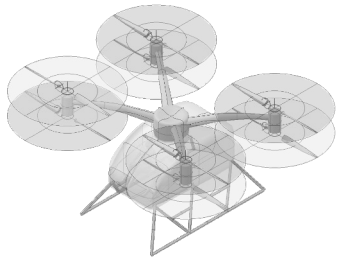
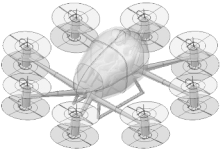
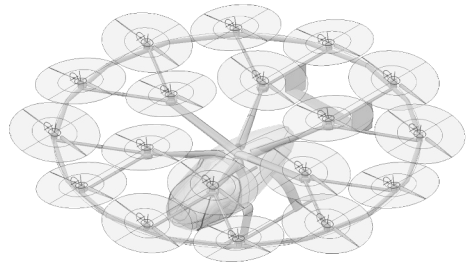
The arrangement of rotors indicate priorities



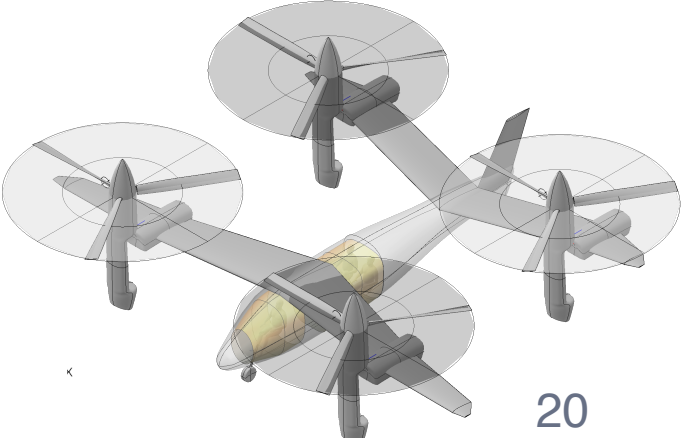
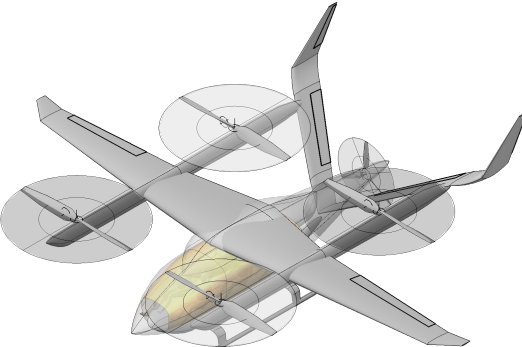
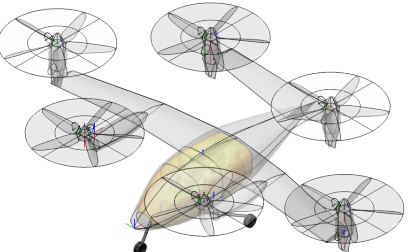
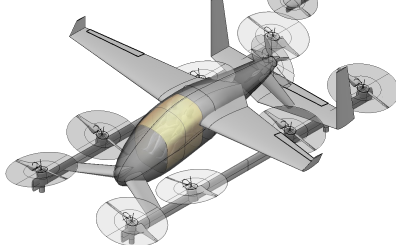
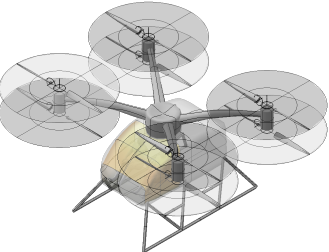
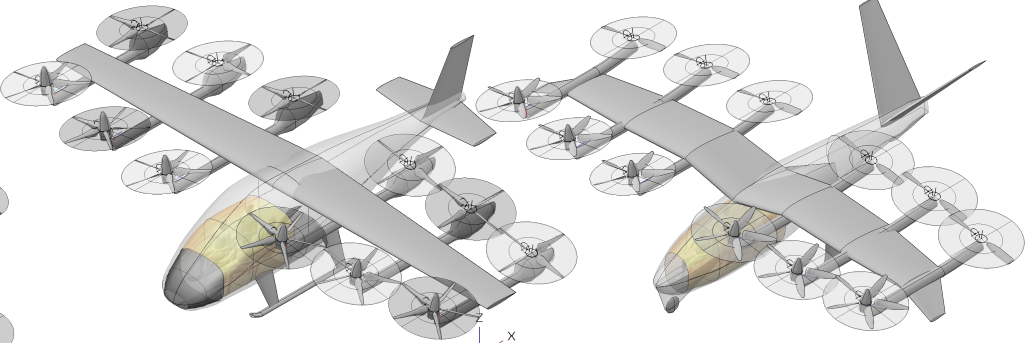
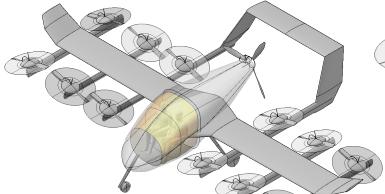
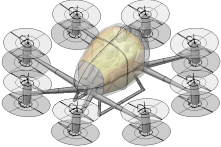
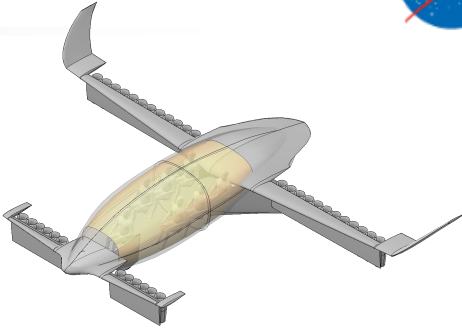
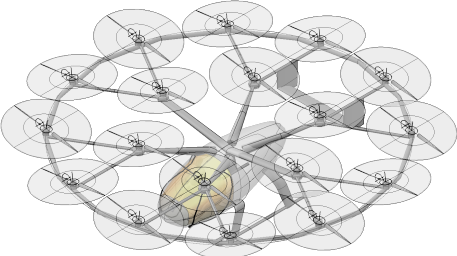
The arrangement of rotors indicate priorities



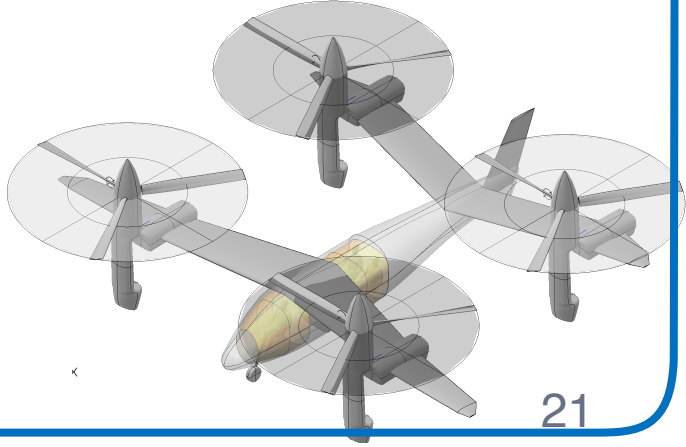
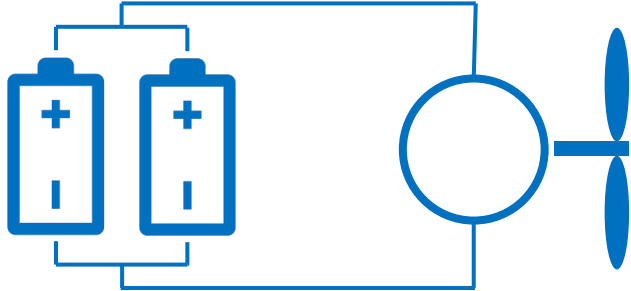
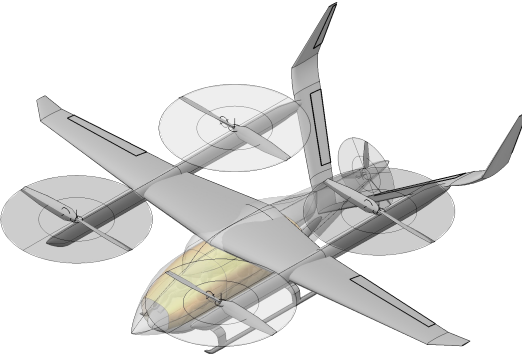
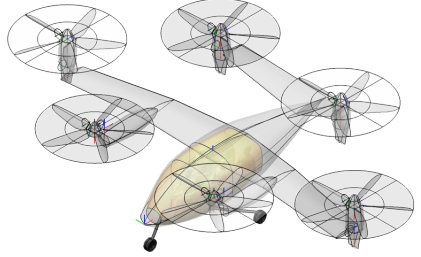
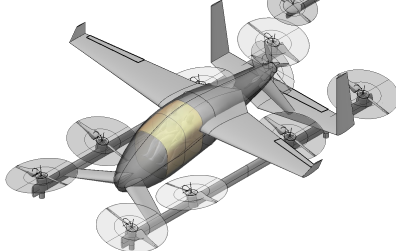
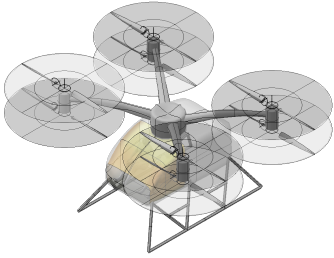
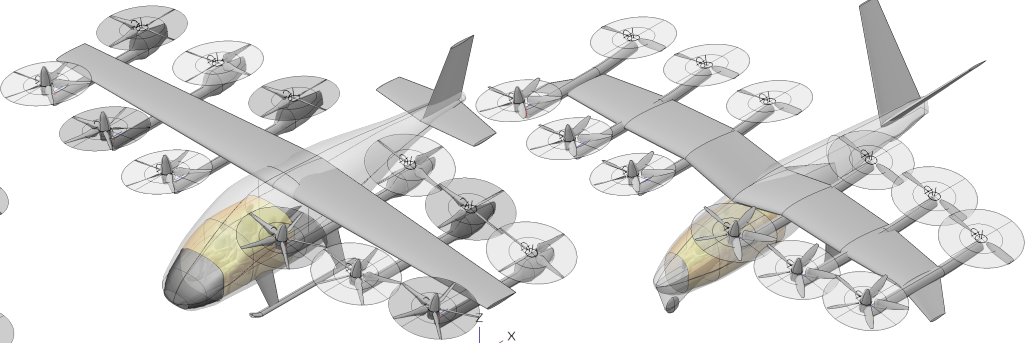
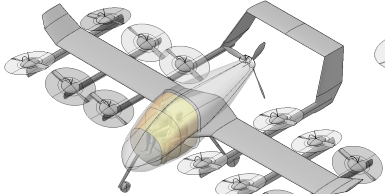
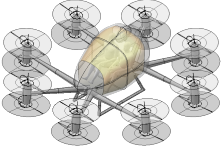
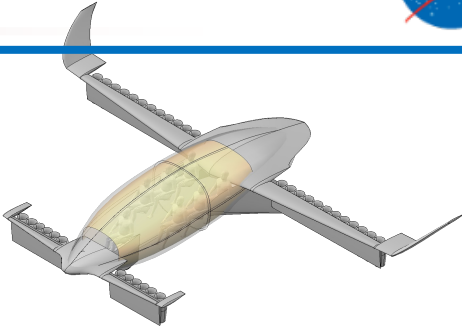
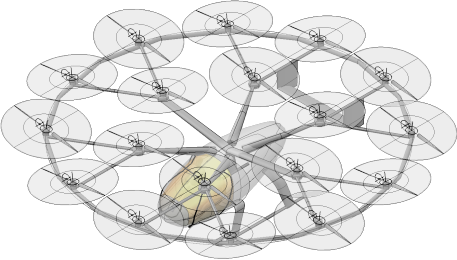
The arrangement of rotors indicate priorities



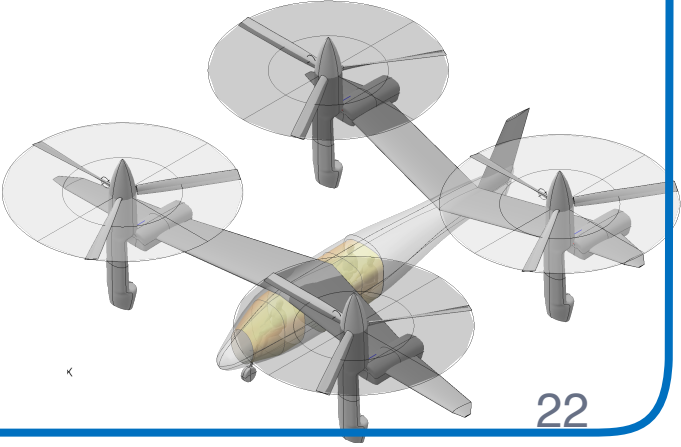
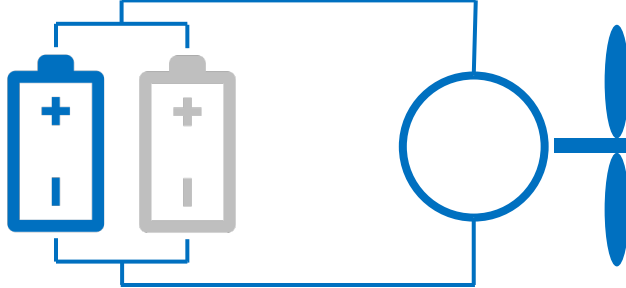
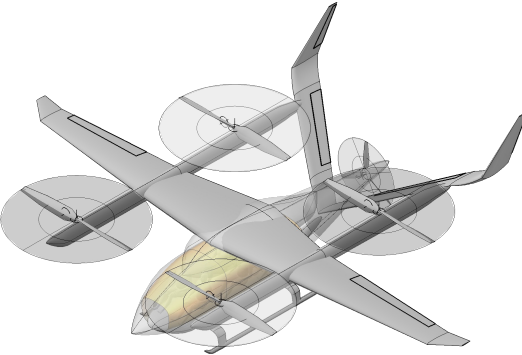
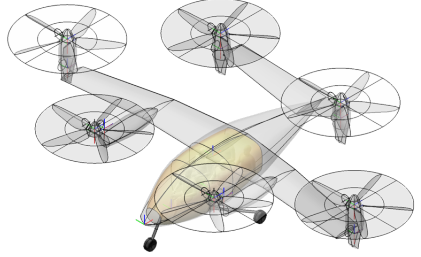
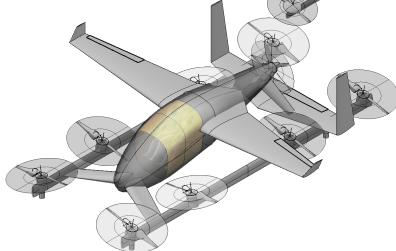
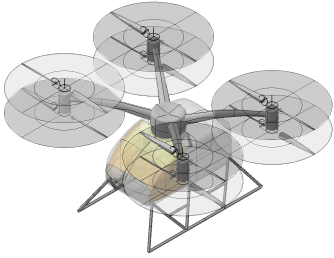
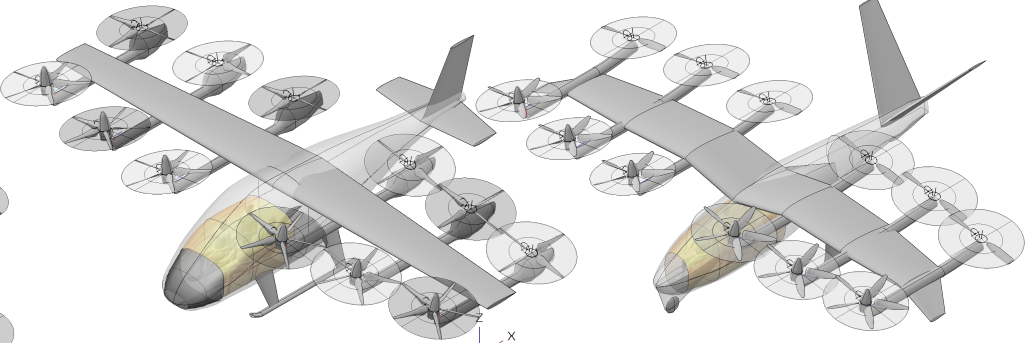
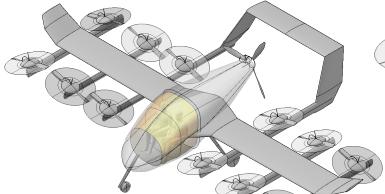
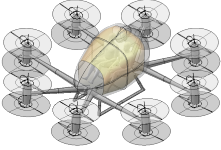
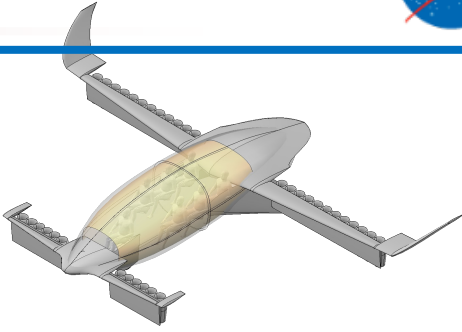
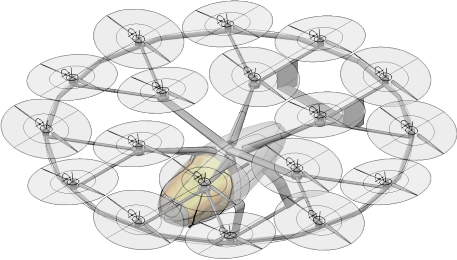
UAM Requirement: More safe



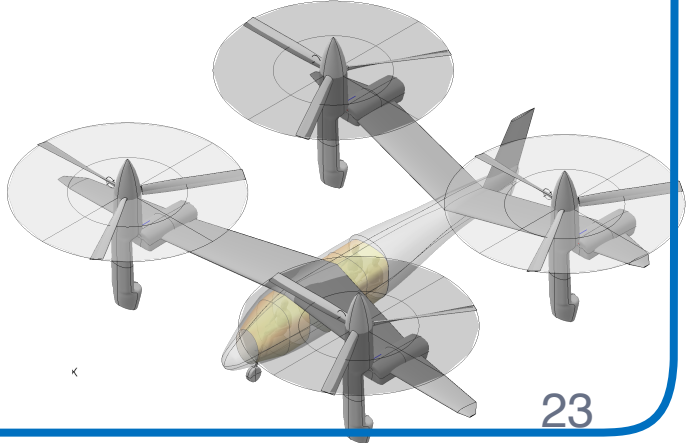
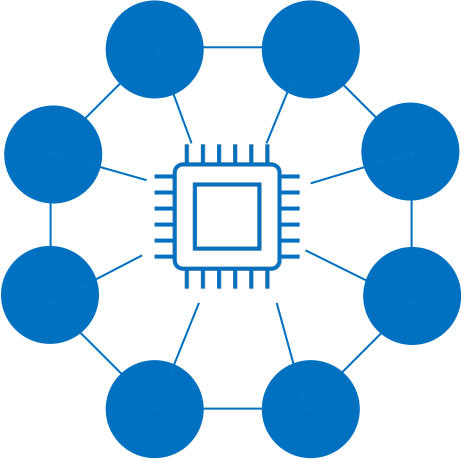
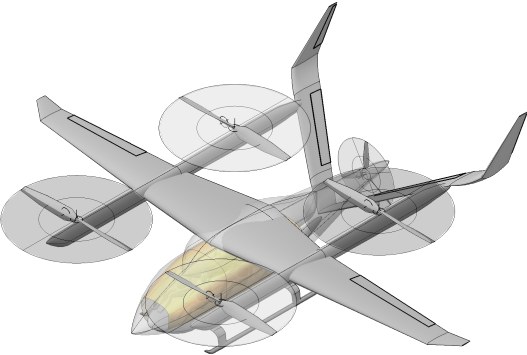
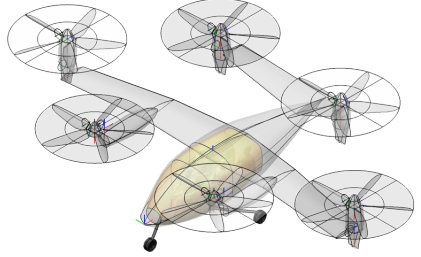
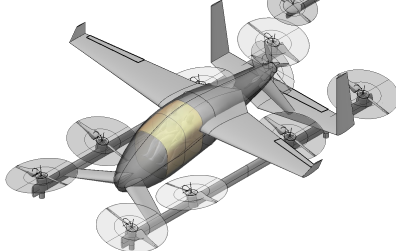
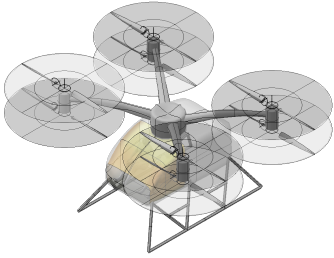
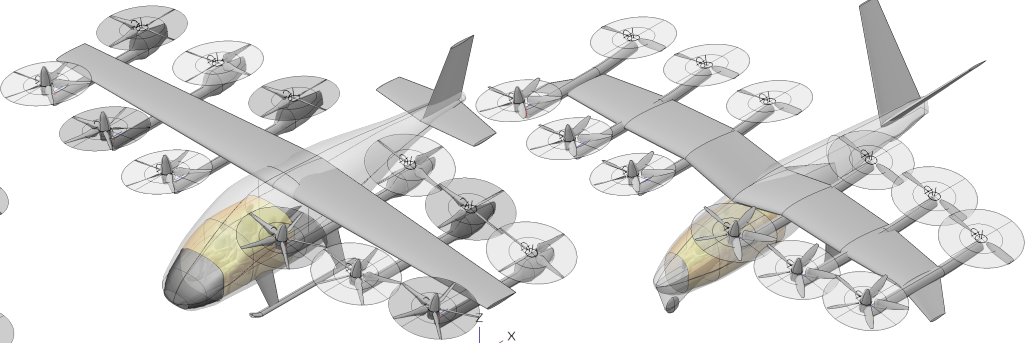
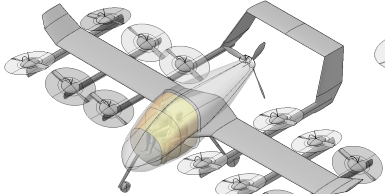
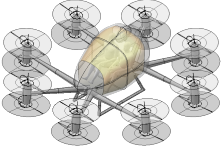
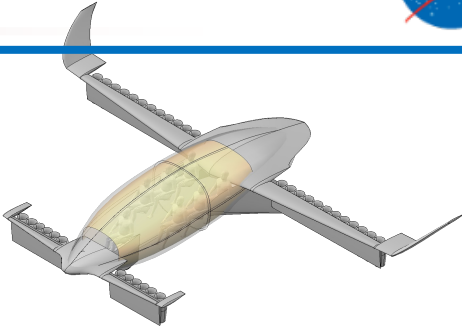
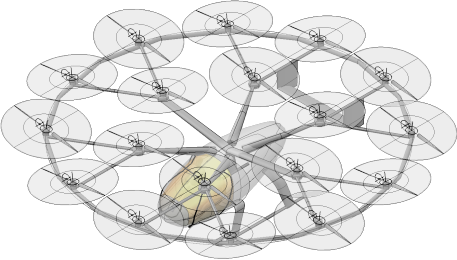
UAM Requirement: More safe



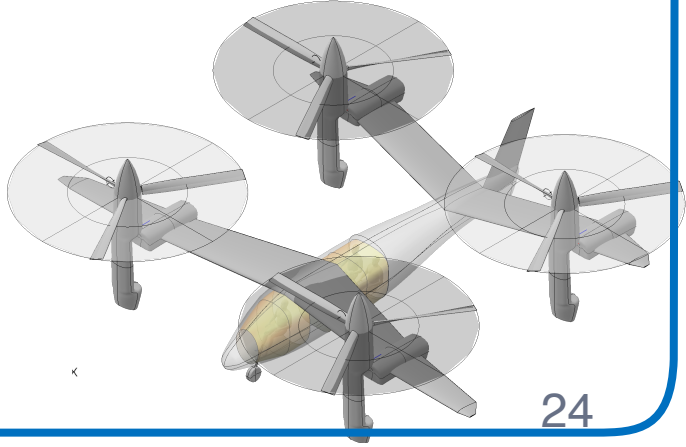
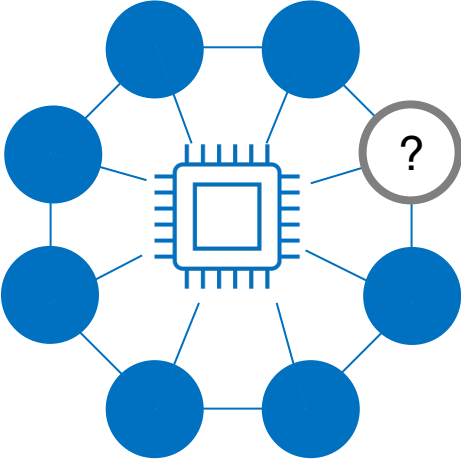
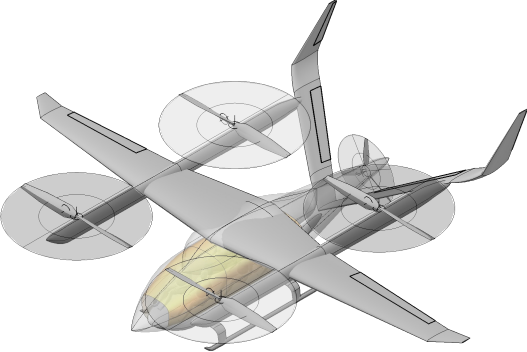
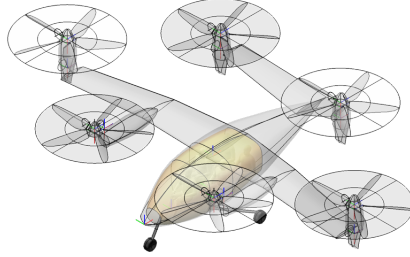
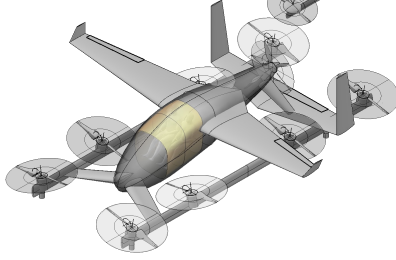
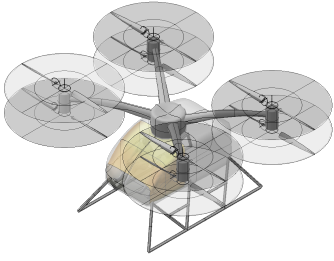
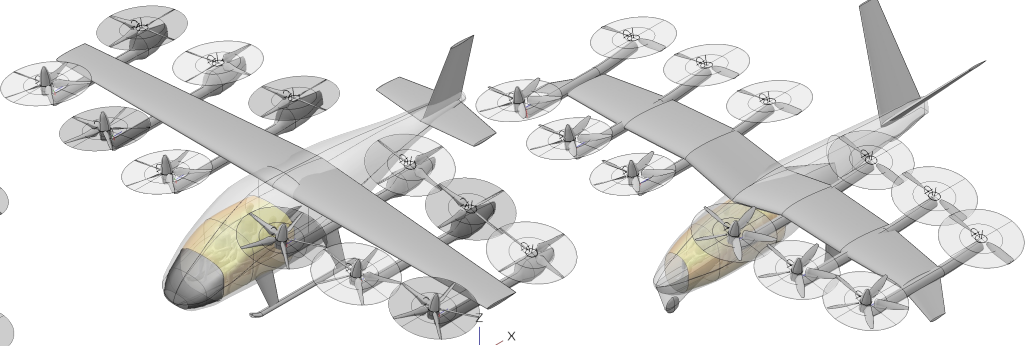
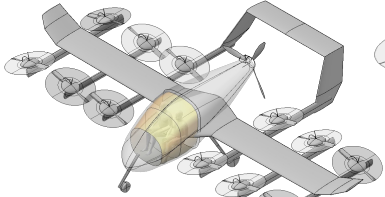
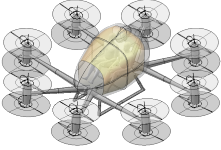
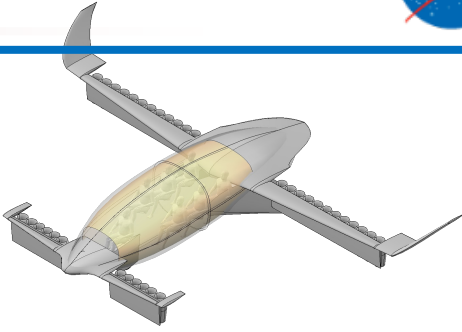
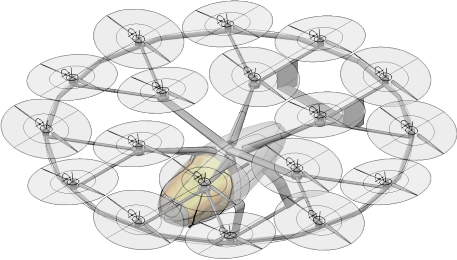
UAM Requirement: More safe



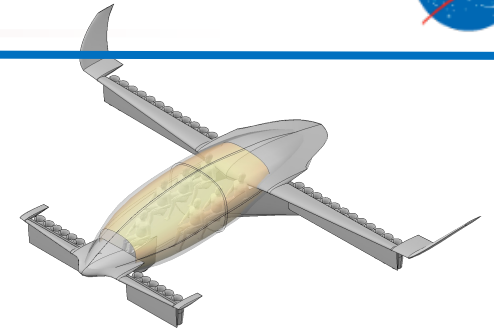
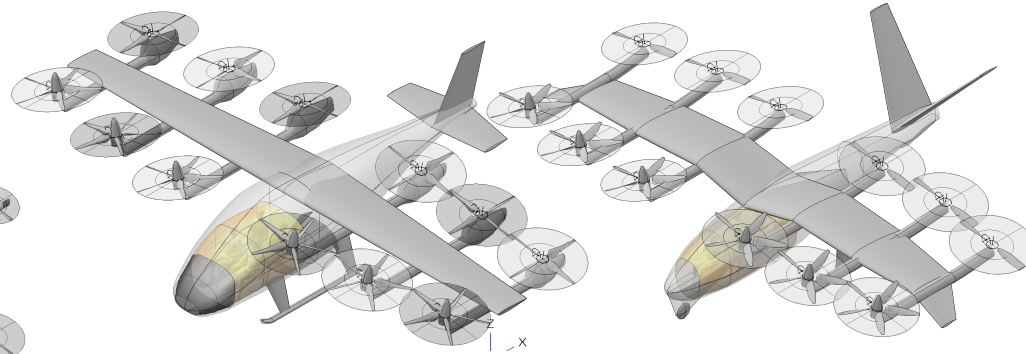
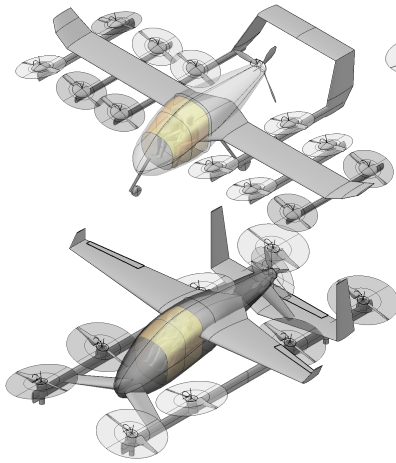
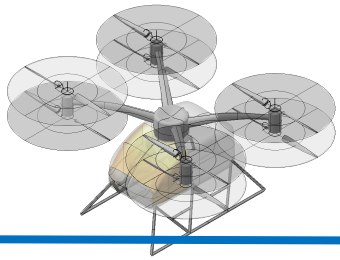
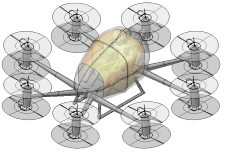
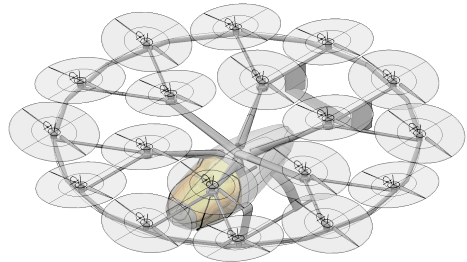
UAM Requirement: More safe



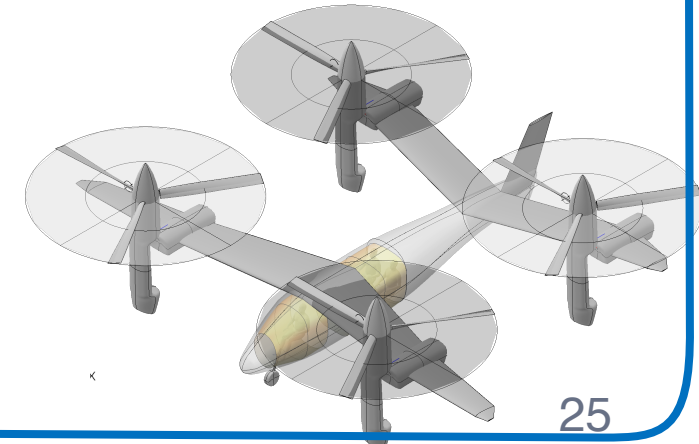
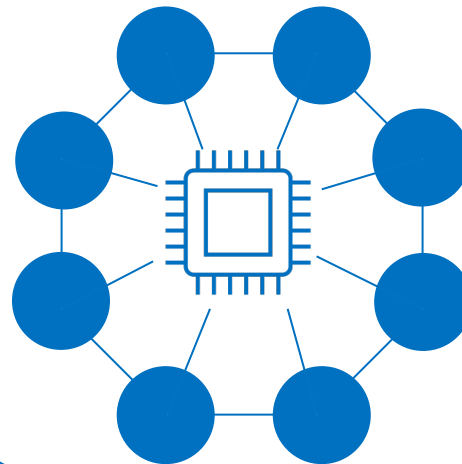
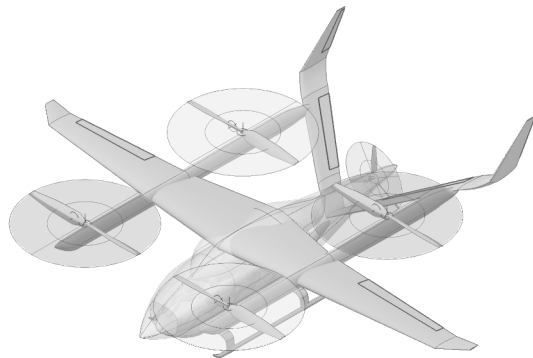
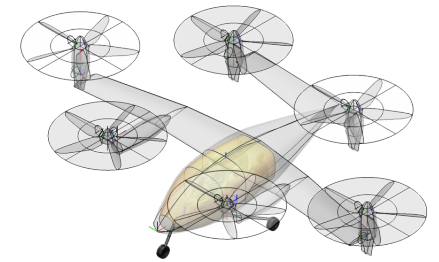
UAM Requirement: More safe



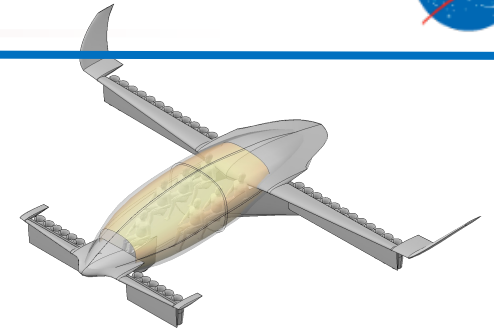
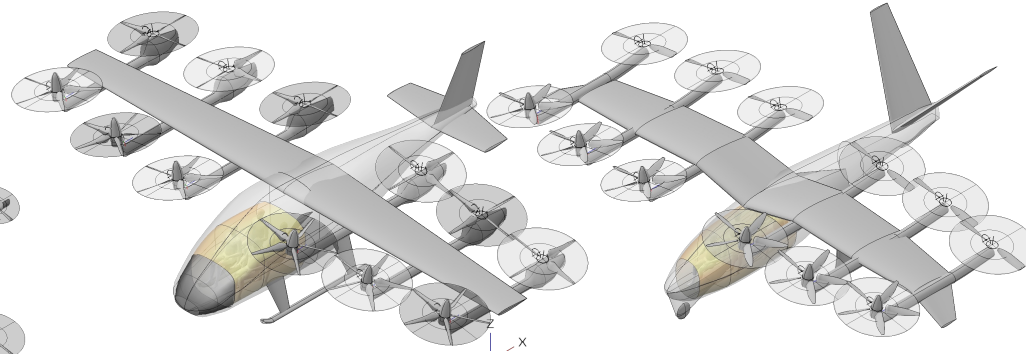
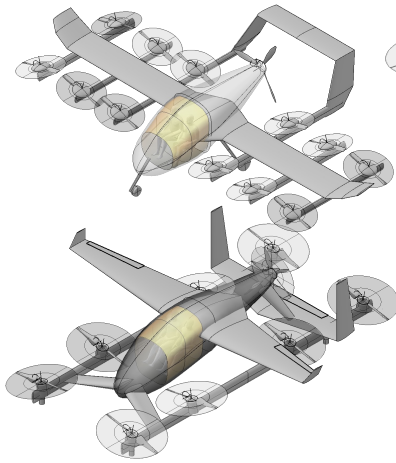
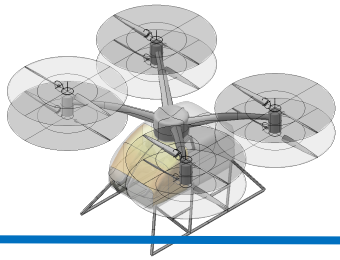
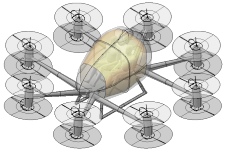
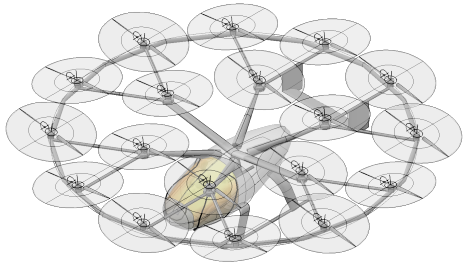
UAM Requirement: More safe



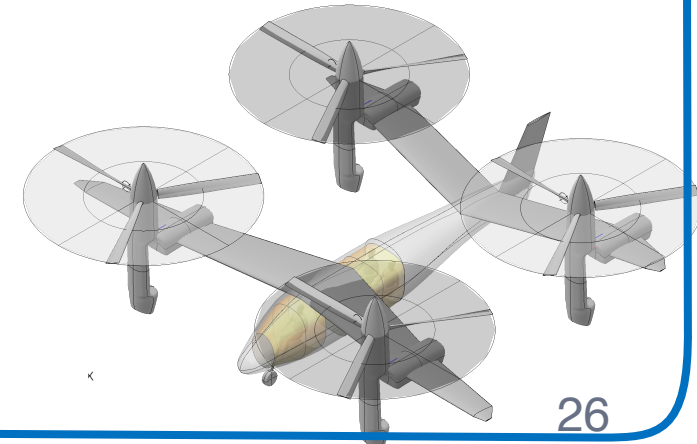
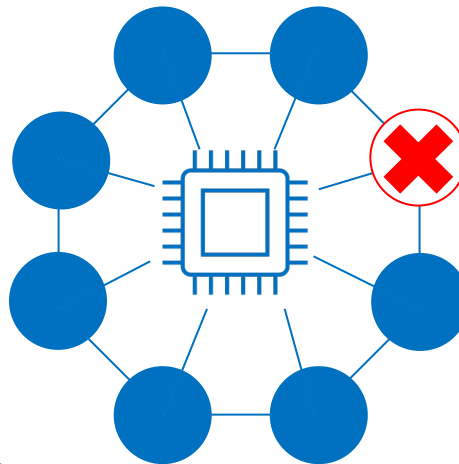
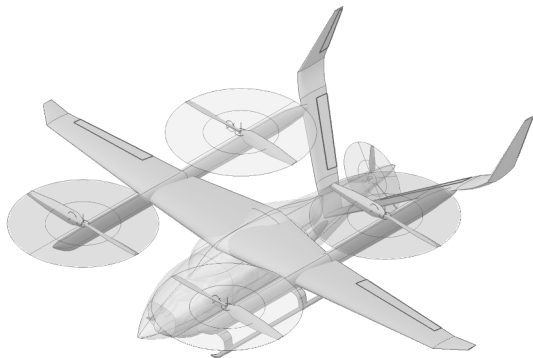
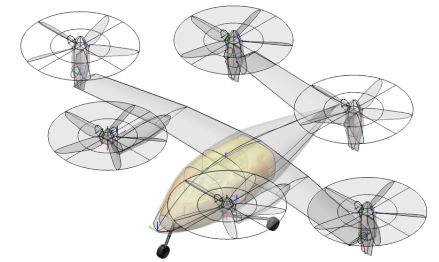
Tolerant to loss of power
in a single lifting rotor



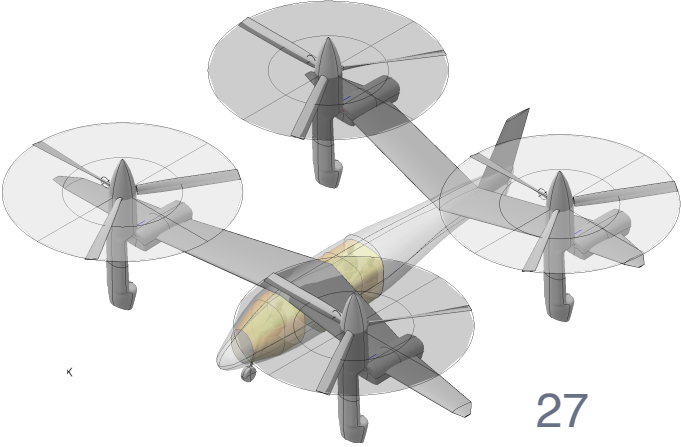
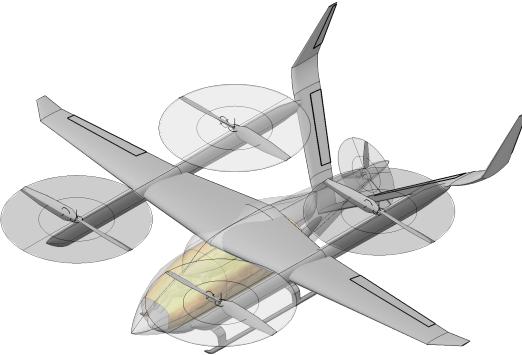
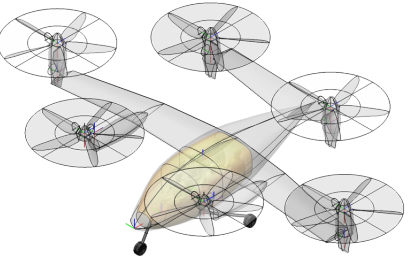
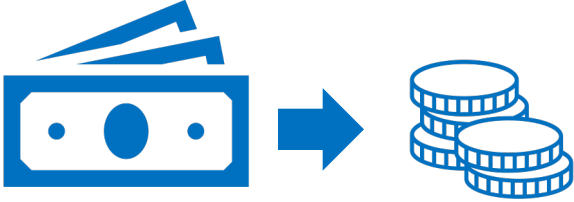
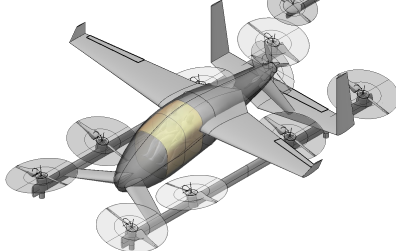
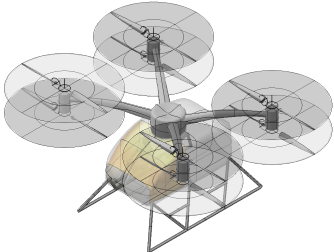
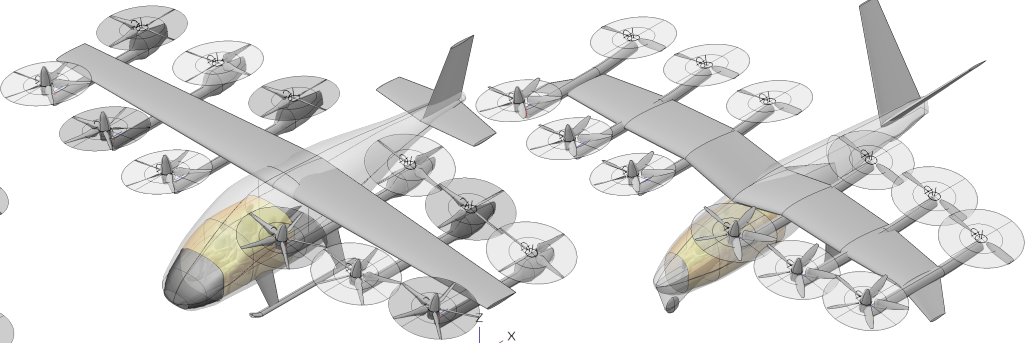
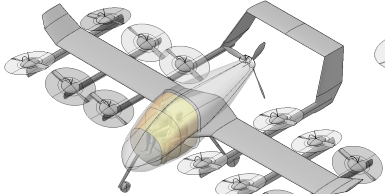
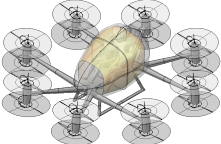
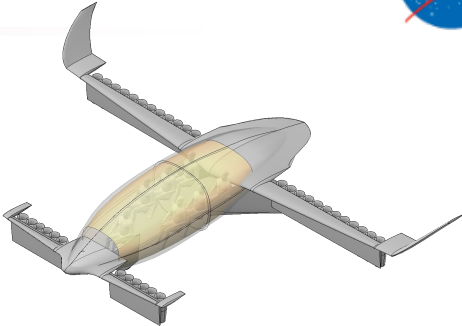
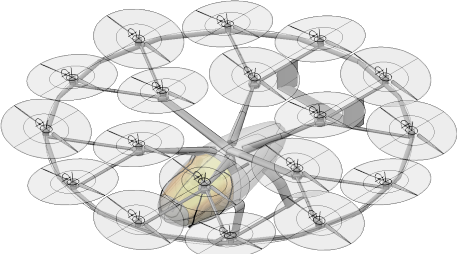
UAM Requirement: More safe



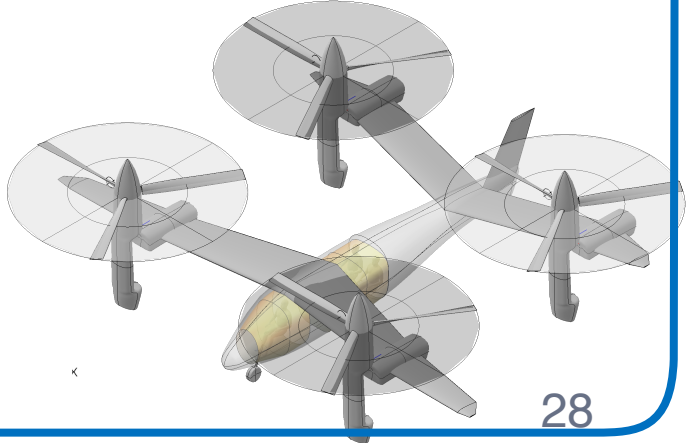
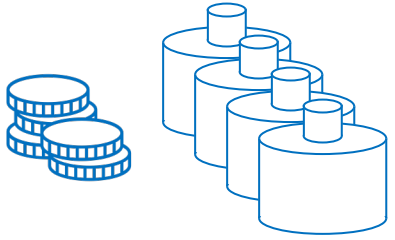
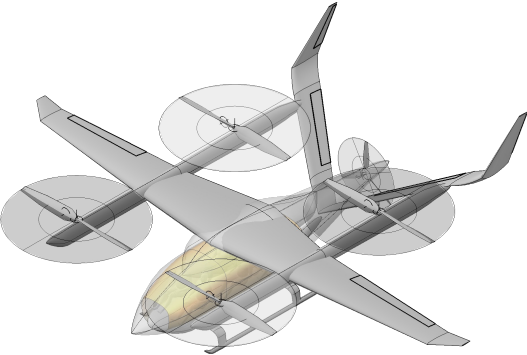
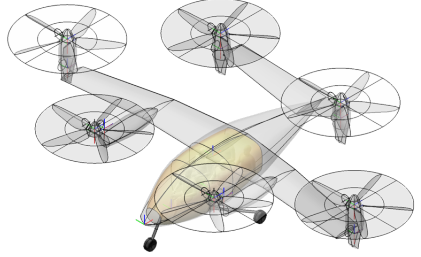
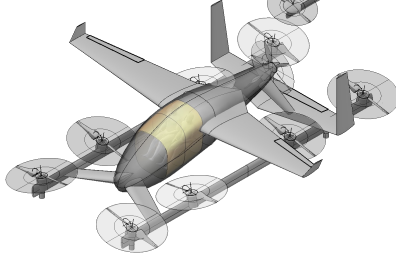
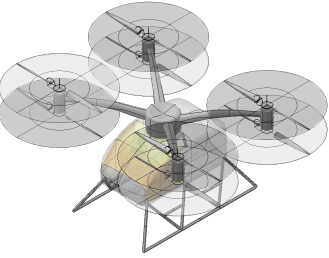
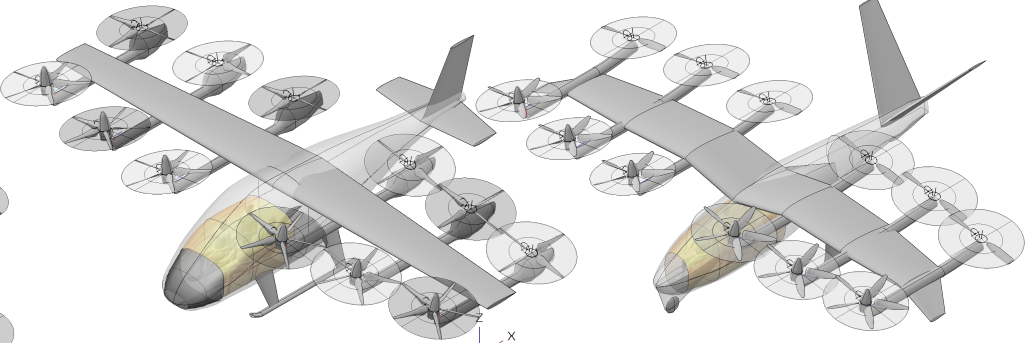
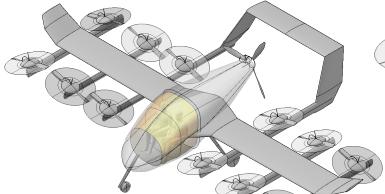
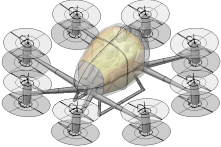
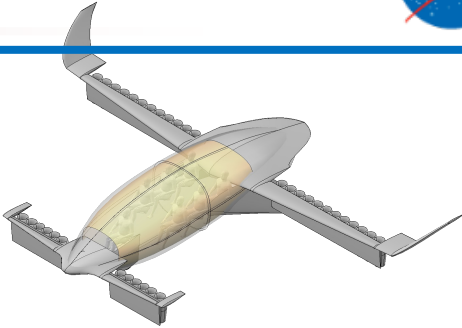
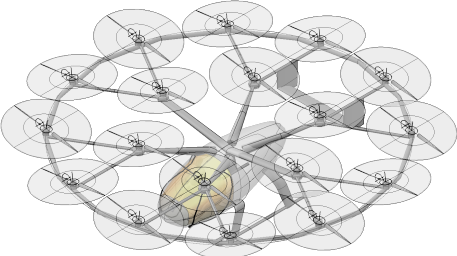
Tolerant to loss of power
in a single lifting rotor



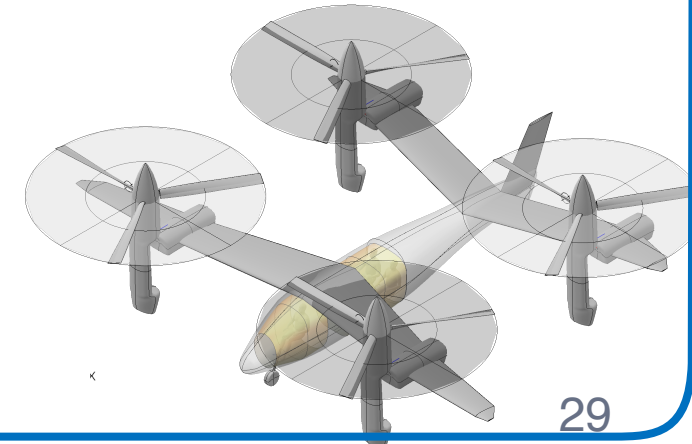
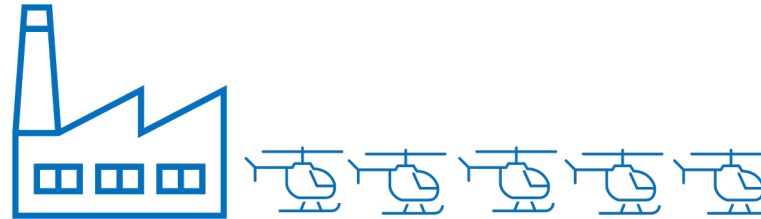
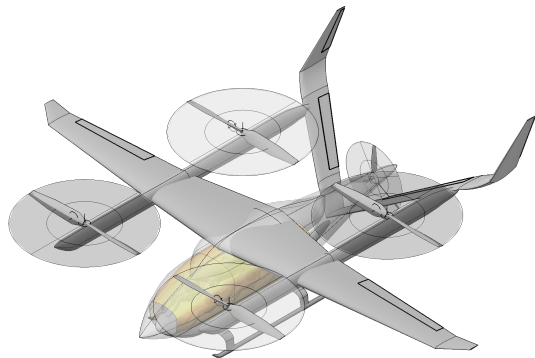
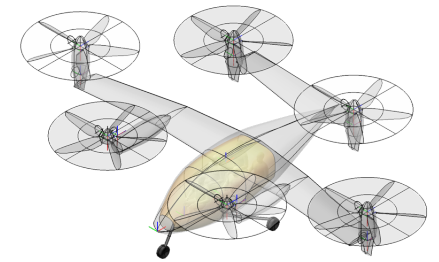
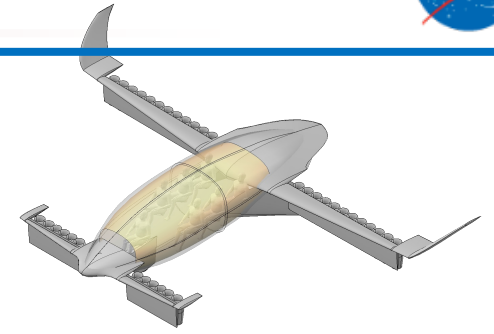
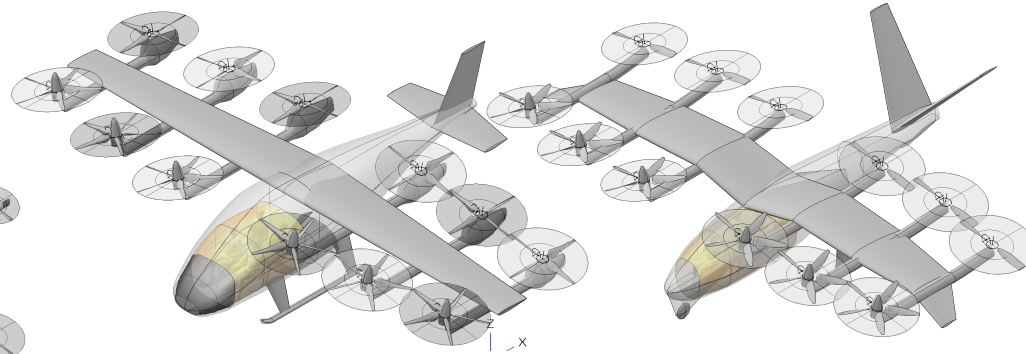
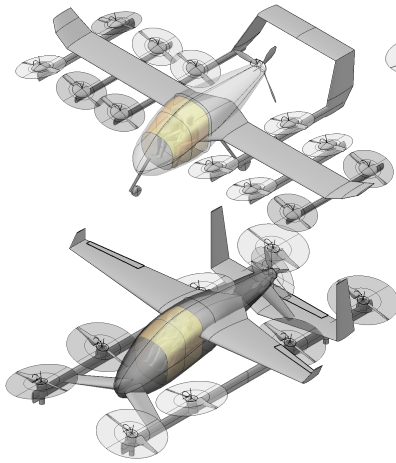
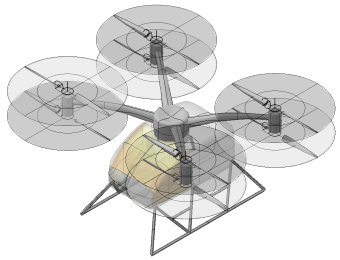
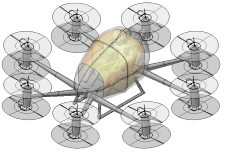
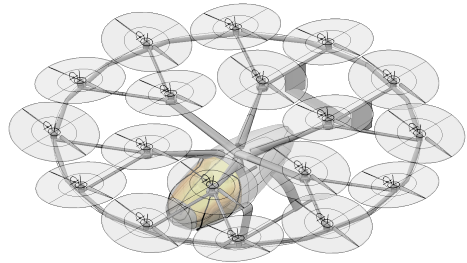
UAM Requirement: Less cost



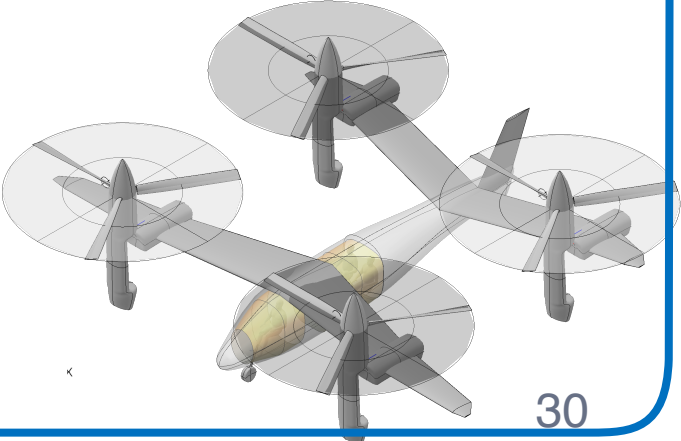
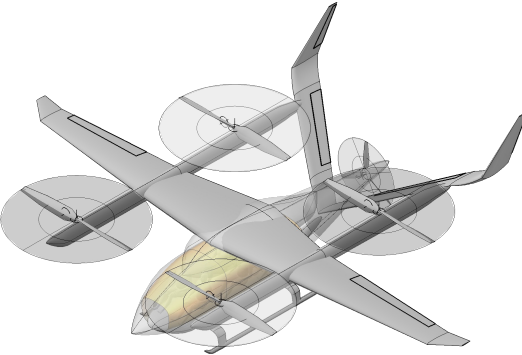
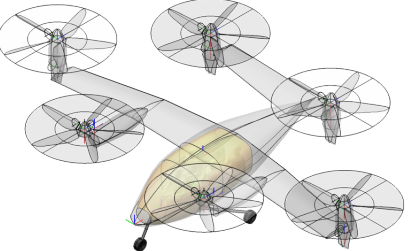
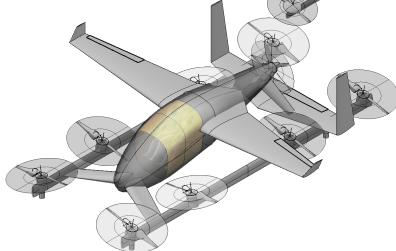
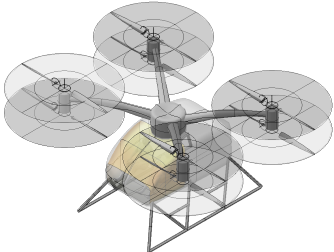
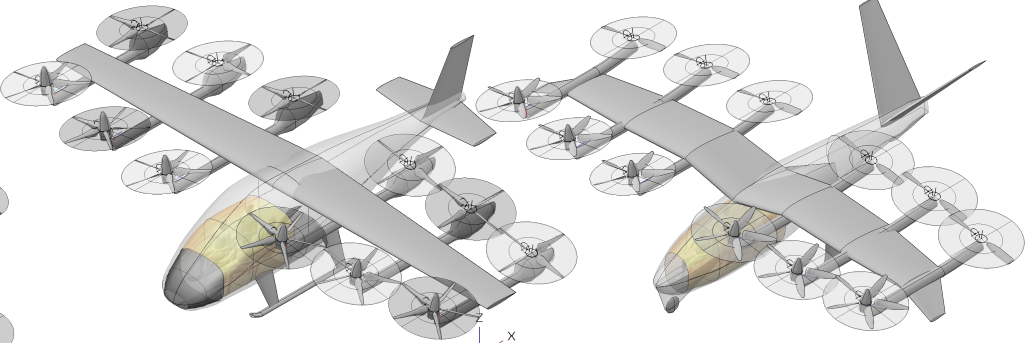
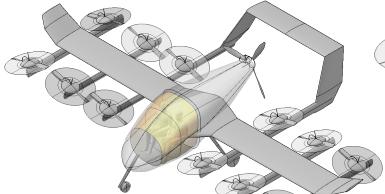
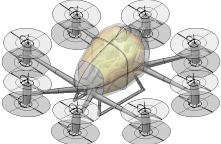
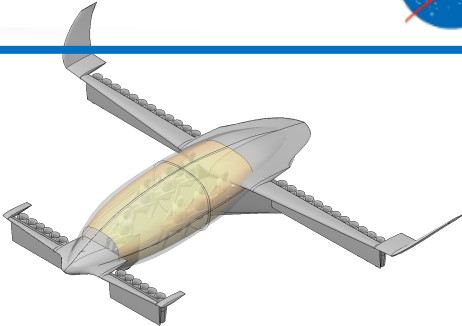
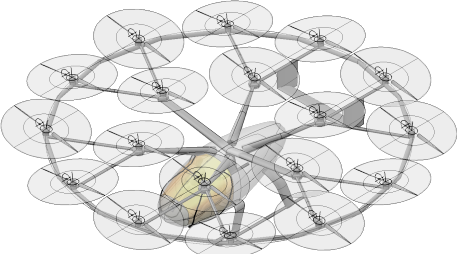
UAM Requirement: Less cost



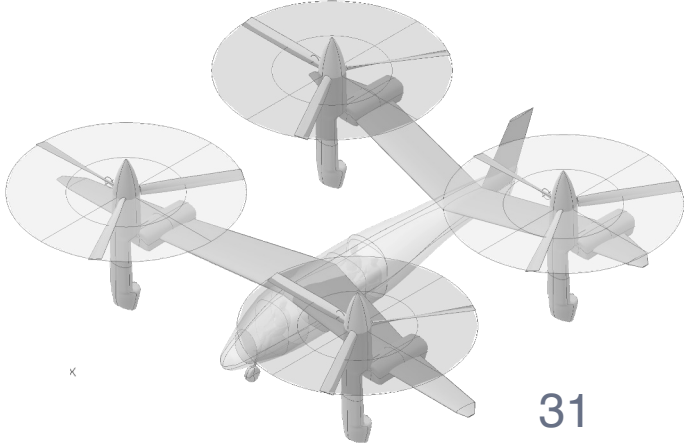
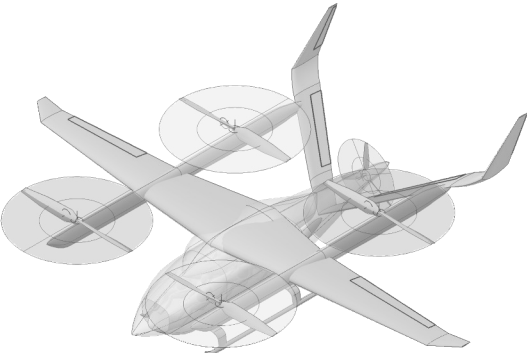
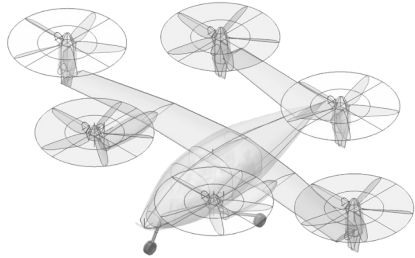
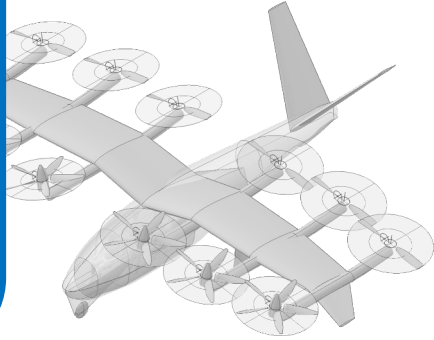
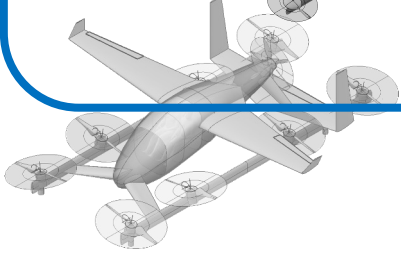
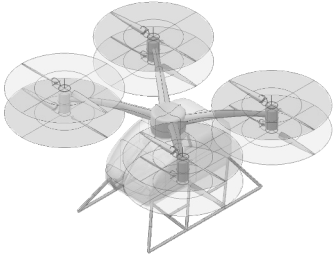
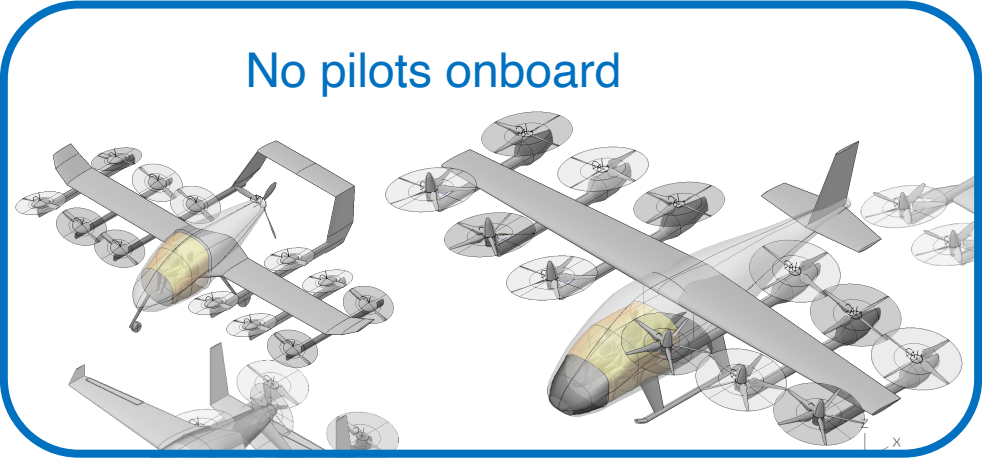
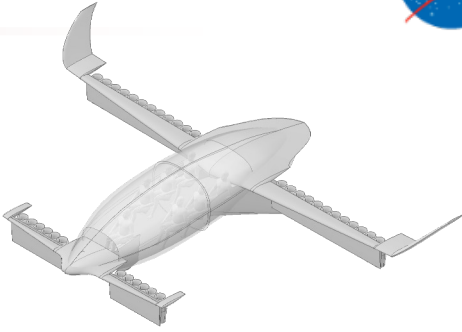
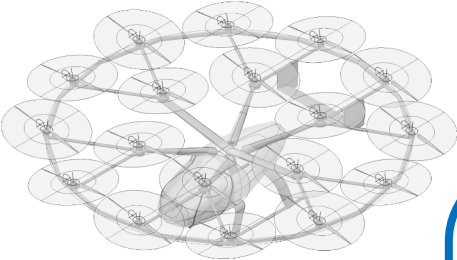
UAM Requirement: Less cost



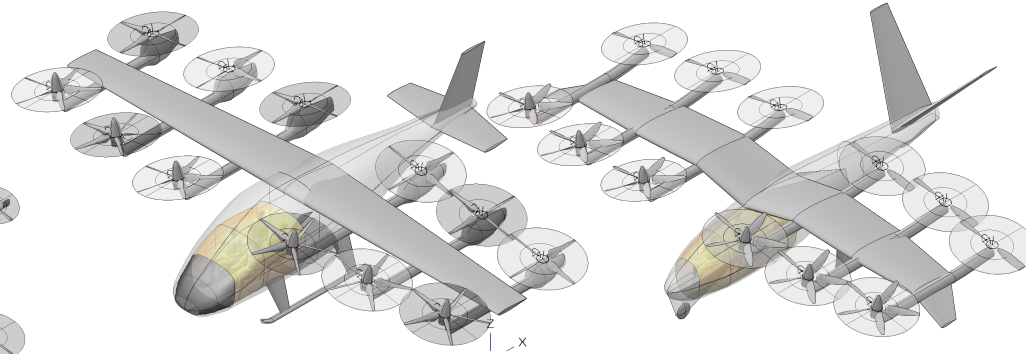
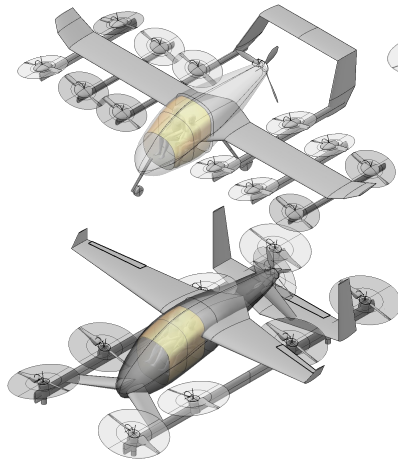
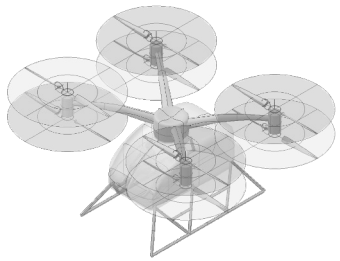
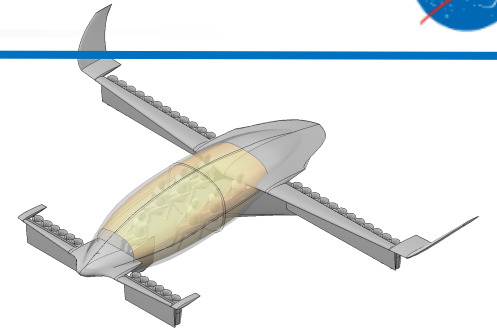
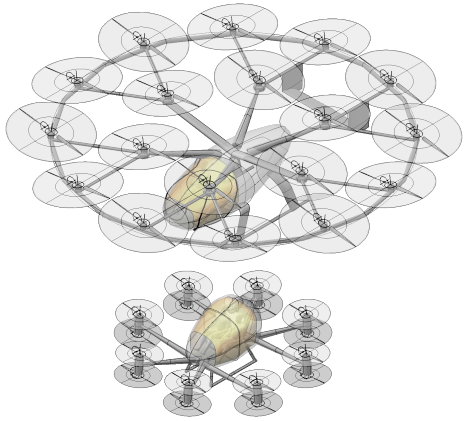
UAM Requirement: Less cost



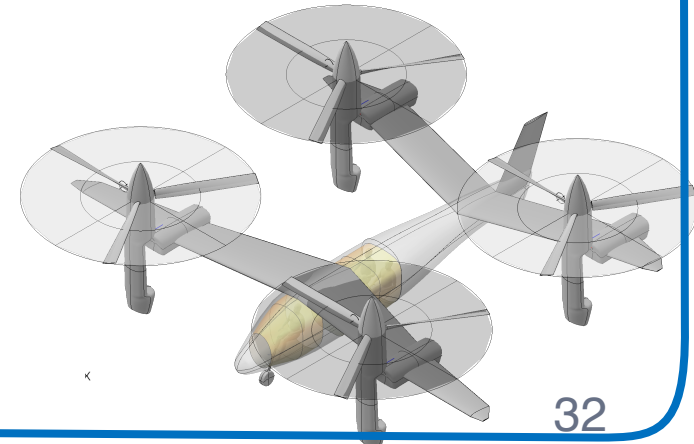
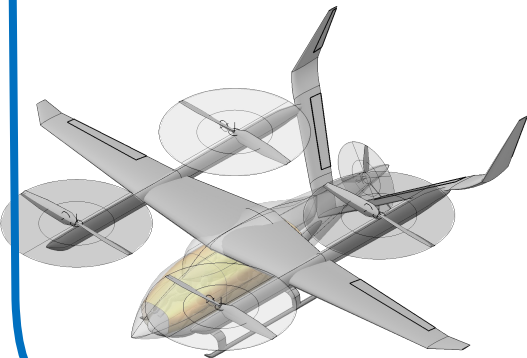
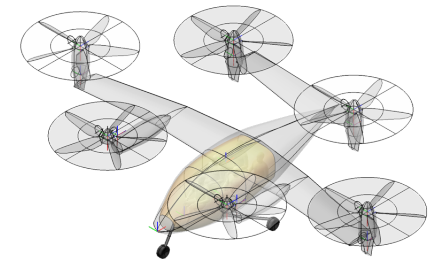
UAM Requirement: Less cost



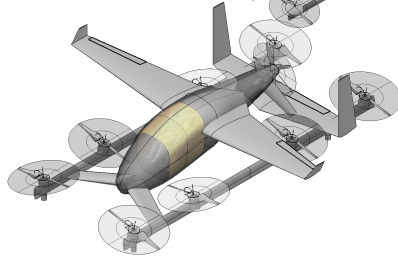
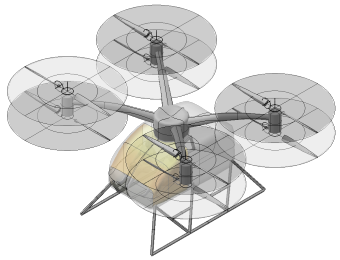
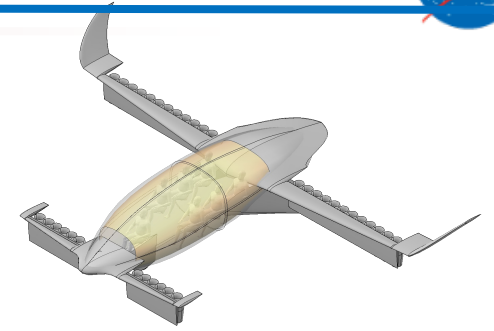
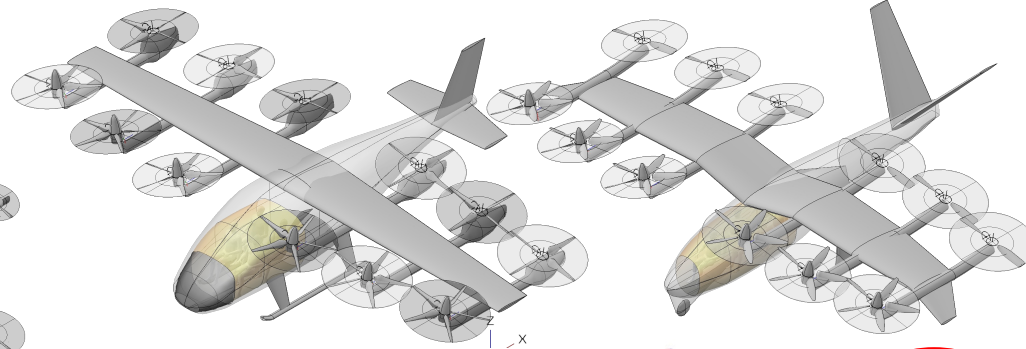
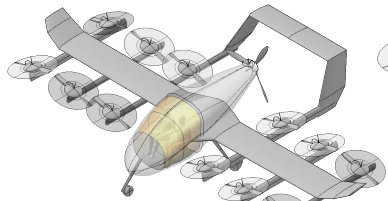
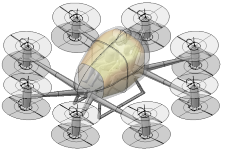
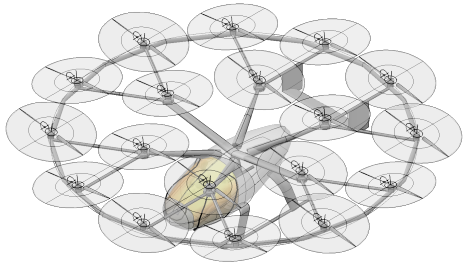
UAM Requirement: Less cost



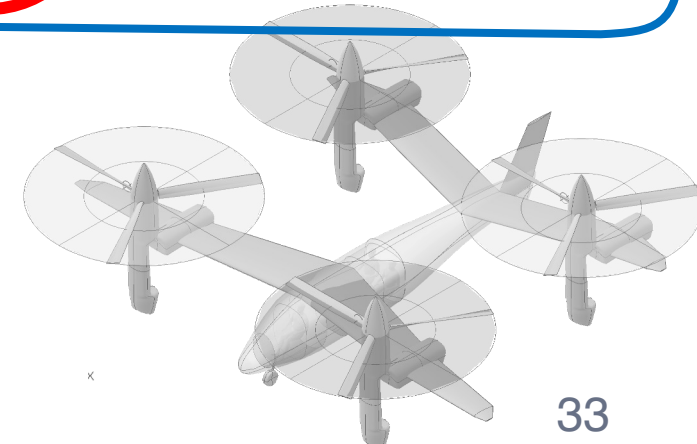
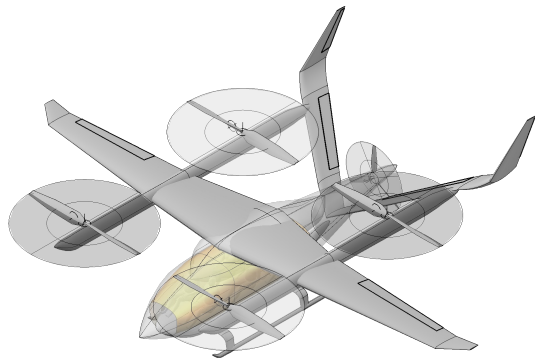
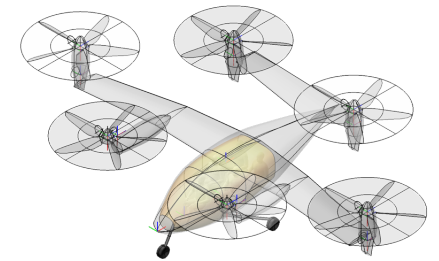
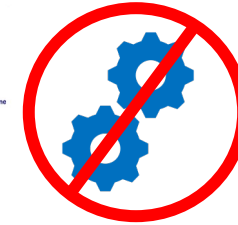
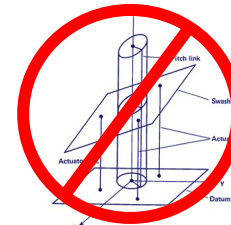
Battery electric propulsion
from cheap electricity



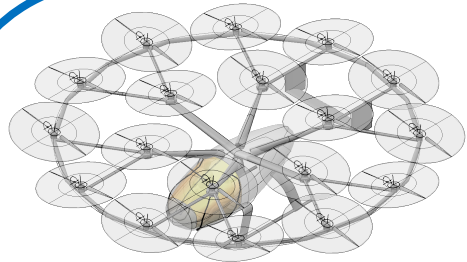
UAM Requirement: Less cost



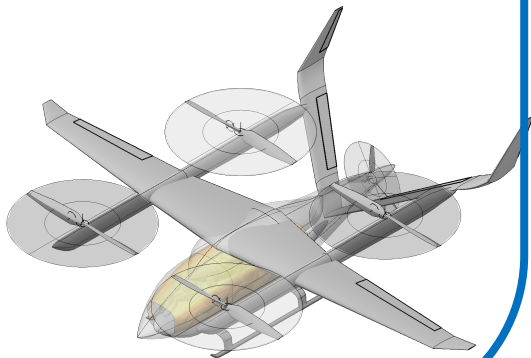
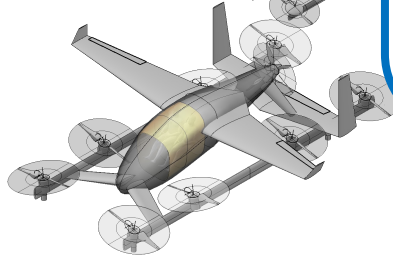
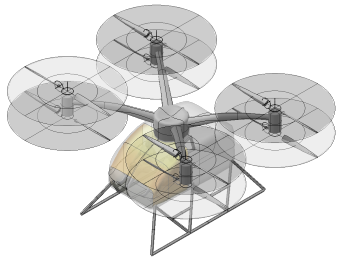
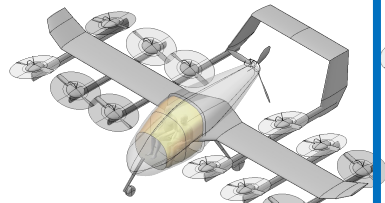
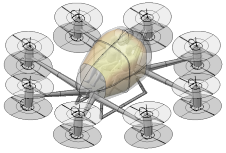
No cyclic rotor control
No gearbox



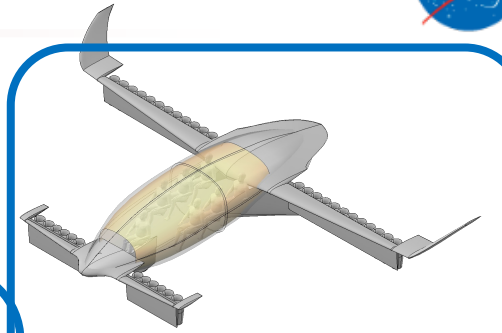
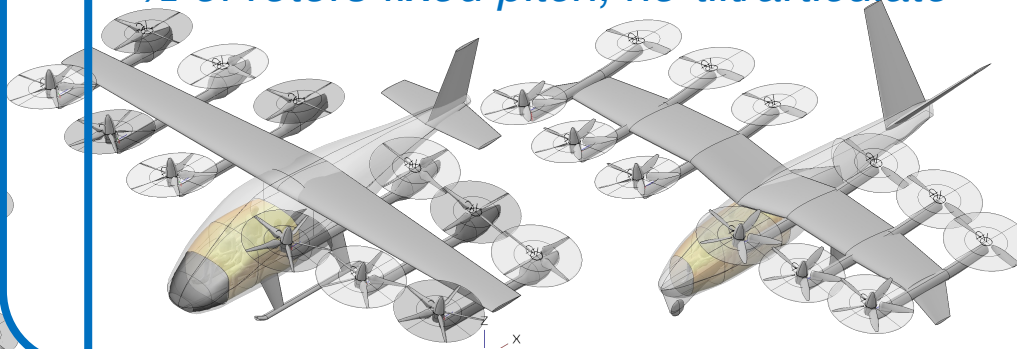
UAM Requirement: Less cost



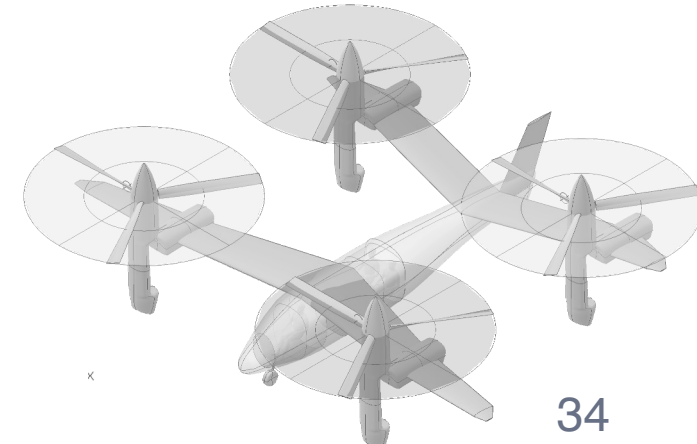
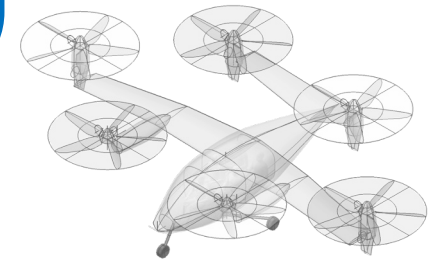
Fixed pitch,
no-articulation,
no-tilt rotors



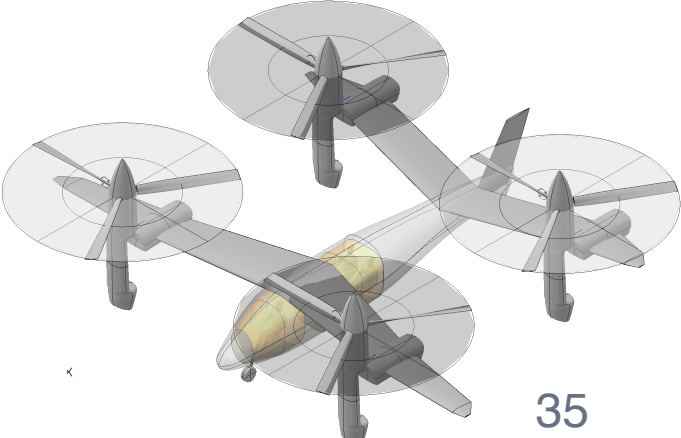
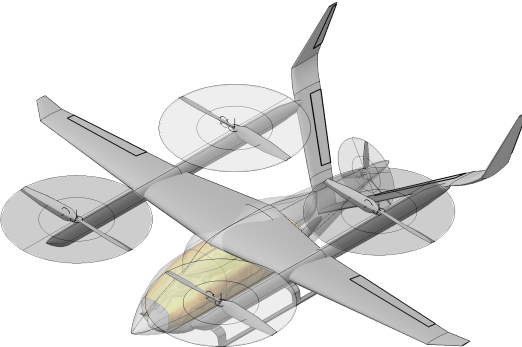
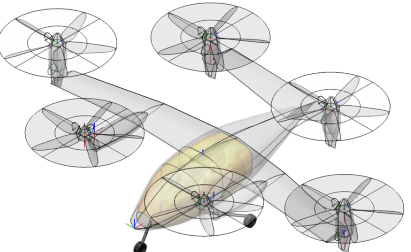
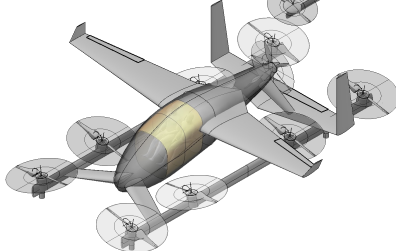
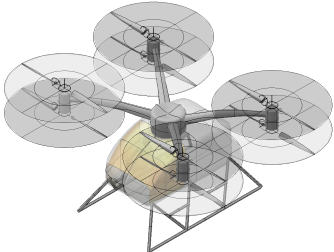
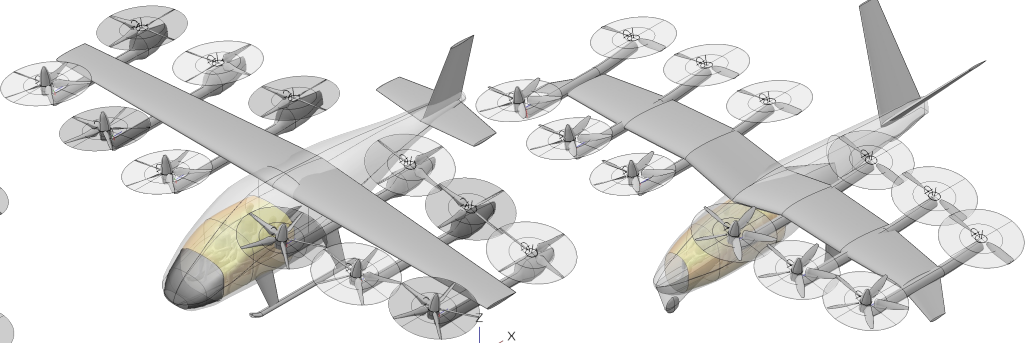
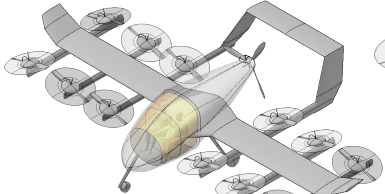
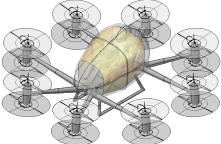
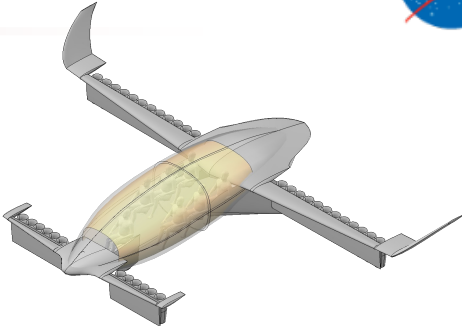
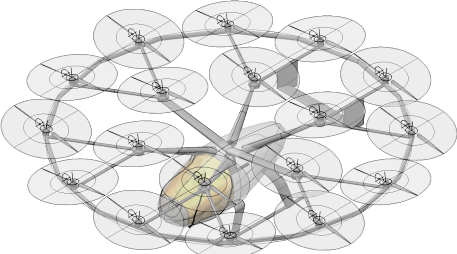
1/2 of rotors fixed pitch, no-tilt/articulate



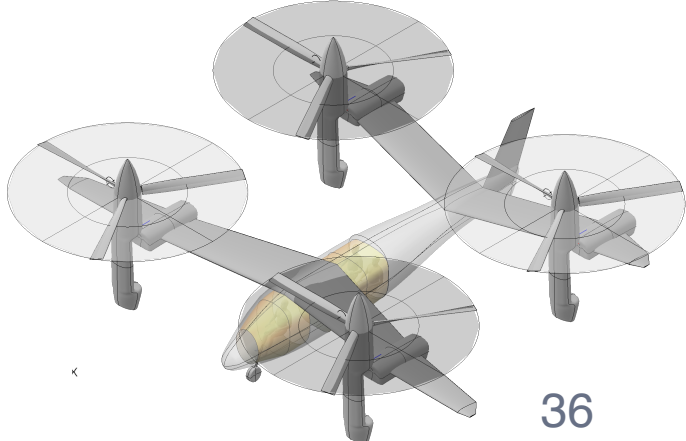
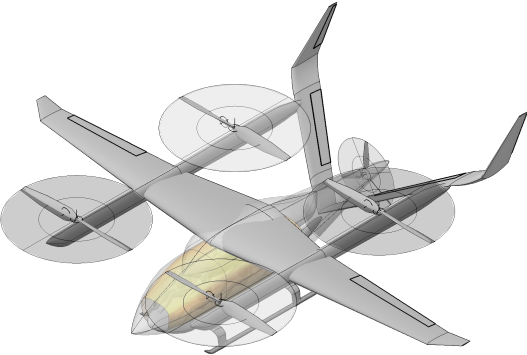
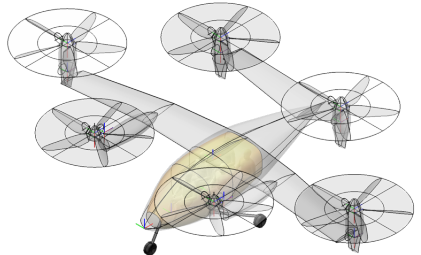
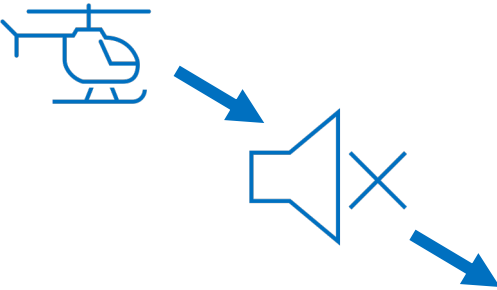
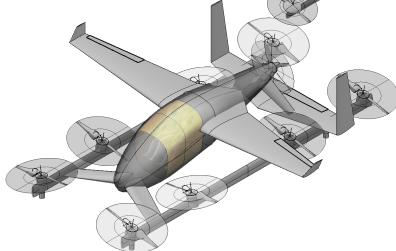
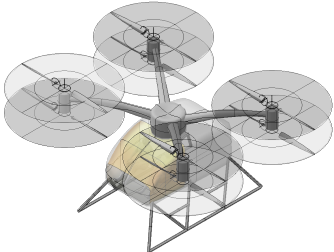
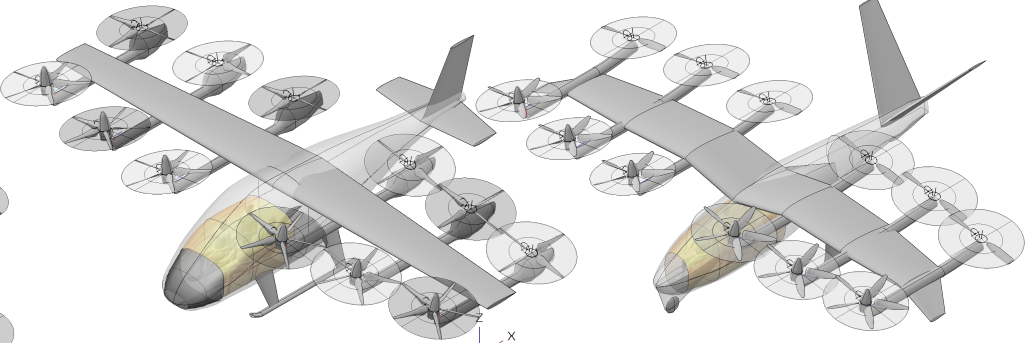
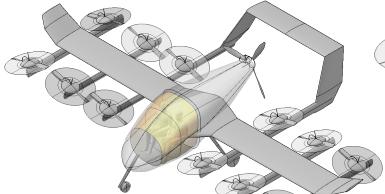
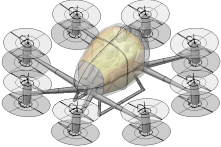
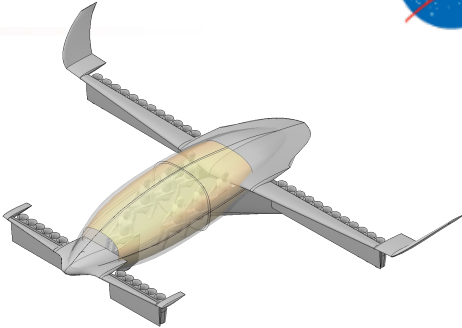
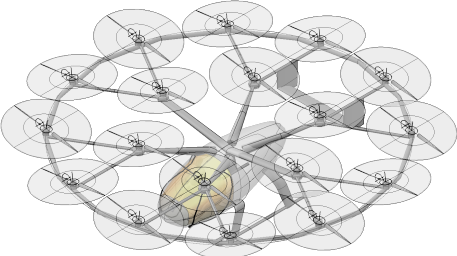
Fixed pitch rotors



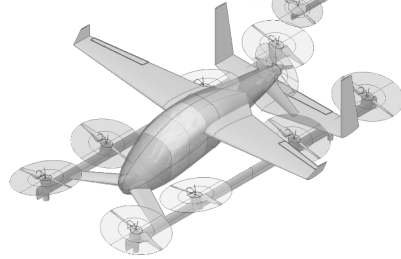
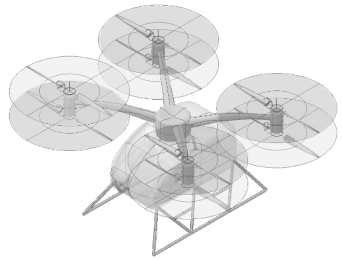
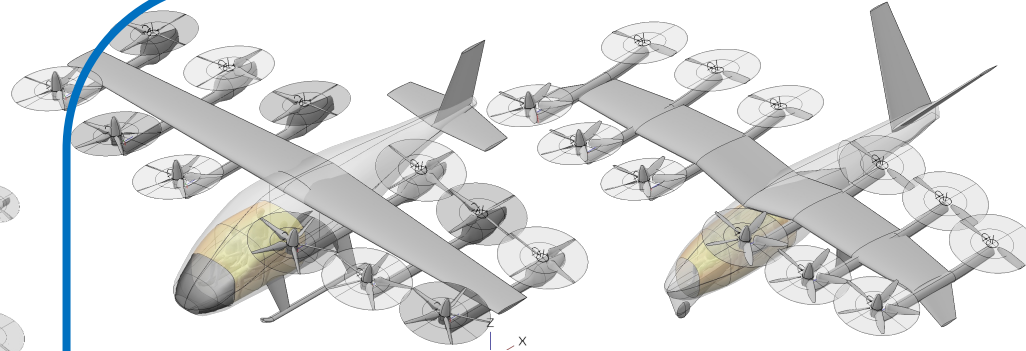
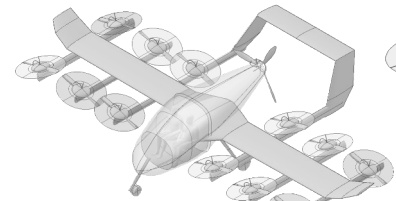
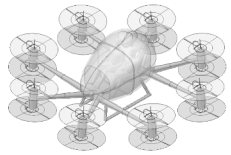
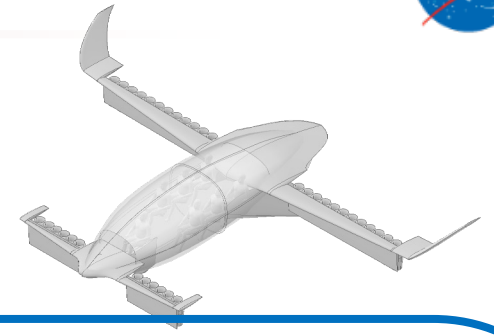
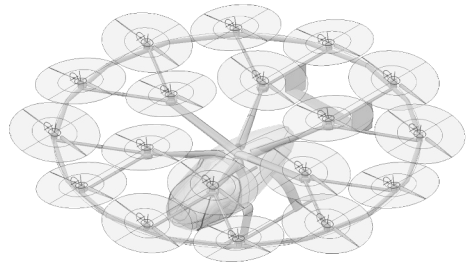
UAM Requirement: Less noise



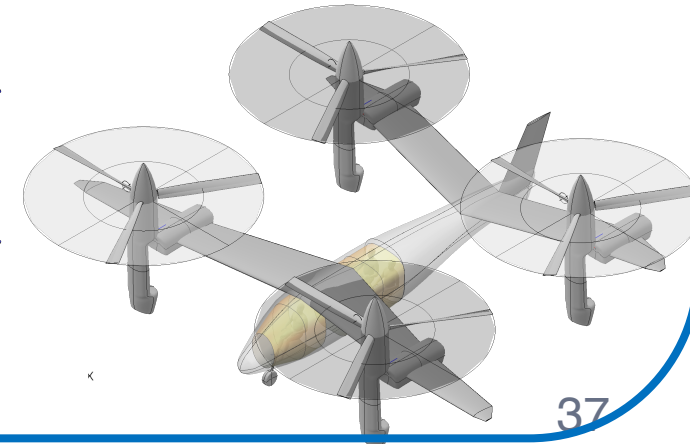
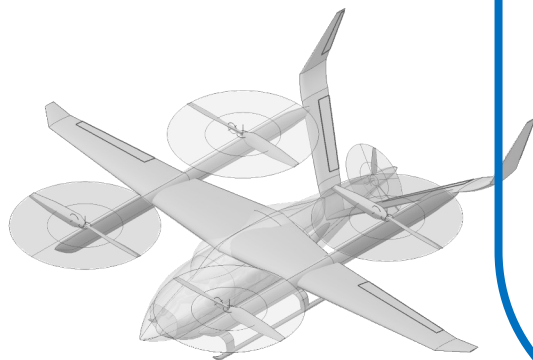
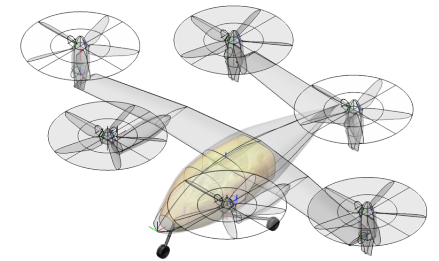
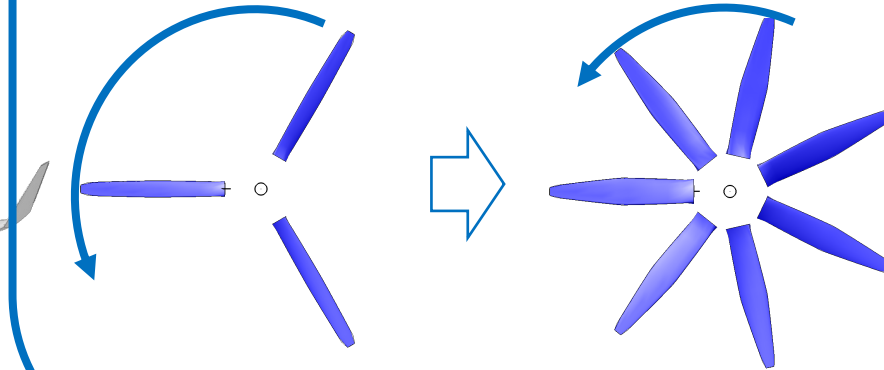
UAM Requirement: Less noise



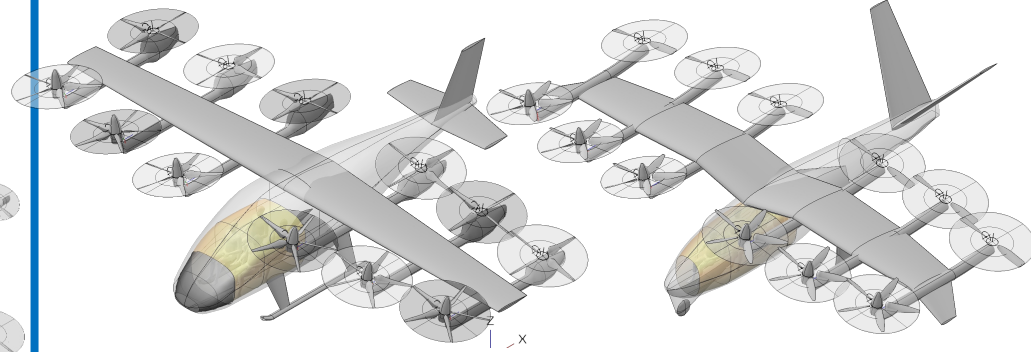
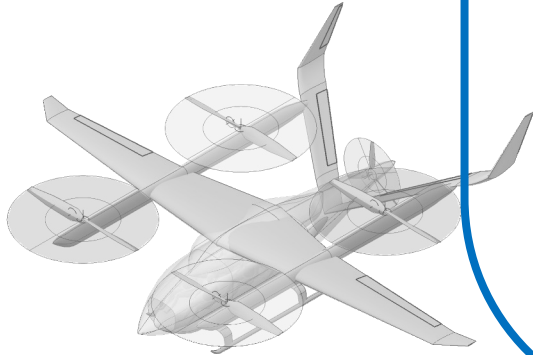
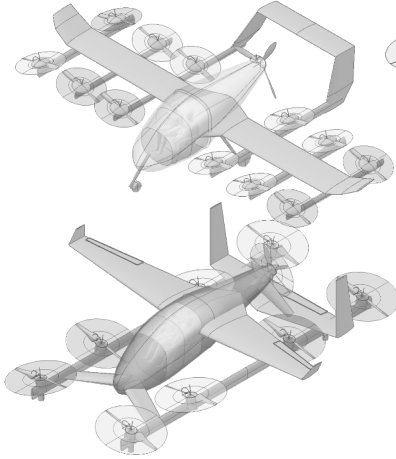
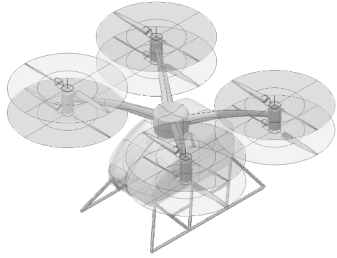
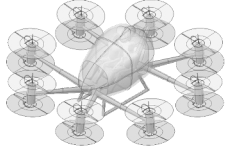
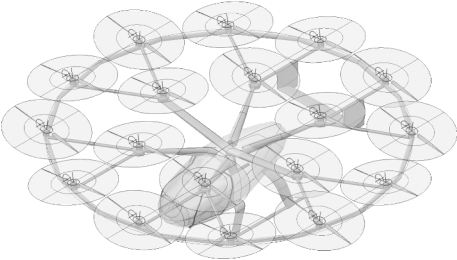
UAM Requirement: Less noise



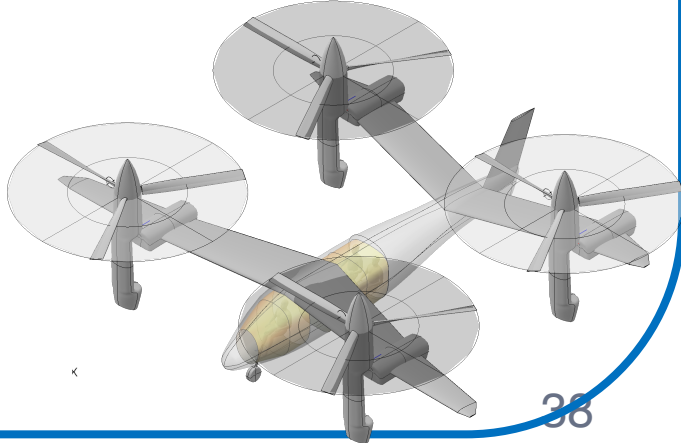
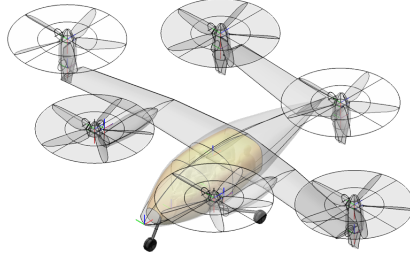
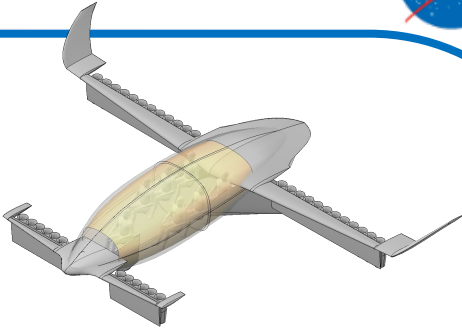
Lower rotor tip speed and more blades in open rotors



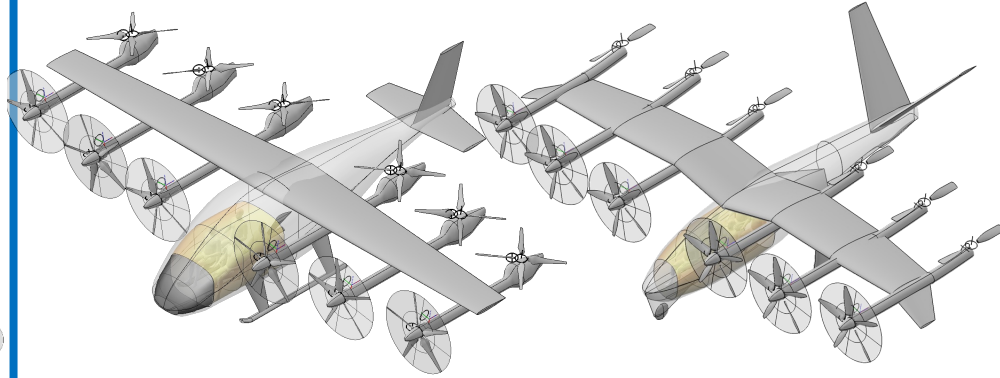
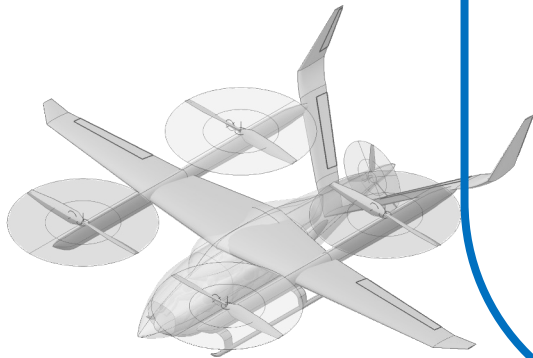
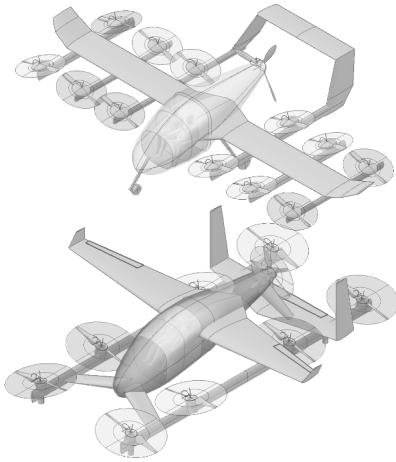
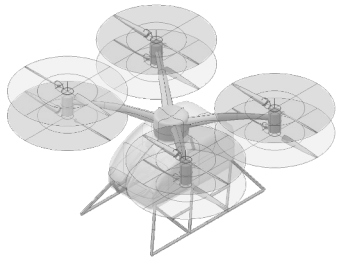
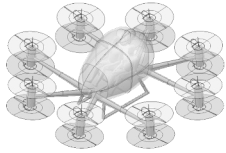
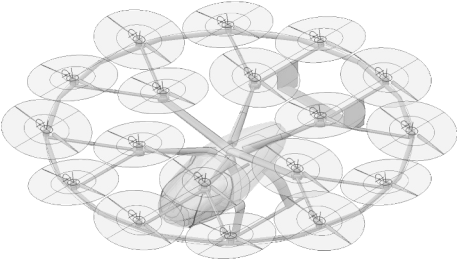
UAM Requirement: Less noise



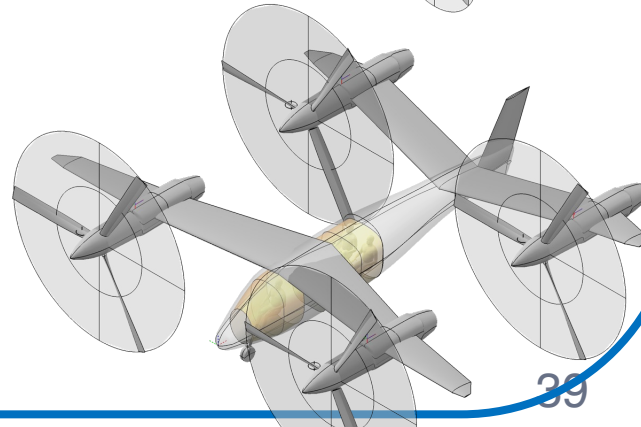
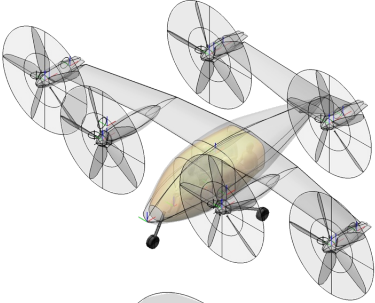
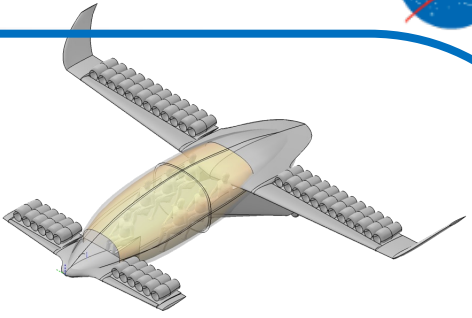
Tilting for cruise



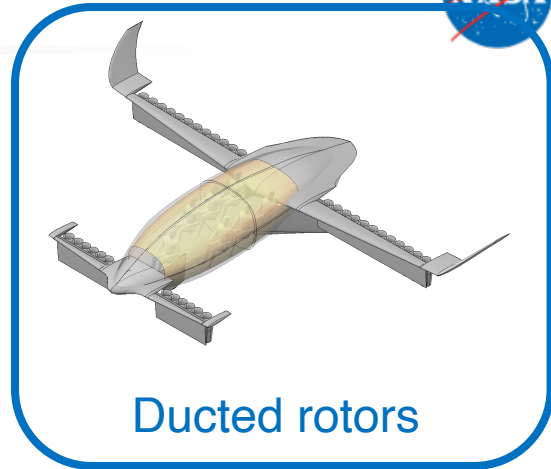
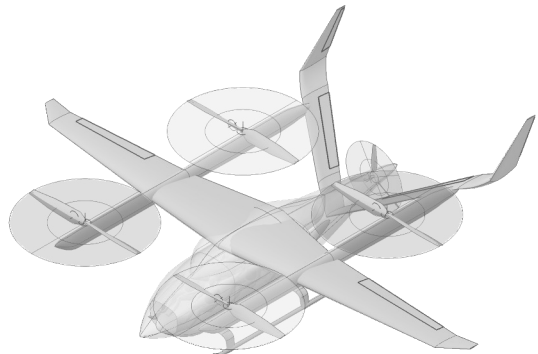
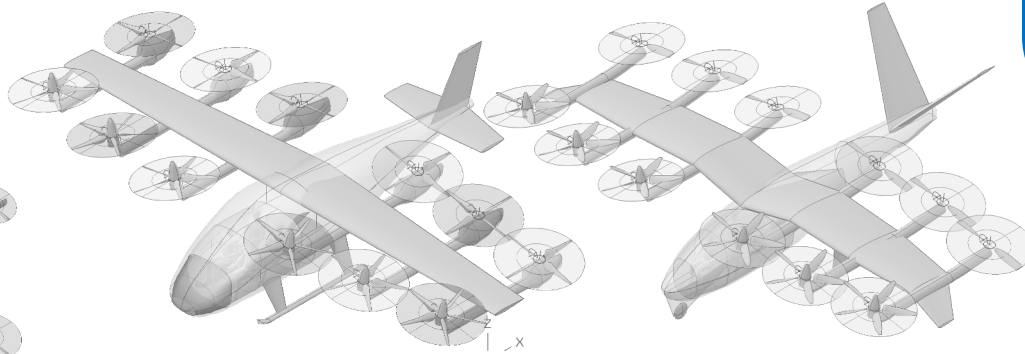
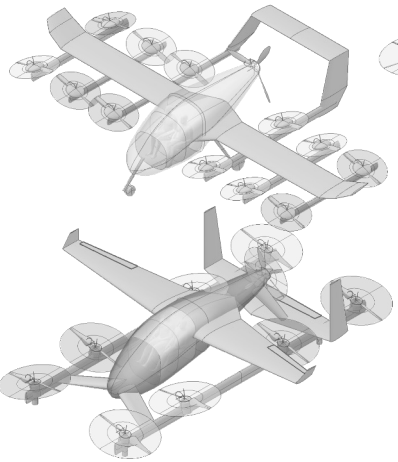
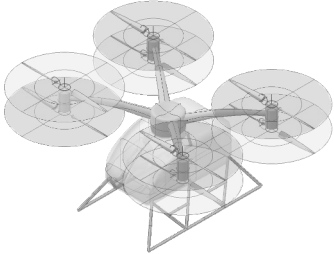
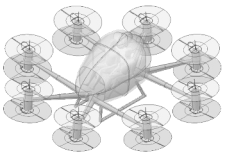
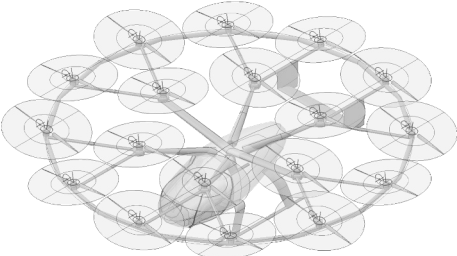
UAM Requirement: Less noise



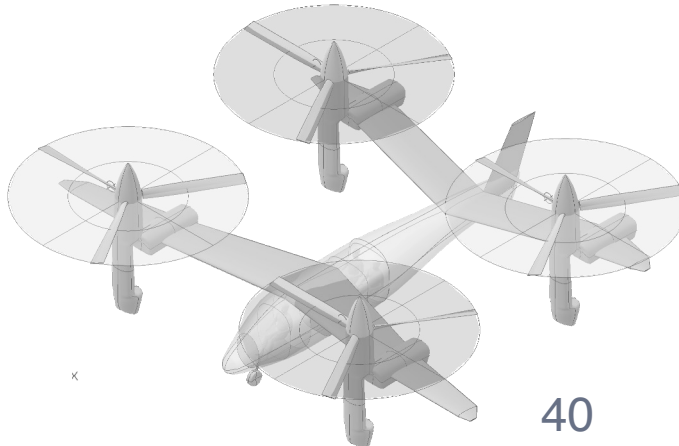
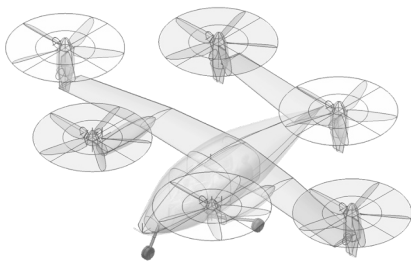
Tilting for cruise



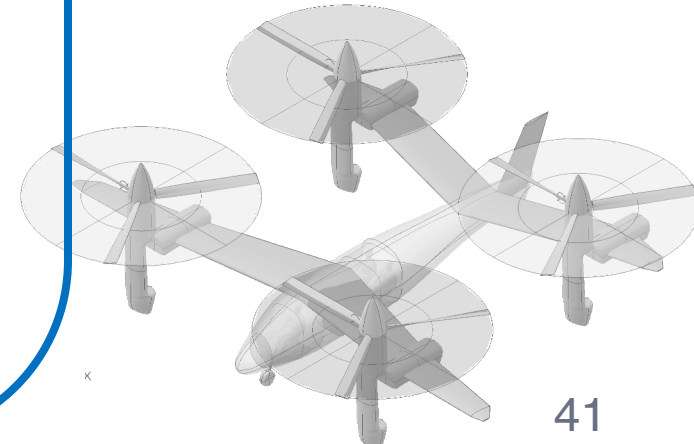
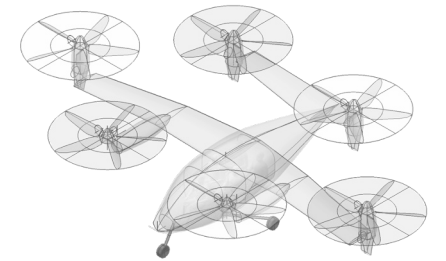
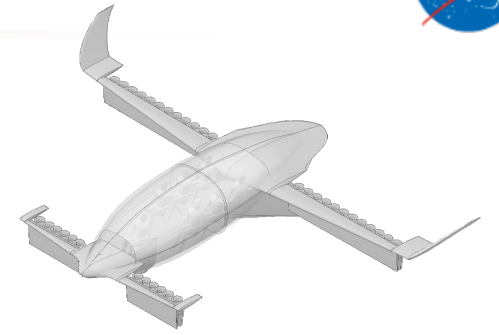
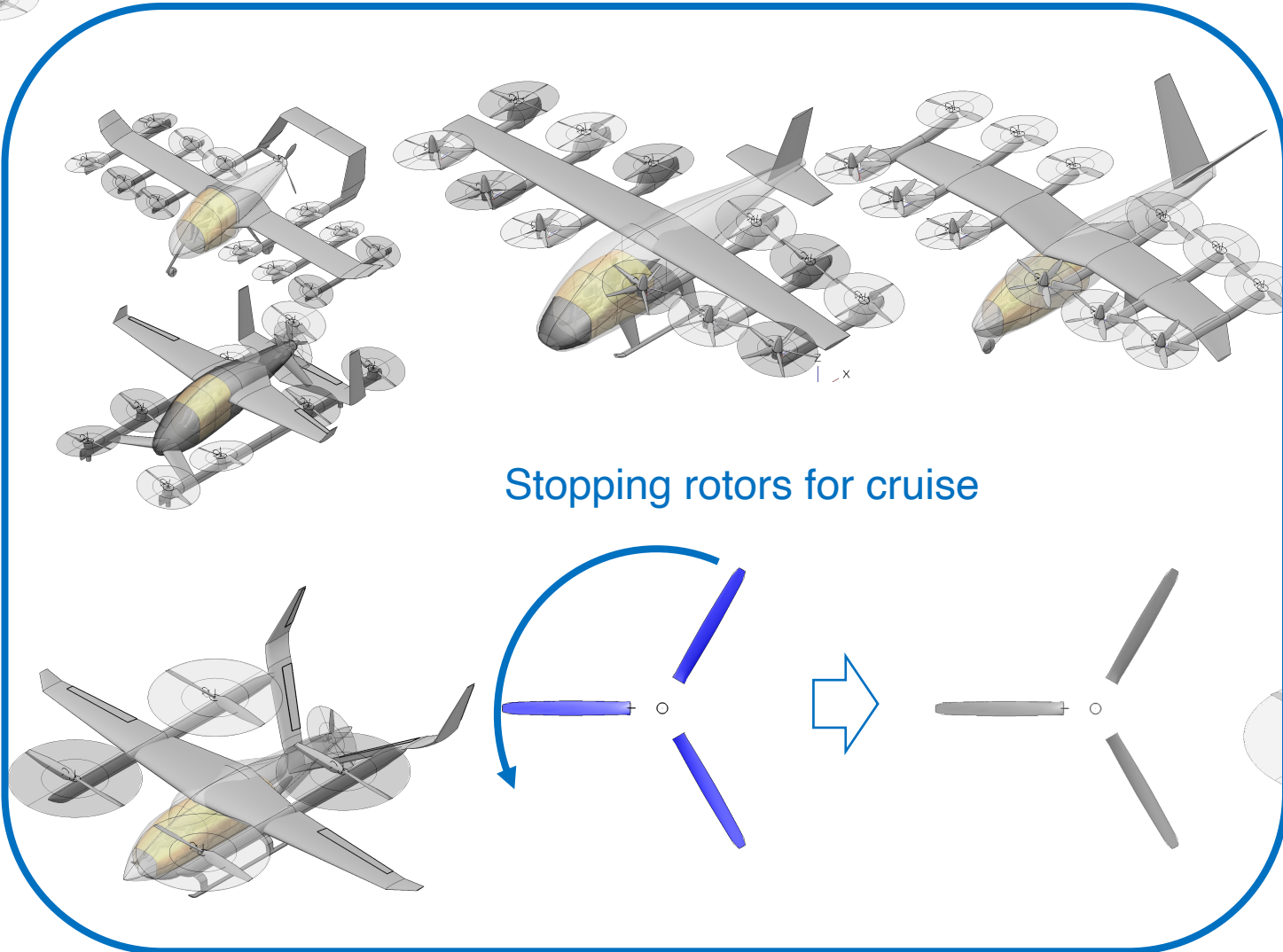
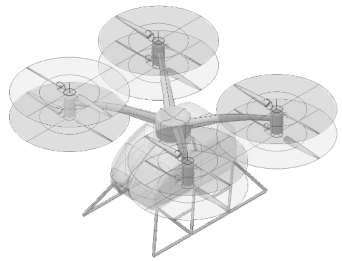
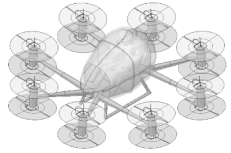
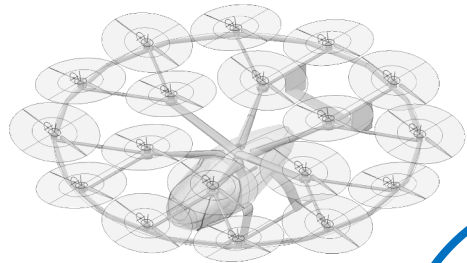
UAM Requirement: Less noise



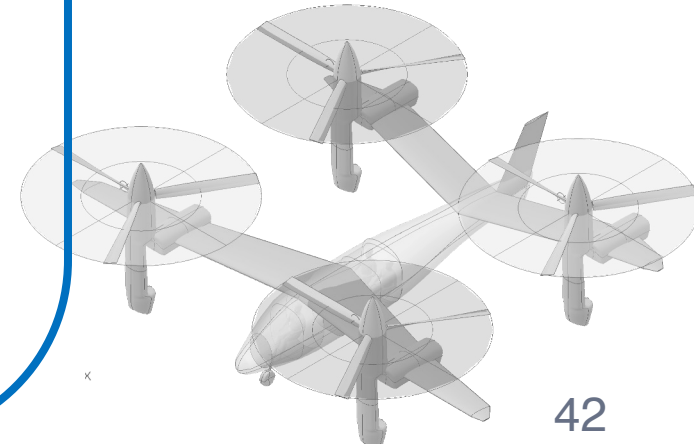
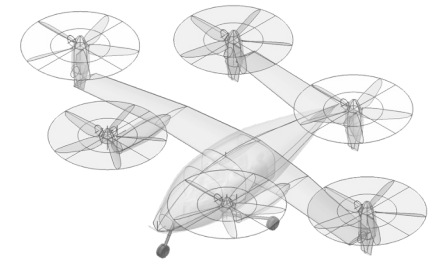
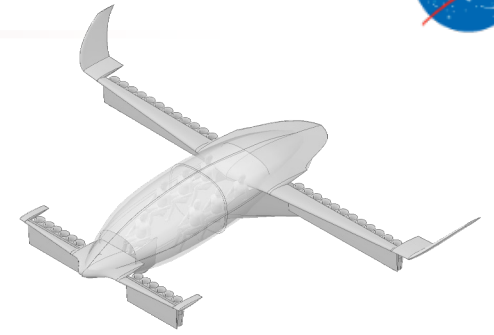
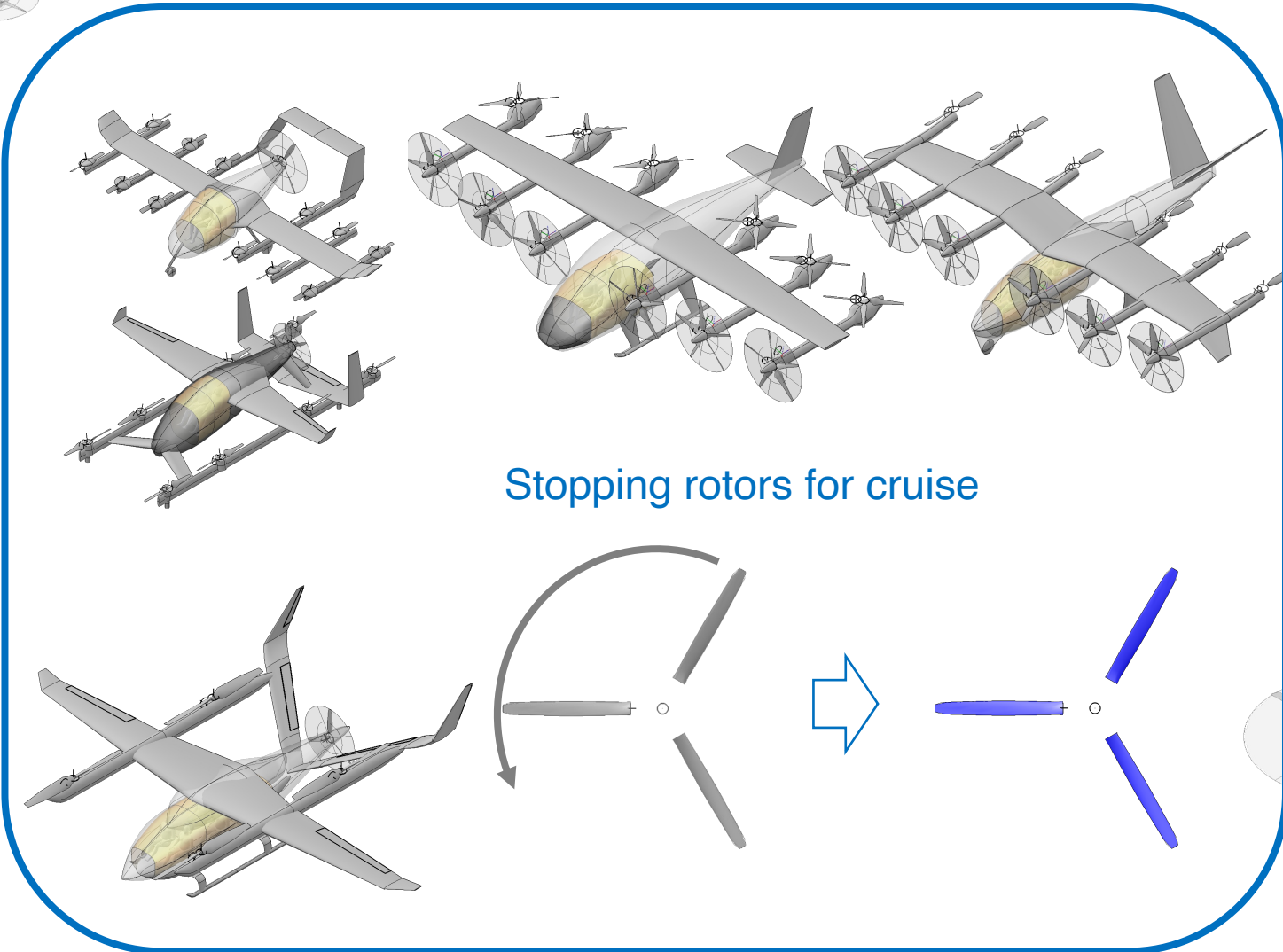
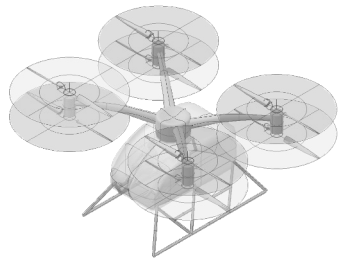
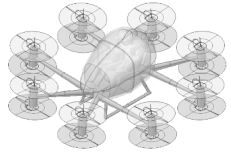
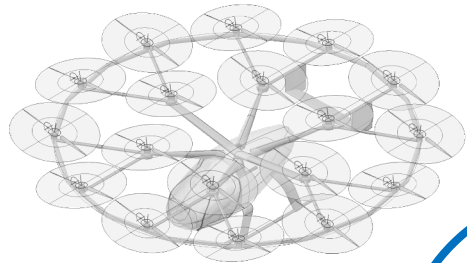
Ducted rotors



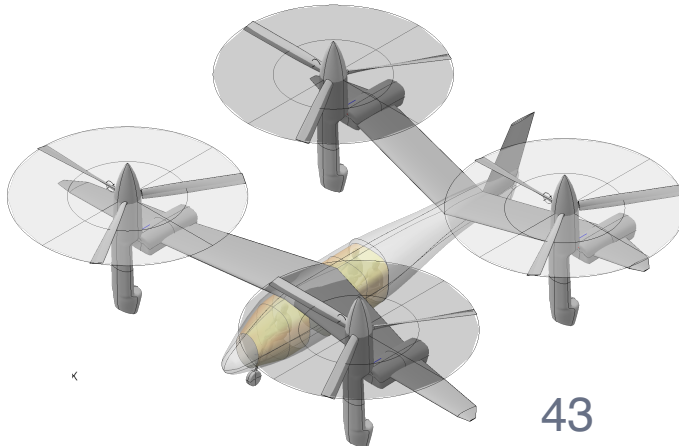
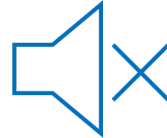
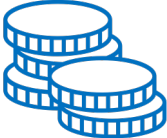
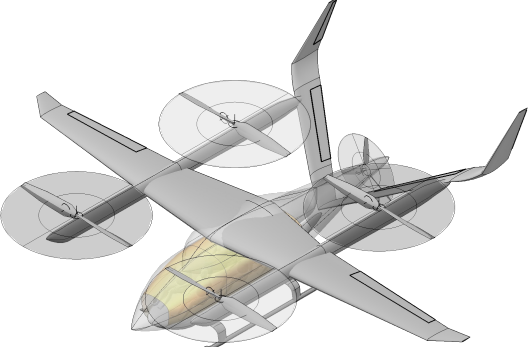
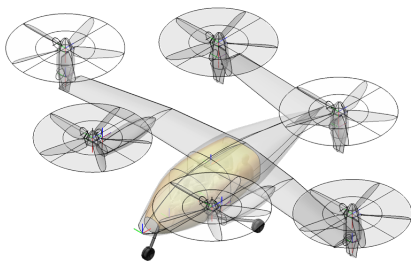
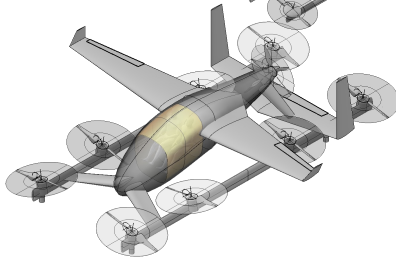
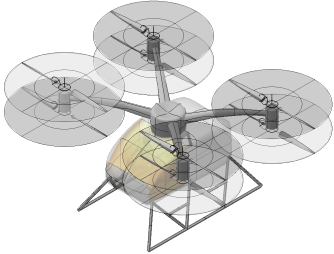
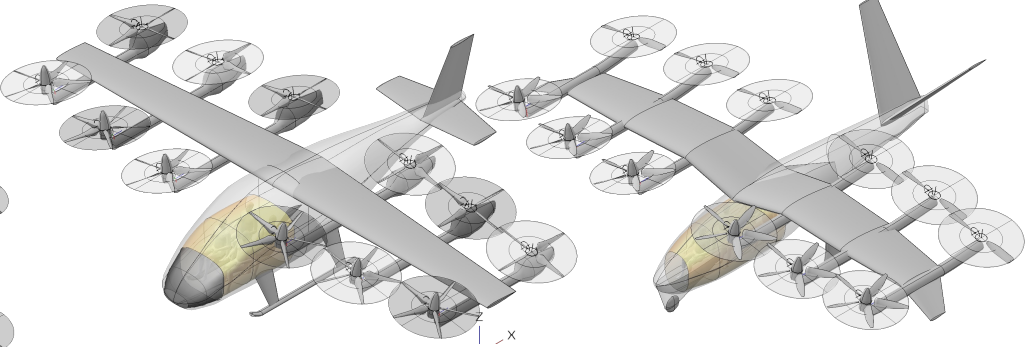
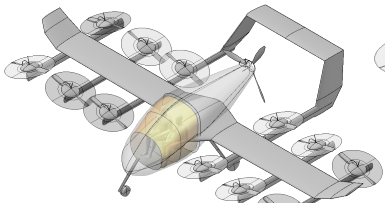
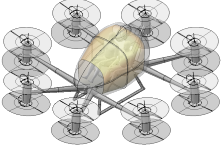
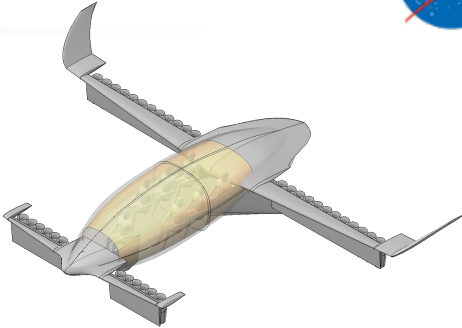
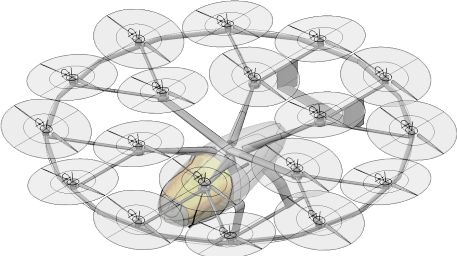
UAM Requirement: Less noise



UAM Requirement: Less noise

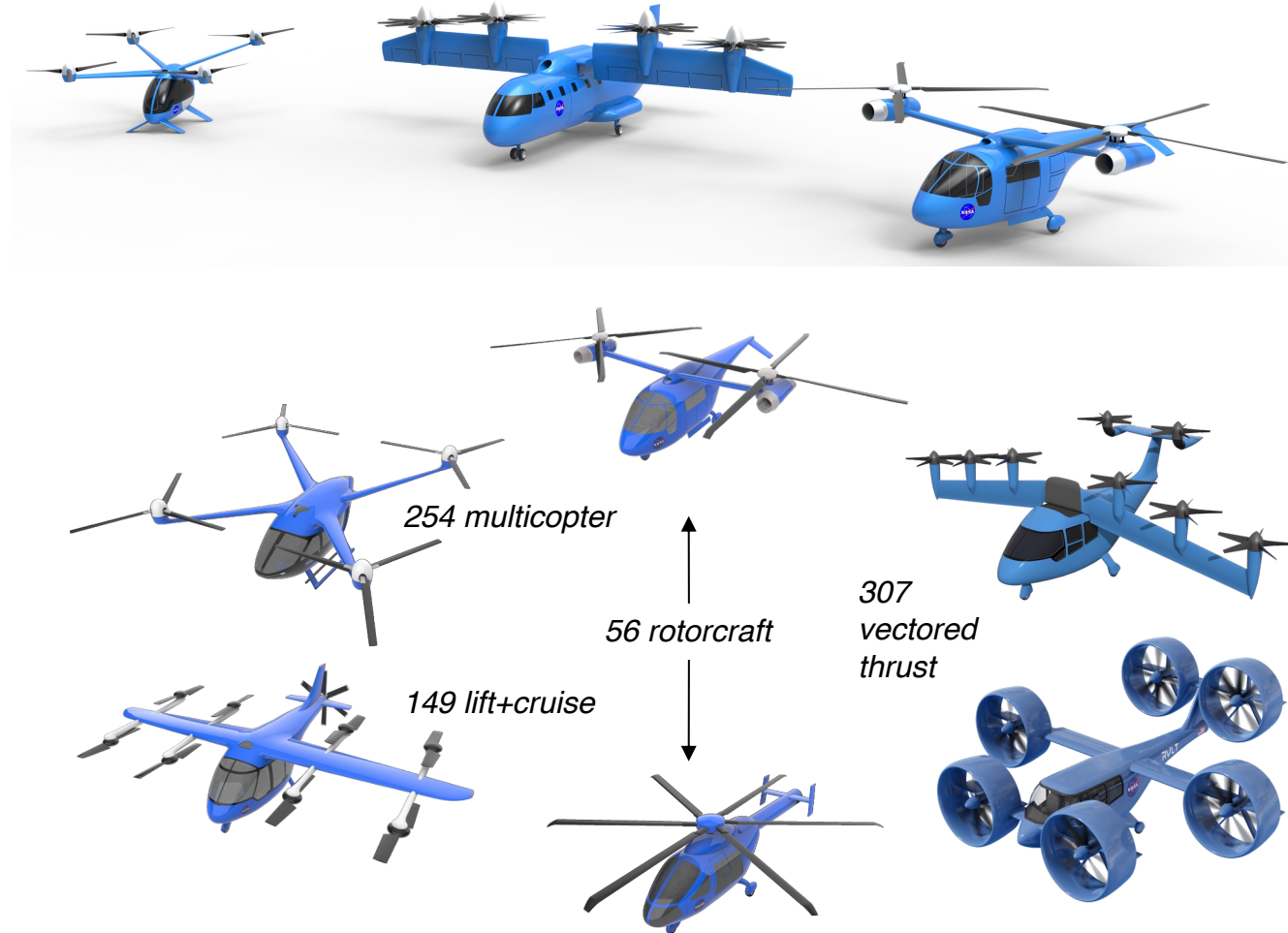


UAM Requirements: More good, less bad



Why we designed the NASA reference vehicles

- Vehicle model inputs and outputs publicly available
 - Discussions can be quantitative
 - Demonstration cases for training
 - Features representative of vehicles
 - Missions and design conditions
 - Margins and reliability requirements
- Focus and guide government research
 - Enable contracted work to be published
 - Assess technology payoff
 - Guide tool development
 - Scope validation tests
- No plan or desire to build the vehicles

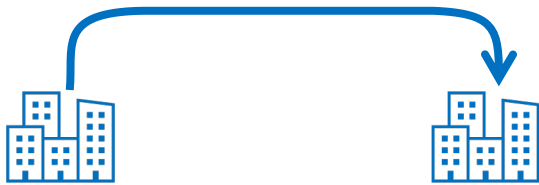


NASA Reference Vehicles quantify relative merits of solutions



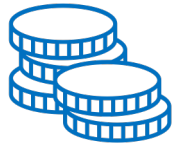
More convenient

Vertical takeoff and landing
Flights of 50 - 100 km
Many operations per hour
High speed



Less Cost

Cheap electric energy
Cheap electric motors
Cheap pilots or no pilots
Simple rotors
No gearboxes
No swashplates
Automotive-scale production



More safe

Many rotors for redundancy
Electrical redundancy
Electrical isolation
Robust fly-by-wire
Autorotation
Mechanical redundancy
Energy margins



Less noise

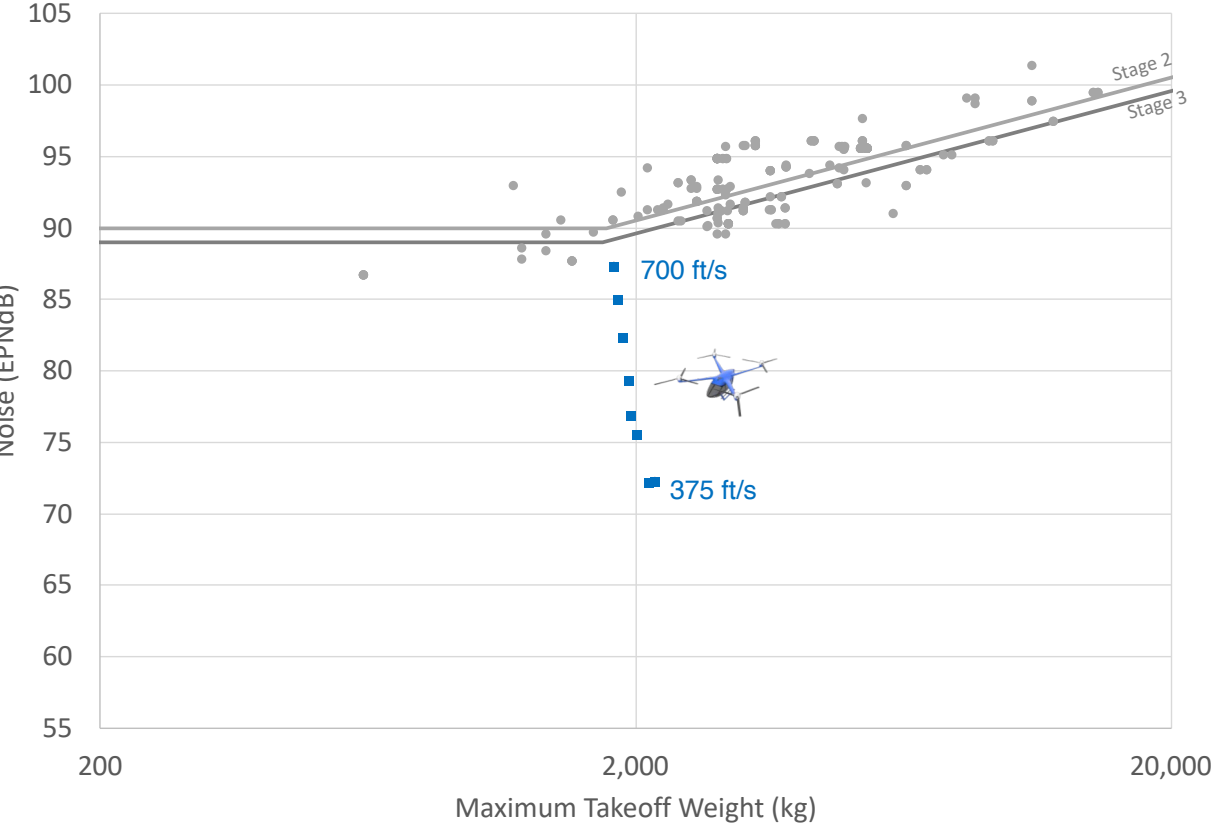
Slower rotor tip speeds
Tilting rotors
Stopping rotors
More blades
Tip shapes
Higher-harmonic rotor control



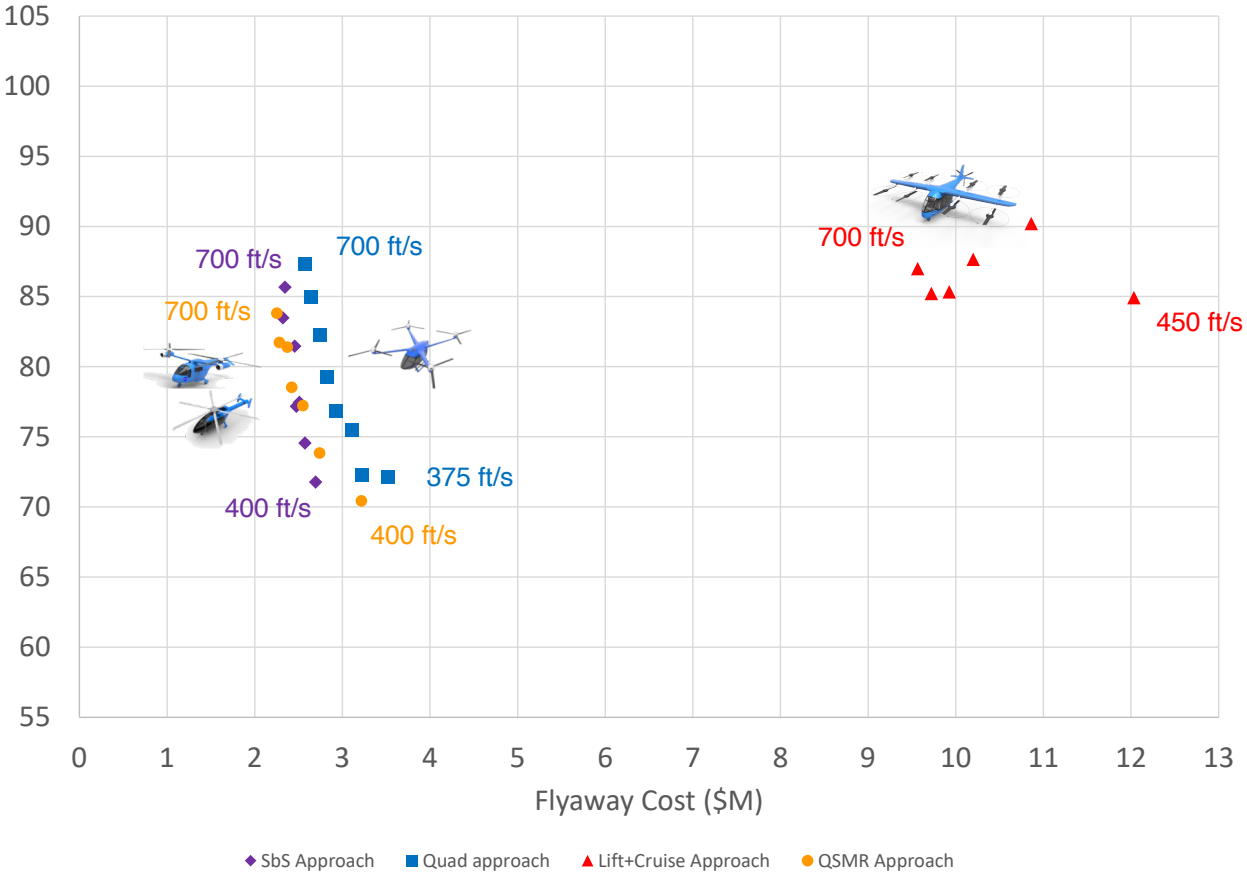
Quantifying the cost of reducing noise via tip speed reduction



Approach



Cost vs Approach Noise, Tip Speed Varies

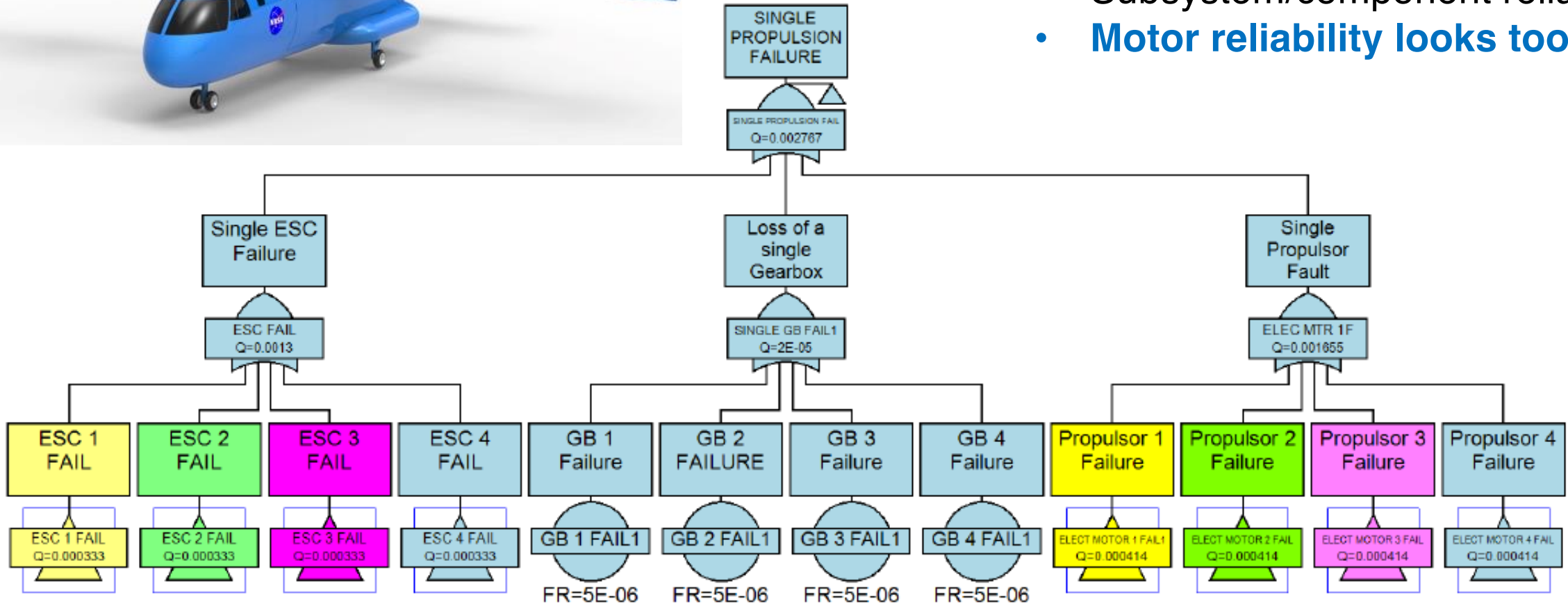


◆ SbS Approach ■ Quad approach ▲ Lift+Cruise Approach ● QSMR Approach

Quantifying the safety of UAM vehicles with varying architectures and components



- Part of rigorous systems engineering
- A lot to learn about electric systems
 - Probabilistic assessments
 - Subsystem/component reliability
 - **Motor reliability looks too low**



NASA Research Areas for UAM eVTOL Vehicles



PROPULSION EFFICIENCY

light, efficient, high-speed electric motors
 power electronics and thermal management
 efficient powertrains
 high power, lightweight battery
 light, efficient small turboshaft engine

SAFETY and AIRWORTHINESS

component reliability and life cycle
 crashworthiness: airframe, occupant, battery
 bird strike
 electric motor reliability assessment
 propulsion system failures
 FMECA (failure mode, effects, and criticality analysis)
 high voltage operational safety
 high voltage protection devices

OPERATIONAL EFFECTIVENESS

Ops in moderate to severe weather
 passenger acceptance/ ride quality
 disturbance rejection (control bandwidth, control design)
 cost (purchase, maintenance, DOC)

PERFORMANCE

aircraft optimization
 rotor shape optimization
 hub and support drag minimization
 airframe drag minimization

ROTOR-ROTOR INTERACTIONS

performance, noise, handling qualities, aircraft arrangement
 vibration and load alleviation

NOISE AND ANNOYANCE

low tip speed
 rotor shape optimization
 flight operations for low noise
 aircraft arrangement/ interactions
 cumulative noise impacts from fleet ops
 metrics and requirements
 human response to noise
 active noise control
 cabin noise
 electric motor noise

STRUCTURE AND AEROELASTICITY

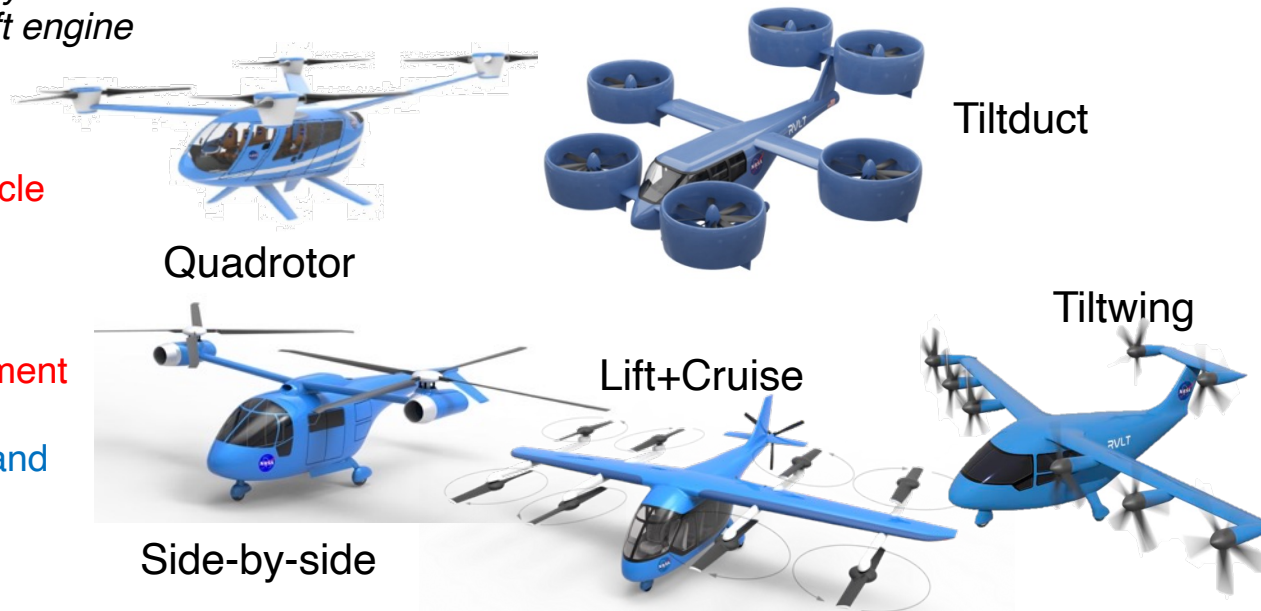
crashworthiness
 durability and damage tolerance
 structurally efficient wing and rotor support
 rotor/airframe stability
 high-cycle fatigue

AIRCRAFT DESIGN

weight, vibration
 handling qualities
 active control

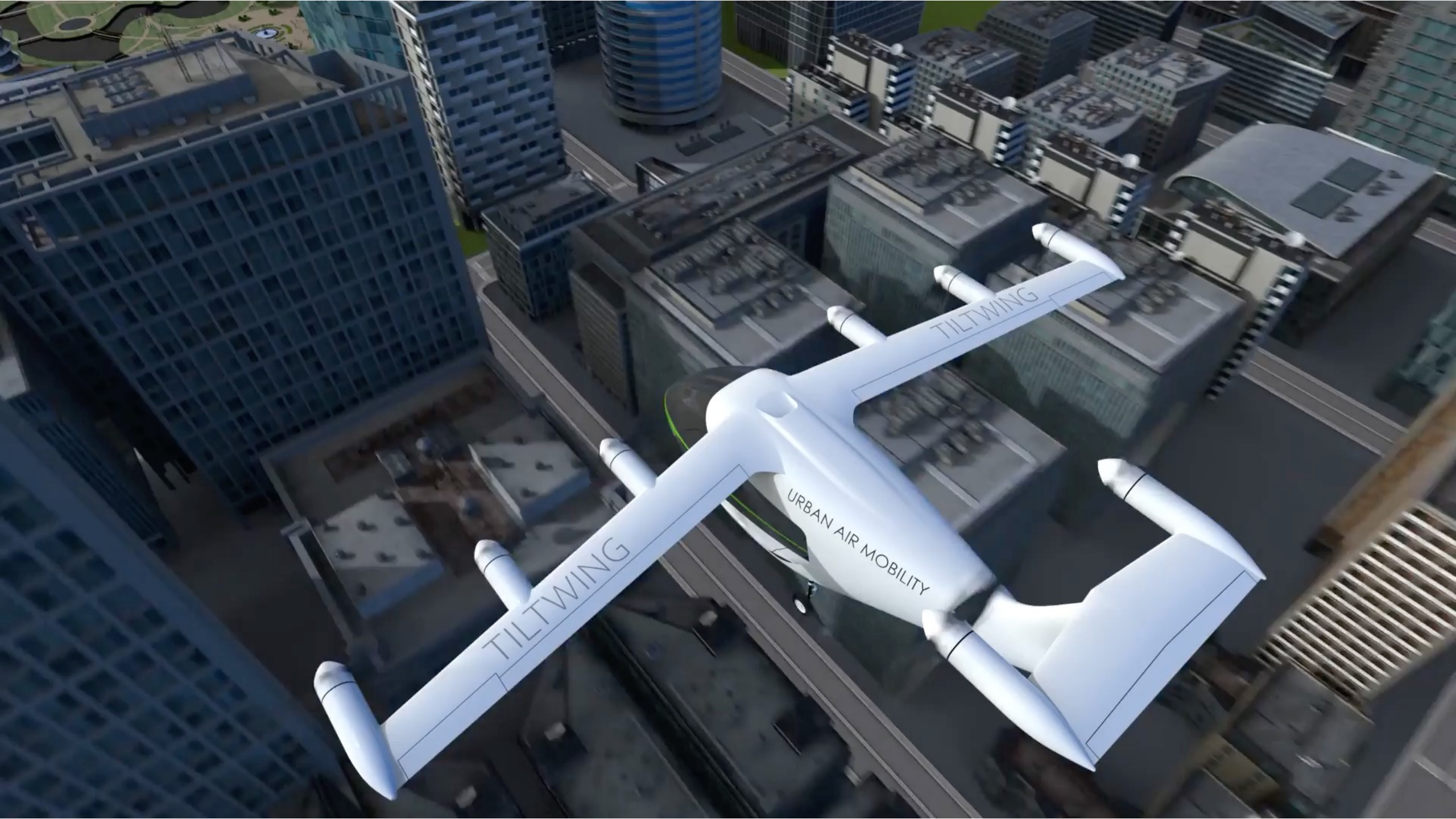
ROTOR-WING INTERACTIONS

conversion/transition
 interactional aerodynamics
 flow control



Red = primary NASA research area

Blue = secondary NASA research area



TILTWING

TILTWING

URBAN AIR MOBILITY

