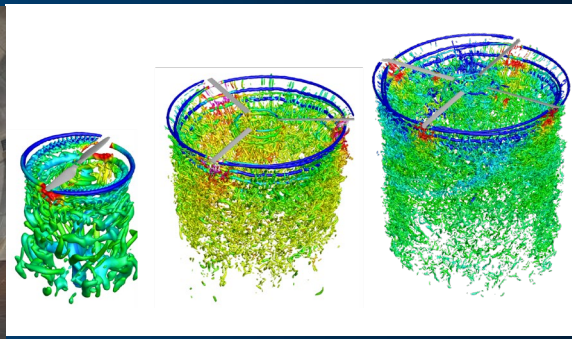


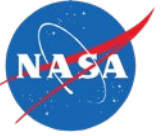



RotorGen: A Simplified Structured Grid Generation Program for High-Fidelity Rotor CFD Simulation

Nicholas Peters, Carlos Pereyra

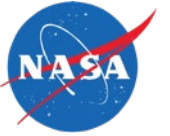
AIAA 2024 Overset Grid Symposium
Dayton Ohio, October 7-10, 2024



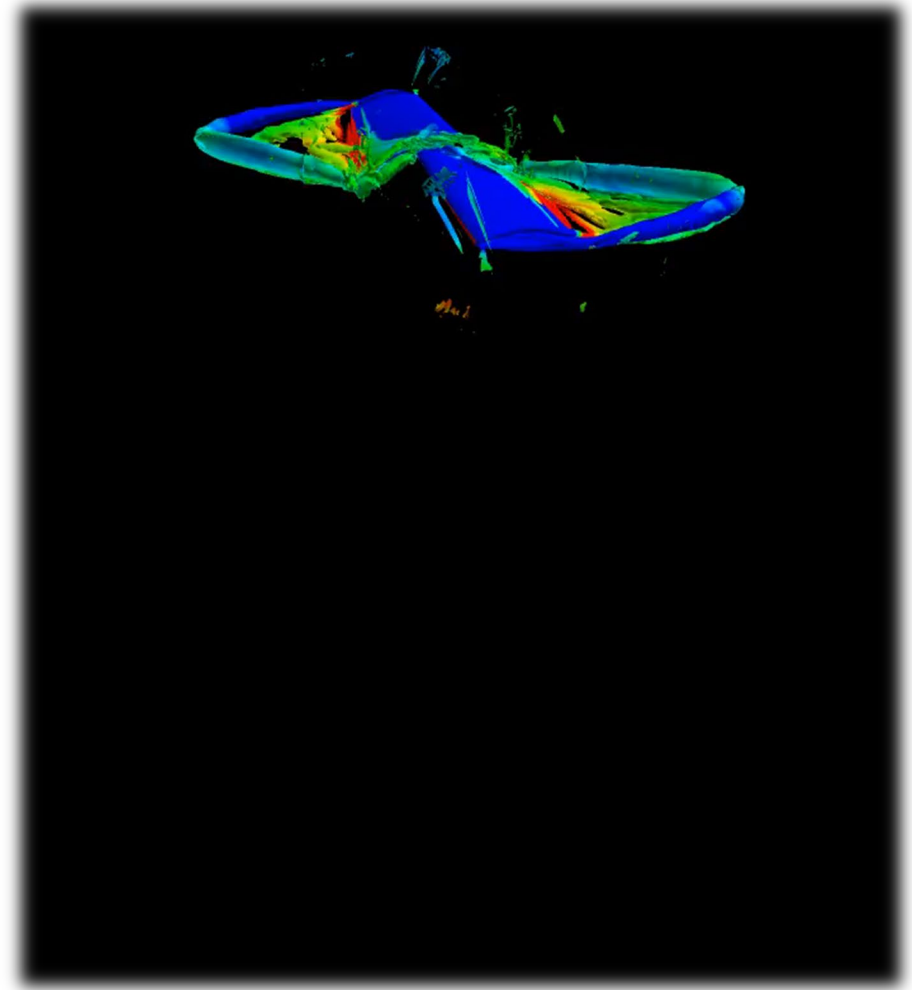


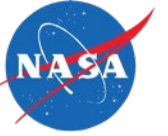
- 
- **Motivation**
 - **Overview of RotorGen**
 - **Preliminary Validation Work**
 - **Concluding Remarks**

Challenges for Rotorcraft-based CFD Simulation in Conceptual Design



- **There exists challenges in leveraging CFD solutions of rotors in early-stage conceptual design.**
 - Limited usage of CAD rotor representations
 - Coupling to structural and trim solvers
 - Rapidly evolving design space
- **Scriptable grid generation programs exist to complete parametrized CFD studies.**
 - These scripts require significant user expertise to generate
- **There is a need for a high-level, simplified tool for parsing conceptual design-level rotor definitions and rapidly generating high-fidelity CFD cases.**

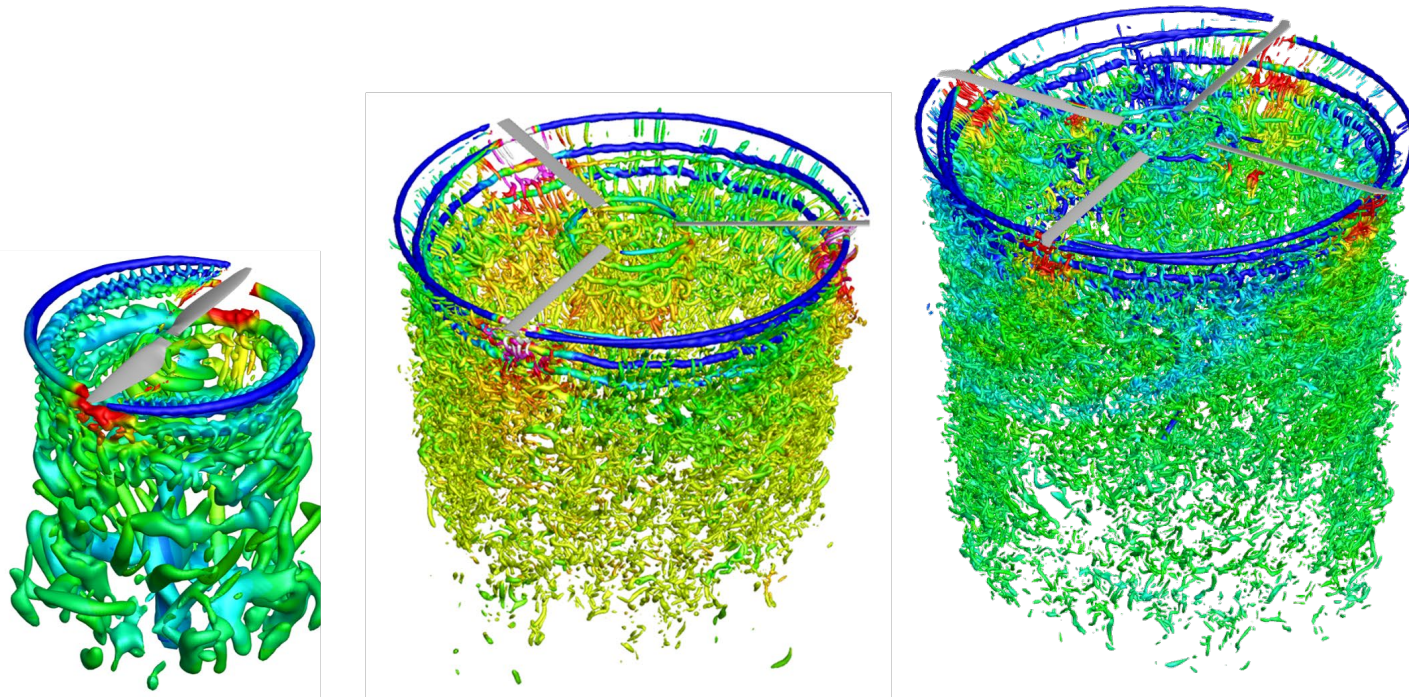




- **Motivation**
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Objectives of RotorGen

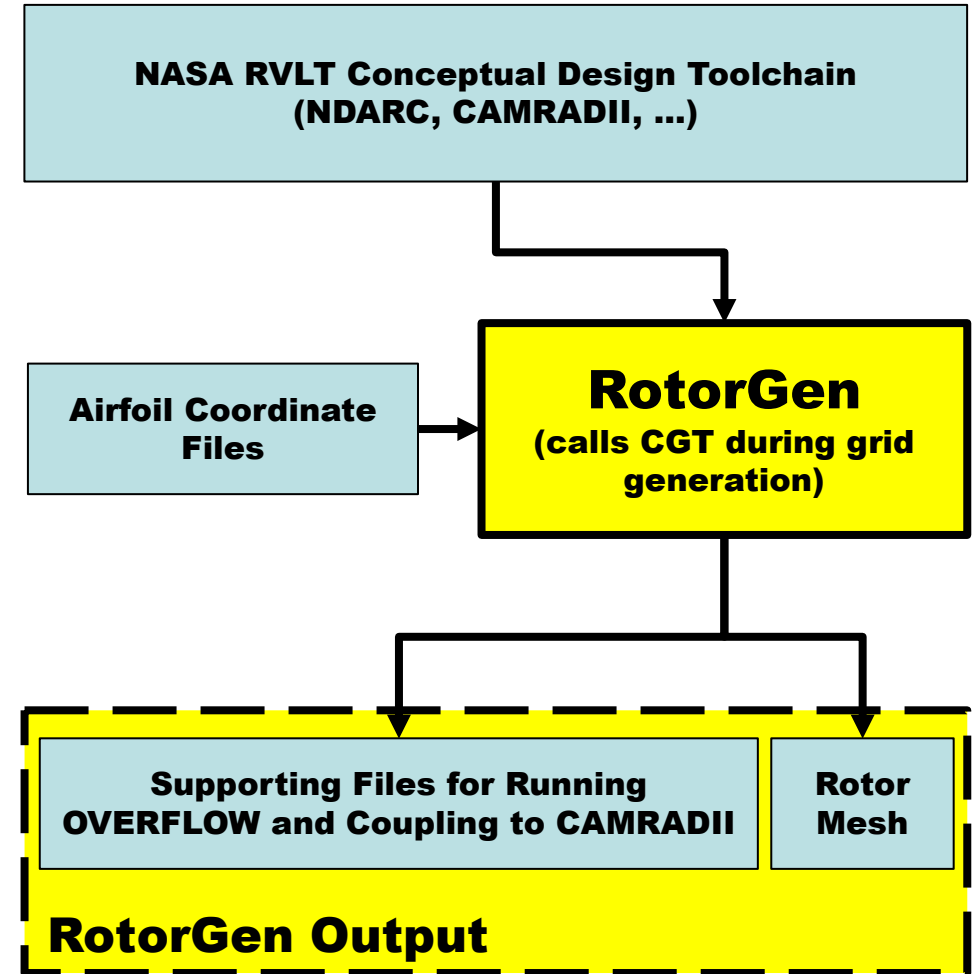
- The objective of RotorGen is to automatically generate OVERFLOW CFD cases based on conceptual design-level rotor definitions.
- In pursuit of this objective, RotorGen promotes the use of high-fidelity solutions among a broader range of rotorcraft-based conceptual design groups.





High Level Workflow with RotorGen

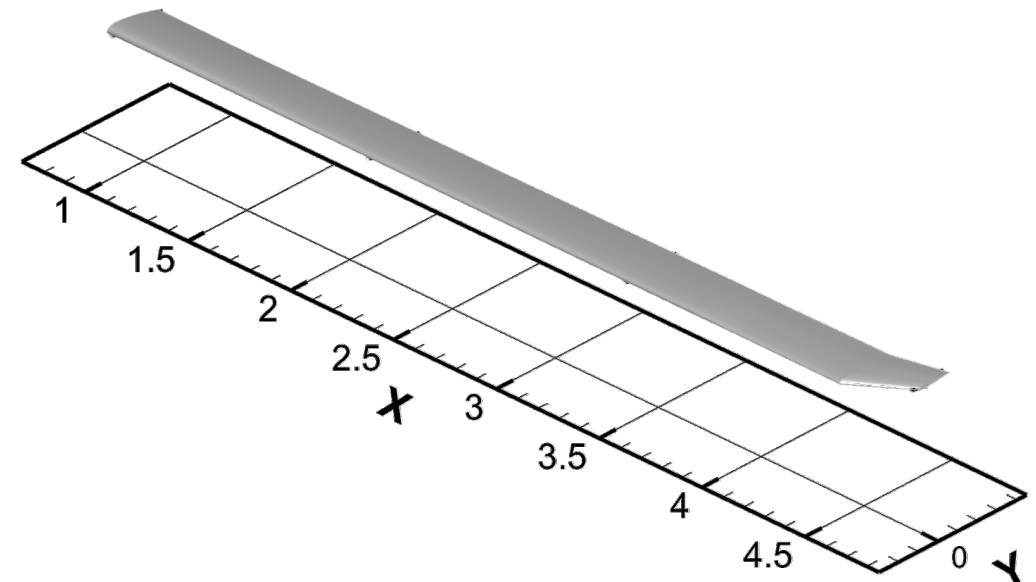
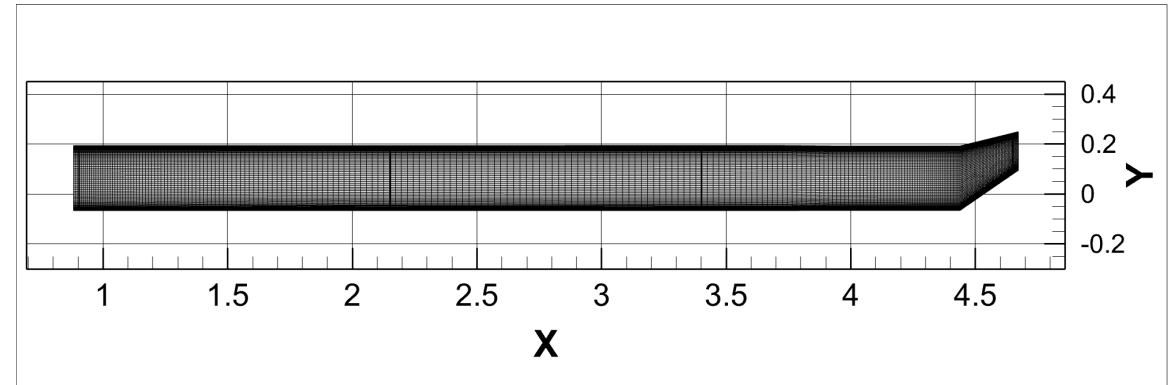
- The NASA RVLT Conceptual Design Toolchain (NDARC, CAMRADII, ...) is utilized to identify a limited number of interesting cases.
- Conceptual design-level definitions for rotor geometric and operating conditions are entered into a single RotorGen input file.
- Chimera Grid Tools (CGT) generates multiple overset volume grids to model the rotor.
- Time to submit CFD cases reduced from several days to a few seconds.
 - Reduce required expertise to run simulation



Defining the Blade Planform

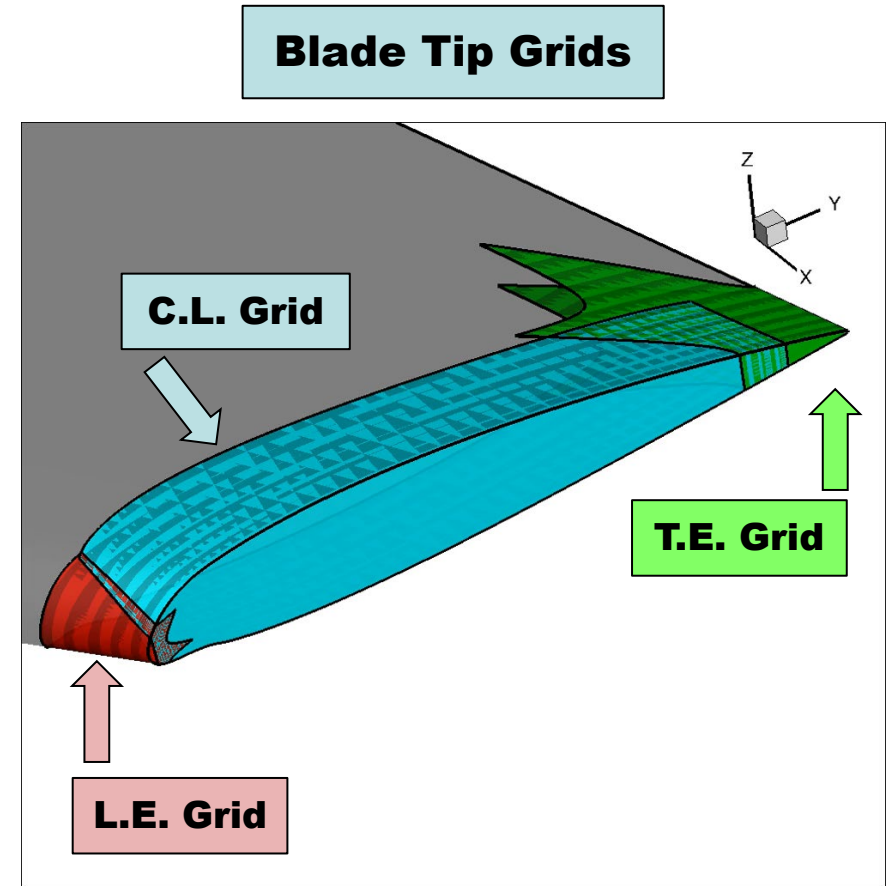
- **Blade planform definitions are provided by the user, detailing geometry.**
 - Definitions define blade spanwise chord, twist, sweep, droop, ...
 - Defines blade $\frac{1}{4}$ chord line
- **Users provide a specific set of airfoil profiles.**
- **User is further allowed detailed control of spanwise mesh refinement.**
 - Varying refinement at user-specified radial stations

Blade Planform



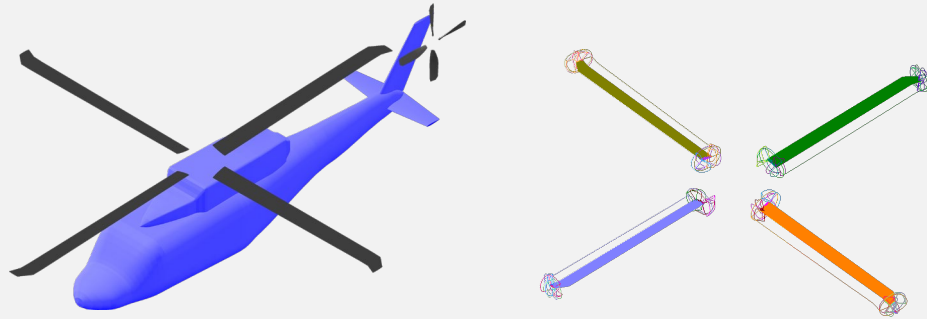
Defining the Blade Tip/Root Cap

- **Tip cap, overset grid generation remains a notoriously rigorous task.**
 - Often a significant source of time spent in grid generation, particularly among non-expert users
- **To maximize the robustness of cap grid generation, three overset grids are used.**
 - Leading Edge (L.E.) Grid
 - Center Line (C.L.) Grid
 - Trailing Edge (T.E.) Grid
- **The splitting of this cap grid has thus far provided sufficient robustness in generating a broad range of rotor tip caps.**

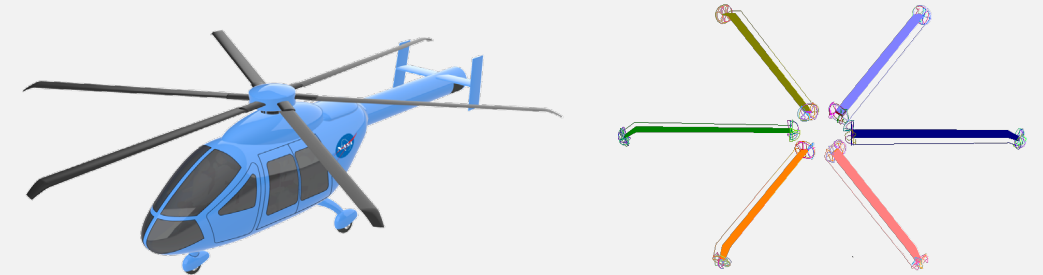


Rapid Rotor Grid Generation Examples

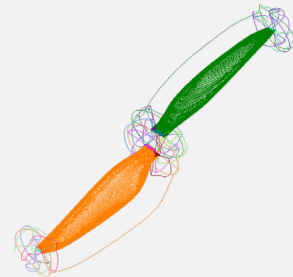
1:4.71 Scaled S-76



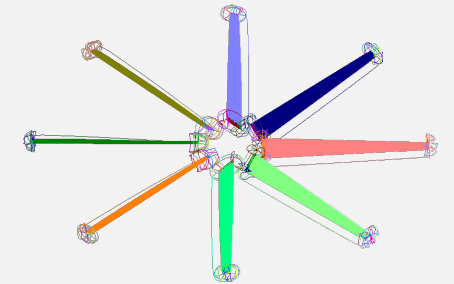
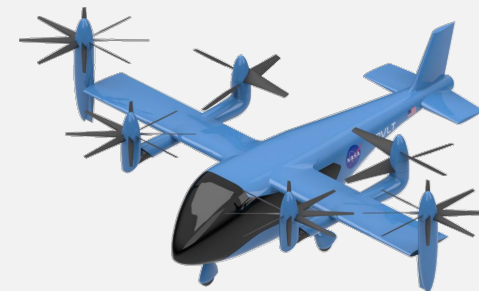
NASA Urban Air Mobility Reference Vehicle (Quiet Single Main Rotor)



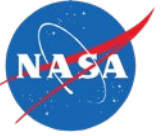
SUI Endurance



NASA Urban Air Mobility Reference Vehicle (Multi-Tiltrotor)



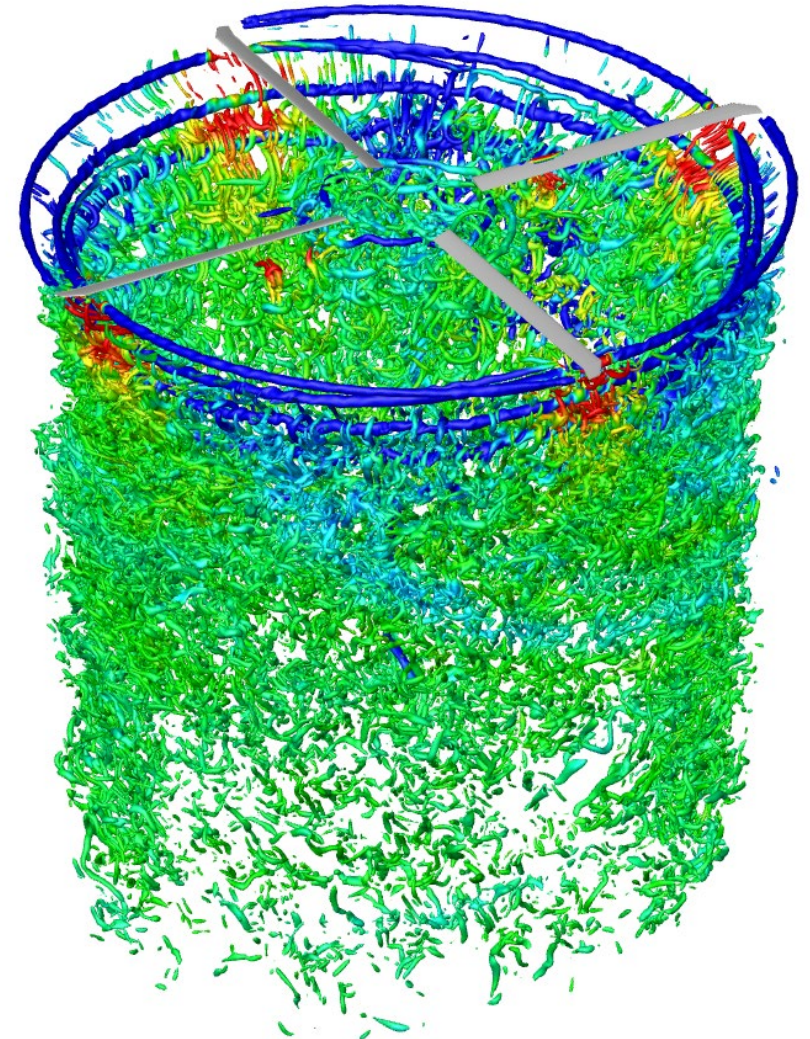
- We've tested grid generation against multiple rotors.
 - Time to generate cases reduces from multiple *days* to a few *seconds*



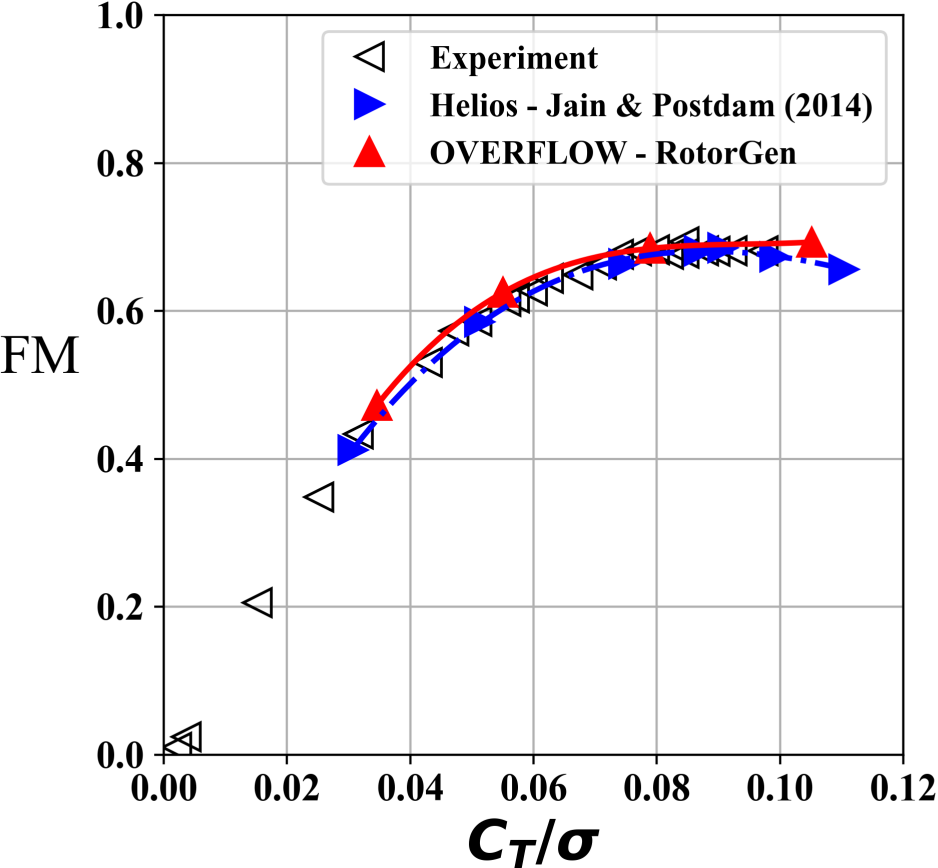
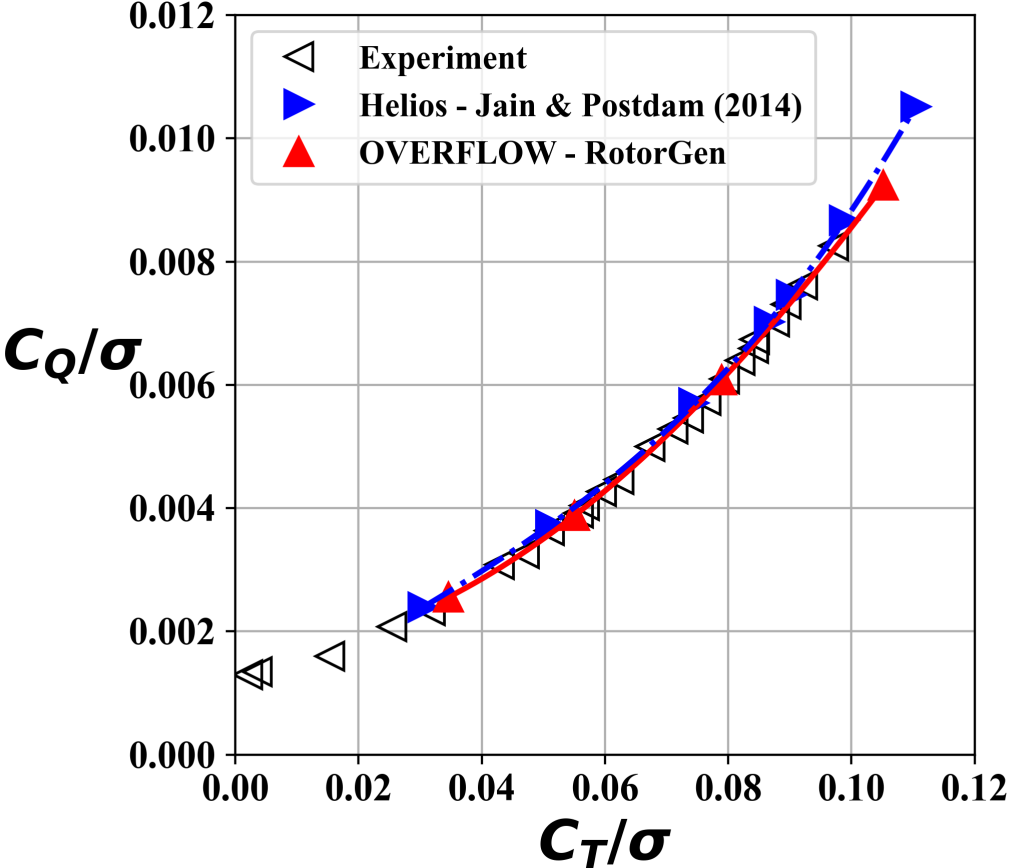
- **Motivation**
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S-76 Validation Study

- **RotorGen was leveraged to complete a collective sweep for an S-76 hovering rotor.**
 - Four collectives are simulated: 4, 6, 8, 10 degs.
 - Radius: 4.67 ft, Tip Mach: 0.61
- **Cases are generated using conceptual design-level definitions.**
 - 5 spanwise rotor definitions
 - 3 airfoil files: SC1013R8, SC1095R8, and SC1095
- **CFD solutions are validated against both experimental measurements and CFD simulations.**
 - Balch, David T. "Experimental study of main rotor/tail rotor/airframe interaction in hover." *Journal of the American Helicopter Society* 30.2 (1985): 49-56.
 - Jain, Rohit K., and Mark A. Potsdam. "Hover predictions on the Sikorsky S-76 rotor using Helios." *52nd Aerospace Sciences Meeting*. 2014.

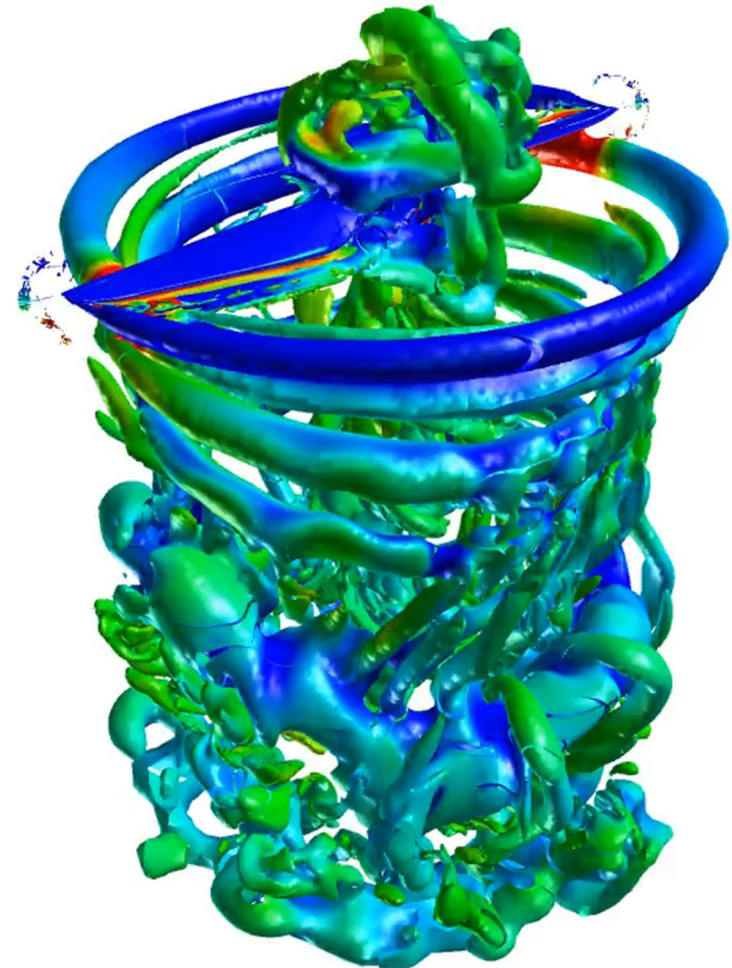


S-76 Performance Coefficients and Figure of Merit

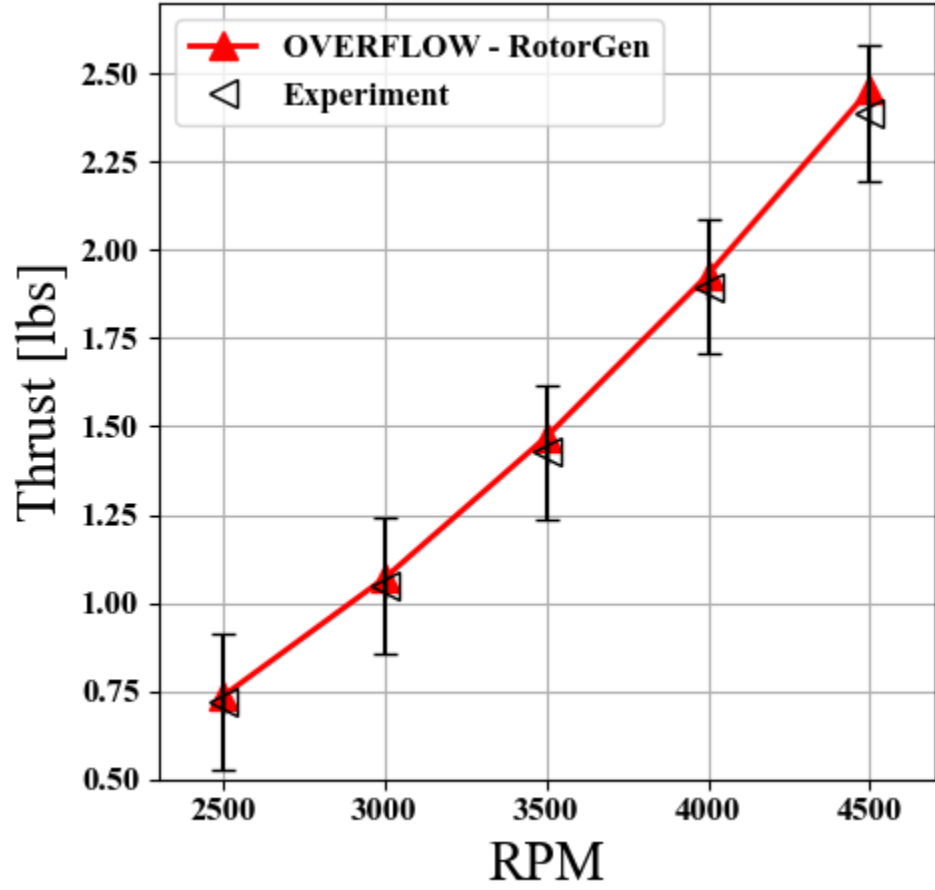
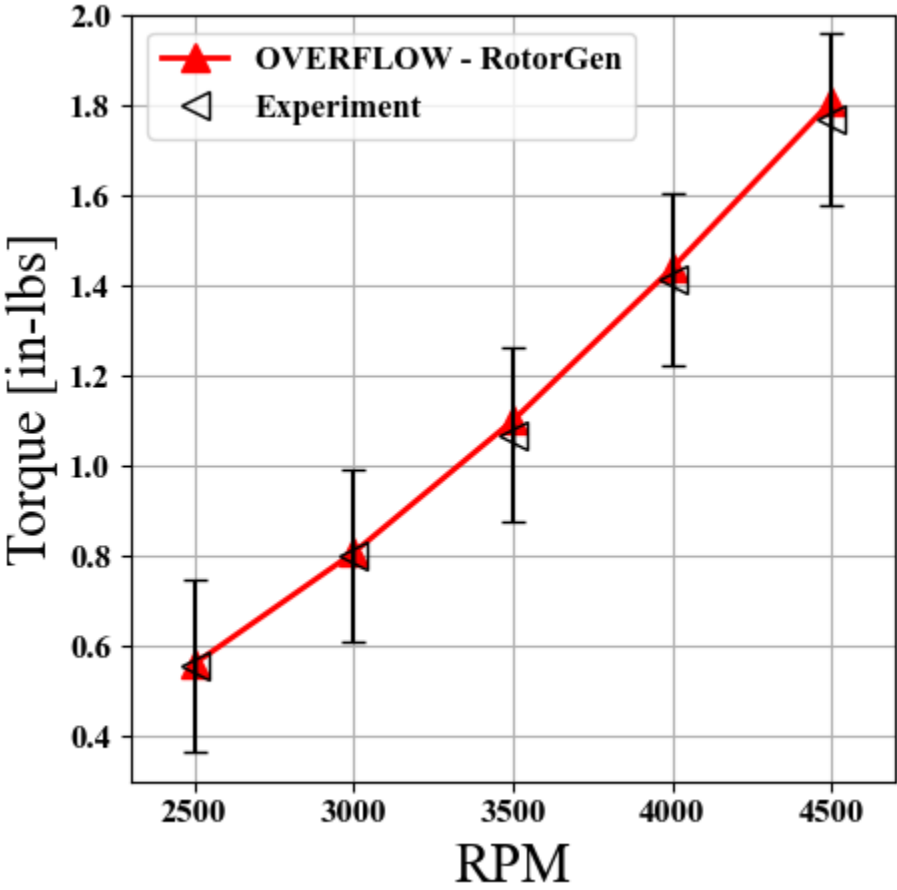


SUI Endurance Validation Study

- **RotorGen was leveraged to complete an RPM sweep for an SUI-Endurance hovering rotor.**
 - Five RPMs are simulated: 2500, 3000, 3500, 4000, 4500 RPM
 - Radius: 0.6 ft, Maximum Tip Mach: 0.3
- **Cases are generated using conceptual design-level definitions.**
 - 22 spanwise rotor definitions
 - 22 airfoil files generated from rotor blade scans
- **CFD solutions are validated against experimental measurements obtained by Carl Russell.**
 - Russell, Carl R., et al. "Wind tunnel and hover performance test results for multicopter uas vehicles." *American Helicopter Society (AHS) International Annual Forum and Technology Display*. No. ARC-E-DAA-TN31096. 2016.

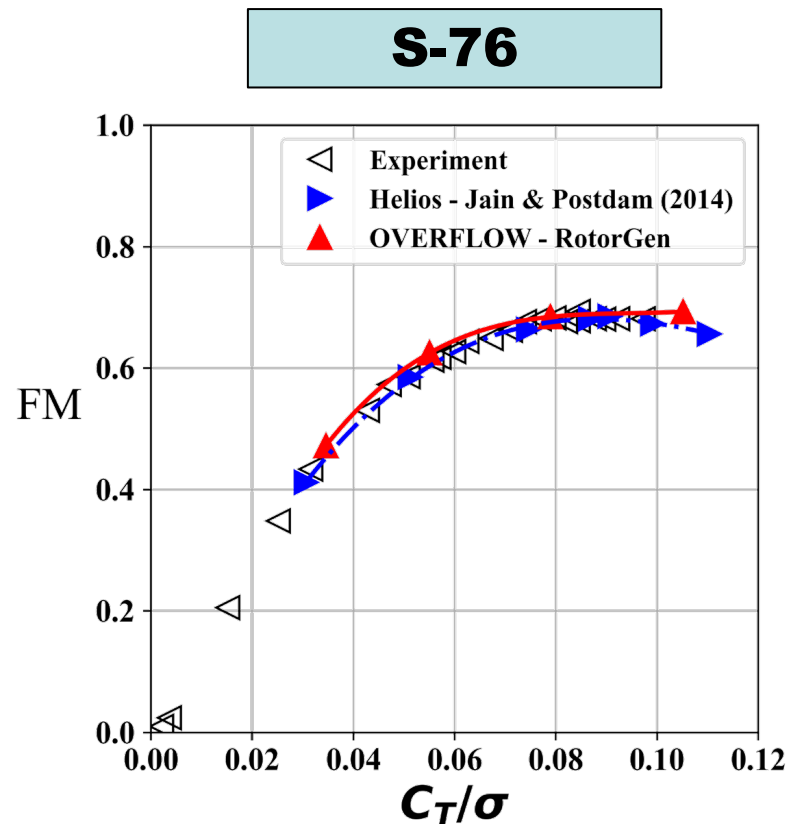


SUI Endurance Performance Versus RPM

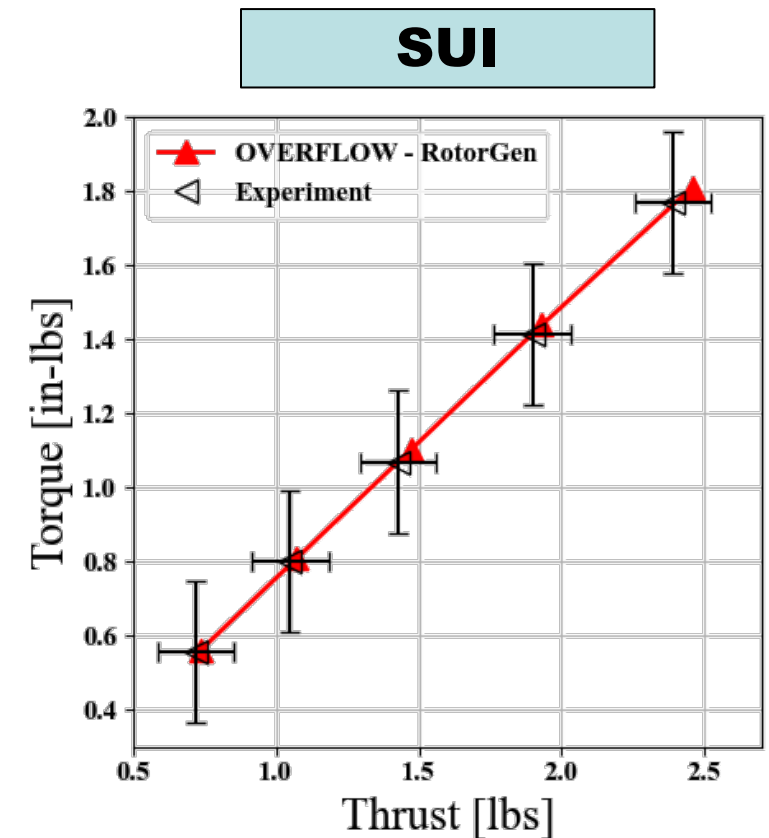


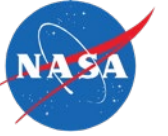
Summary of Validation Effort

- RotorGen-generated cases provided reasonably accurate hovering predictions.
- Hovering performance predictions were achieved for various rotor geometries and scales.



	S-76	SUI
Radius	4.67 ft	0.6 ft
Tip Mach	0.6	0.15 - 0.3
Blades	4	2





- **Motivation**
- **Overview of RotorGen**
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Concluding Remarks

- **Thus far, this effort has completed several key, initial tasks.**
 - A robust, high-level procedure for generating rotor grids has been identified.
 - This procedure has been validated for two hovering rotor performance predictions.
- **In the near term, the preliminary validation effort will be expanded to include a broadened range of applications.**
- **As next steps, we plan to expand RotorGen's capability to include the automation of coupling between CAMRADII and OVERFLOW for aeroelastic modeling and rotor trimming.**
- **The long-term objective of this effort is to publicly release RotorGen.**
- **This effort requires feedback from the community.**



Acknowledgements

- **RVLT management for funding support in this effort**
- **NASA Ames Aeromechanics branch for help understanding conceptual design rotor definitions**
- **Ethan Romander for support in reviewing both OVERFLOW input files and generated grids**
- **Carlos Pereyra for significant contributions with code generation**
- **CGT group for assistance in grid generation**

