

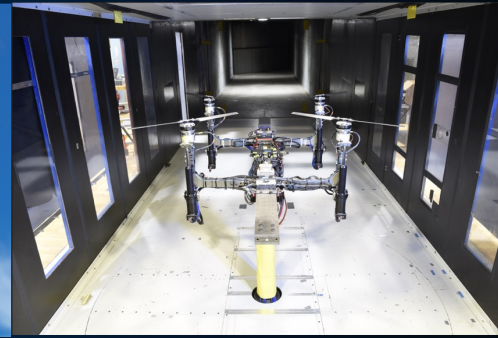


# Updates to NASA Urban Air Mobility Reference Vehicles Incorporating Recent Technology, Policy, and Economic Developments

Christopher Silva

Conceptual Design Tech Lead and Senior Technical Advisor  
NASA Revolutionary Vertical Lift Technology Project

VFS Electric VTOL Symposium  
*Santa Clara, California*  
*6-8 February 2024*

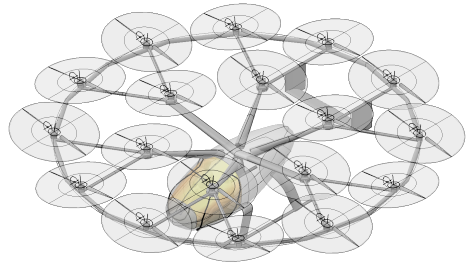






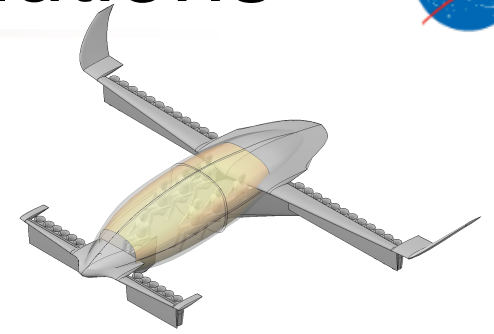


# We observe very diverse VTOL vehicle solutions



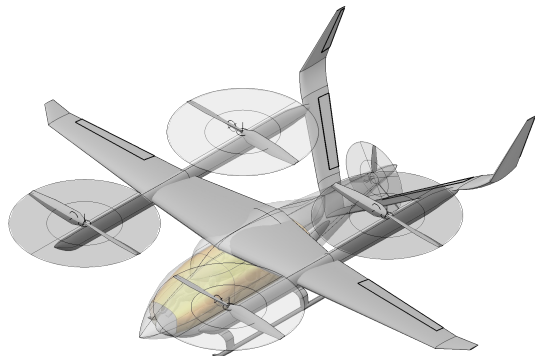
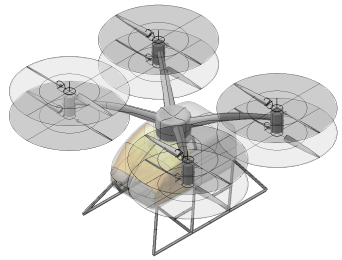
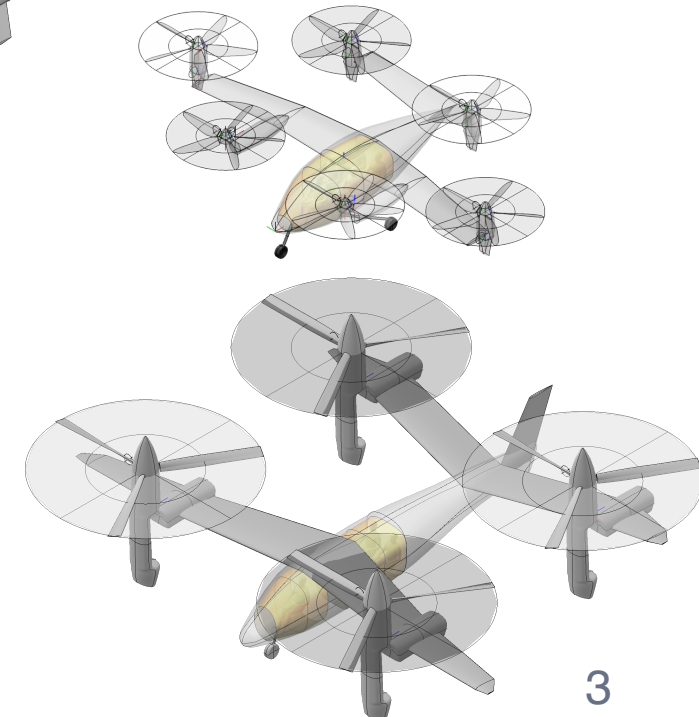
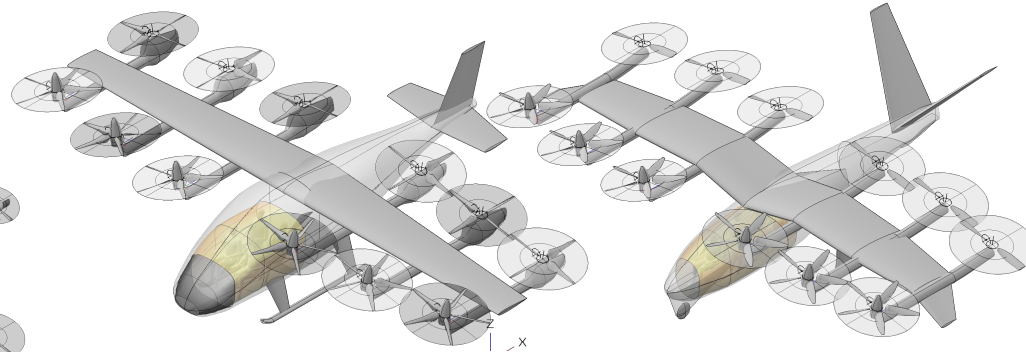
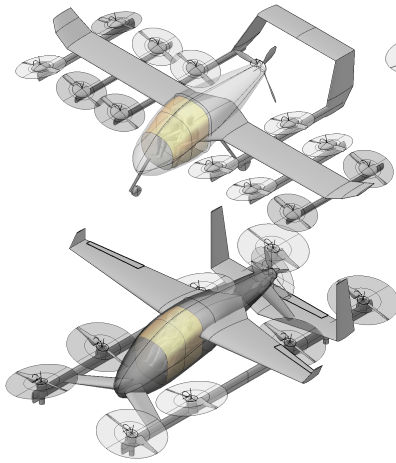
Stopping rotor  
Lift+cruise

Stop some rotors,  
tilt some rotors



Tiltrotors and tiltwings

Multicopters



# Why we designed the NASA reference vehicles

- **Vehicle model inputs and outputs publicly available**

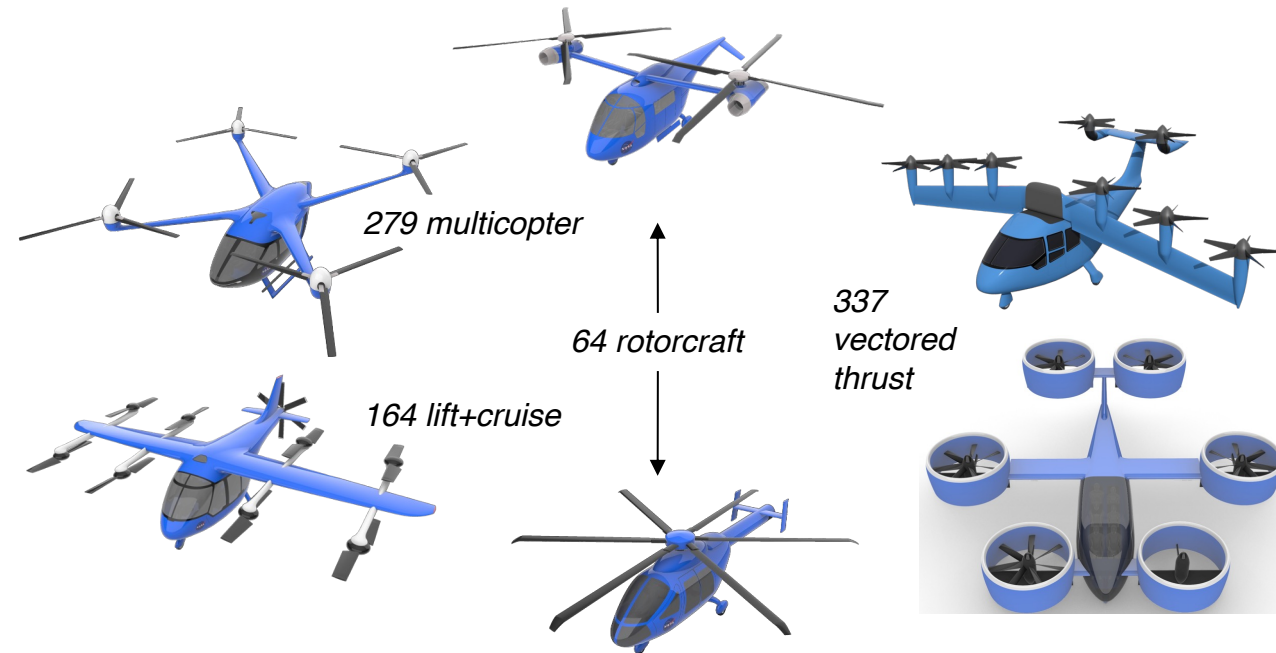
- Discussions can be quantitative
- Demonstration cases for training
- Features representative of vehicles
- Missions and design conditions
- Margins and reliability requirements

- **Focus and guide government research**

- Enable contracted work to be published
- Assess technology payoff
- Guide tool development
- Scope validation tests

- **No plan or desire to build the vehicles**

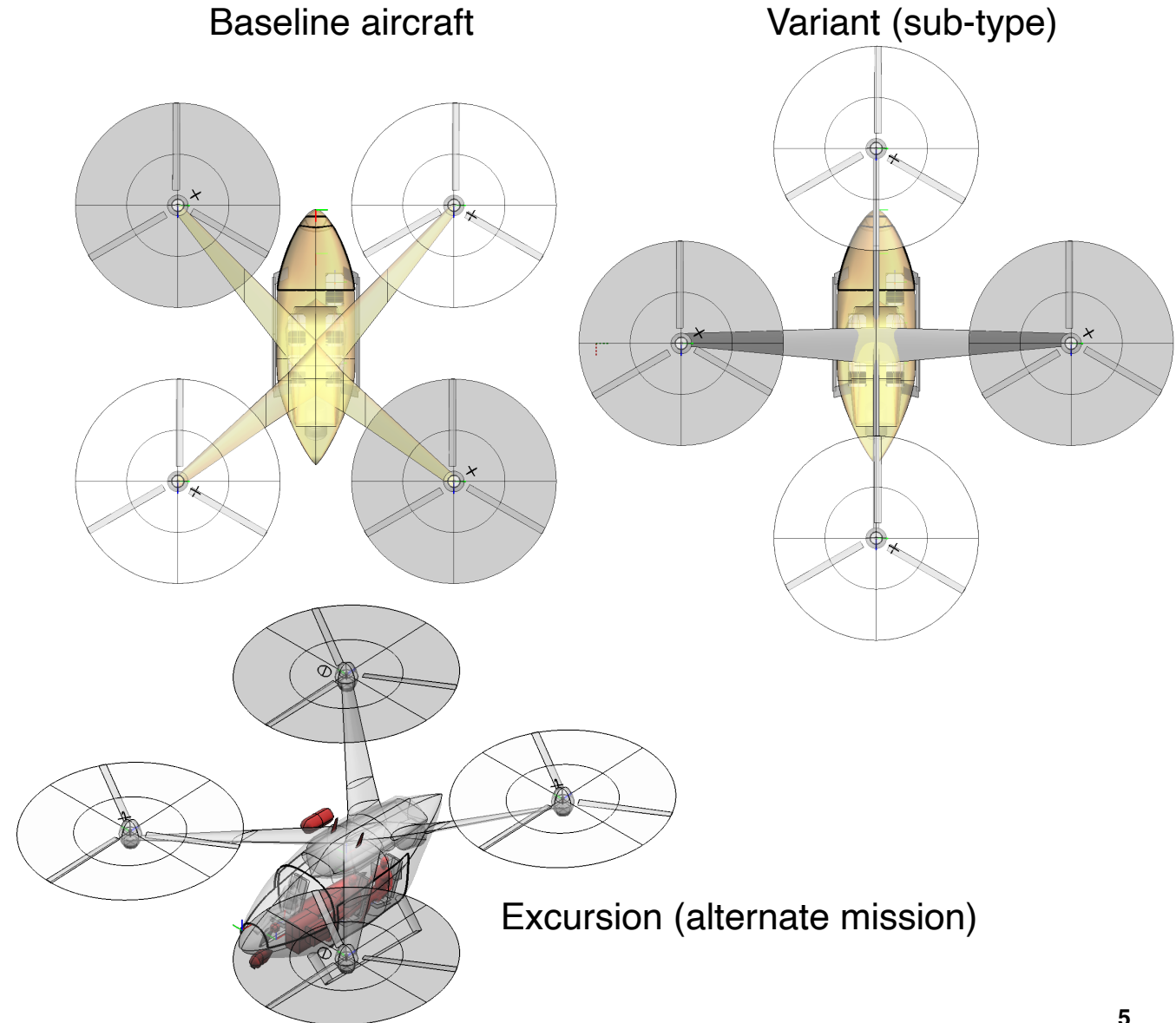
- But these are not cartoons





# What is the composition of a reference vehicle model?

- **Baseline, excursions, and variants**
- **Reports**
- **NDARC design and off-design**
- **Geometry**
- **Comprehensive Analysis**
- **Structures**
- **Flight Dynamics**







# What is the composition of a reference vehicle model?

- Baseline, excursions, and variants
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- Geometry
- Comprehensive Analysis
- Structures
- Flight Dynamics

## Papers and journal articles

**Concept Vehicles for VTOL Air Taxi Operations**

Wayne Johnson      Christopher Silva      Eduardo Soils  
 NASA Ames Research Center      Science & Technology Corp.  
 Moffett Field, California      Moffett Field, California  
 wayne.johnson@nasa.gov      christopher.silva@nasa.gov      eduardo.soils@nasa.gov

**ABSTRACT**

Concept vehicles are presented for air taxi operations, also known as urban air mobility or on-demand mobility applications. Considering the design-space dimensions of payload (passengers and pilot), range, aircraft type, and propulsion system, three aircraft are designed: a single-passenger (250-lb payload), 50-nm range quadrotor with electric propulsion; a six-passenger (1200-lb payload), 4x50 = 200-nm range side-by-side helicopter with hybrid propulsion; and a fifteen-passenger (3000-lb payload), 8x50 = 400-nm range tiltwing with turbo-electric propulsion. These concept vehicles are intended to focus and guide NASA research activities in support of aircraft development for emerging aviation markets, in particular VTOL air taxi operations. Research areas are discussed, illustrated by results from the design of the concept vehicles.

**INTRODUCTION**

Urban air taxi operations, also known as urban air mobility or on-demand mobility applications, are enabled by vertical take-off and landing (VTOL) capability, power and energy requirements are minimized by using low disk-loading rotors, and short range requirements permit consideration of non-traditional propulsion concepts. The community of innovation has recognized that technology advances in structures, automation and control, energy generation-storage-utilization, and tools for design and analysis, coupled with pressures of resource availability and population density, make this the right time to explore new ways to move people and goods (ref. 1). The objective of the present work is to identify concept vehicles that can be used to focus and guide NASA research activities in support of aircraft development for emerging aviation markets, in particular VTOL air taxi operations.


To meet this objective, the designs are carried far enough to identify crucial technologies and research requirements,

Presented at the AHS Technical Conference on Aeromechanics Design for Transformative Vertical Flight, San Francisco, CA, January 16-19, 2018. This is a work of the U.S. Government and is not subject to copyright protection.

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## NASA publications

NASA/TM-20210017971



**Design of a Tiltwing Concept Vehicle for Urban Air Mobility**

*Stena K. S. Whiteside, Beau P. Pollard, Kevin R. Antcliff, Nikolas S. Zawodny, and Xiaofan Fei*  
Langley Research Center, Hampton, Virginia

*Christopher Silva*  
Ames Research Center, Moffett Field, California

*Glenn L. Medina*  
Universities Space Research Association, Hampton, Virginia

June 2021





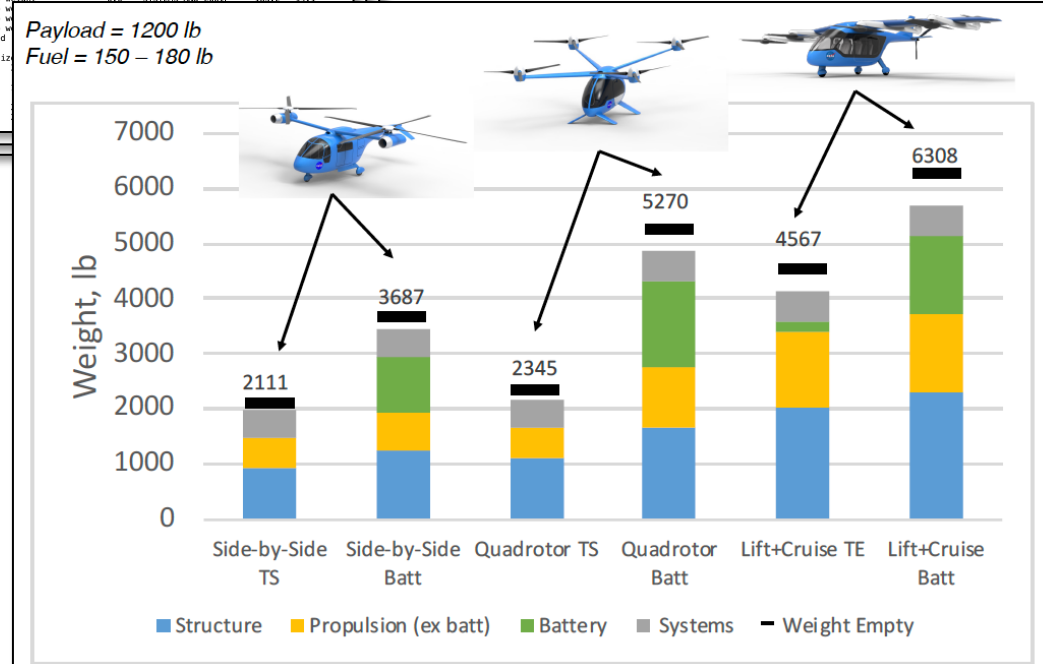
# What is the composition of a reference vehicle model?

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NDARC job outputs (text)

Multicopter 1: 4 main rotors				Aircraft design (fixed or calculated)			
components: 4 rotors, 0 wings, 0 tails				aircraft: calc DGM calc WE			
propulsion group: 4 engine groups				rotor 1: calc radius fix diskload fix CW/s fix Vtip calc sigma			
components: 0 jet groups, 0 charge groups				rotor 2: calc radius fix diskload fix CW/s fix Vtip calc sigma			
Disk loading = 2.58 (Aref = 519.1)				rotor 3: calc radius fix diskload fix CW/s fix Vtip calc sigma			
Power loading = 14.19 (Pav = 91.4)				rotor 4: calc radius fix diskload fix CW/s fix Vtip calc sigma			
				aircraft: calc WMTD calc fueltank calc SDGW calc D/q calc download			
Rotor 1: main/multicopter				Drag			
Rotor 2: main/multicopter				D/q (ft*2)			
Rotor 3: main/multicopter				cruise hel vert			
Rotor 4: main/multicopter				contingency			
Rotor 1				fuselage			
kind 1 2 3				fuse fit & fix			
DL 2.500 2.500 2.500				rotor-body int			
CW/s 0.1000 0.1000 0.1000				landing gear			
FDGW 0.250 0.250 0.250				rotor 1 hub			
Tdesign 324.5 324.5 324.5				rotor 1 pylon			
Pdesign 19.9 22.5 19.9				rotor 1 spinner			
farea 1.000 1.000 1.000				rotor 2 hub			
radius 6.427 6.427 6.427				rotor 2 pylon			
sigma (T-wt) 0.0466 0.0433 0.0433				rotor 2 spinner			
sigma (geom) 0.0896 0.0466 0.0466				rotor 3 hub			
chord (T-wt) 0.43 0.29 0.29				rotor 3 pylon			
chord (geom) 0.47 0.31 0.31				rotor 3 spinner			
FTWsigma 0.9286 0.9286 0.9286				rotor 4 hub			
taper 0.7500 0.7500 0.7500				rotor 4 pylon			
aspect ratio 13.72 20.50 20.50				rotor 4 spinner			
nblade 3 3 3				engine 1 nac			
rotation 3 3 3				engine 2 nac			
prop group 1 prin 1 dep 1 dep				engine 3 nac			
Vtip 450.0 550.0 550.0				engine 4 nac			
gear 1.000 1.222 1.222				cruise CD=f/Aref 0.0068			
Vtip cruise 450.0 550.0 550.0				cruise CD=f/Swet 0.0068			
P1limit_rs 87.0 87.0 87.0				cruise (D/q)/(W/1000)^2/3			
flap freq 1.030 1.030 1.030				vertical DL/T=(D/q)/Aref			
Lock number 3.67 3.20 3.20				fuselage length 8.00			
aut rotor index 1.402 1.835 1.835				nose height 4.00			
incidence 0.0 0.0 0.0				fuselage width 5.00			
cant angle 0.0 0.0 0.0				cabin surface area 86.40			
Rotor 1				floor area 19.20			
kind main				rotor 1 blade tip			
DL 2.500				rotor 2 blade tip			
CW/s 0.1000				rotor 3 blade tip			
FDGW 0.250				rotor 4 blade tip			
Tdesign 324.5				landing gear fixed			
Pdesign 18.1				growth factor			
farea 1.000				hover operating siz			
radius 6.427				length			
sigma (T-wt) 0.0433				fuel tank 1 h3			
sigma (geom) 0.0466				fuel tank 1 kh			
chord (T-wt) 0.29				tank 1 act N3			
chord (geom) 0.31				tank 1 MJ/liter			
FTWsigma 0.9286				tank 1 xmbd			
taper 0.7500							

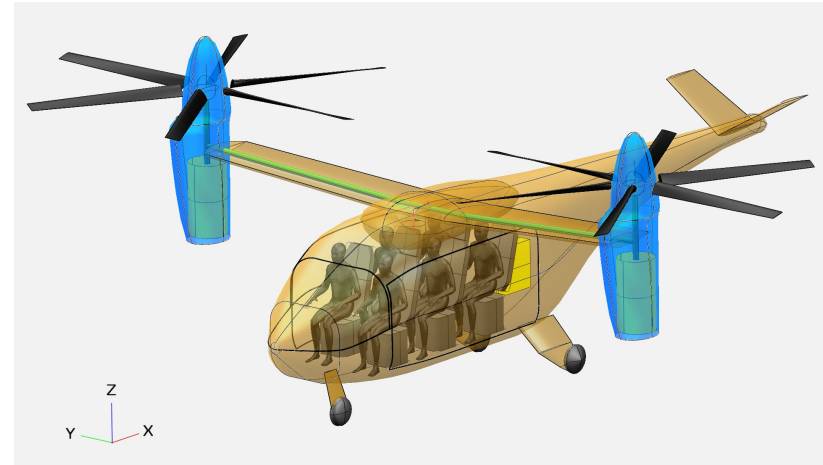
Charts of results





# What is the composition of a reference vehicle model?

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- **Geometry**
- Comprehensive Analysis
- Structures
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OpenVSP for initial design



Watertight, smoothed geometry for CFD



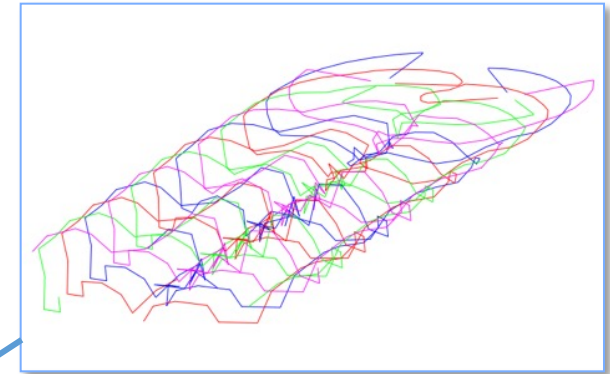
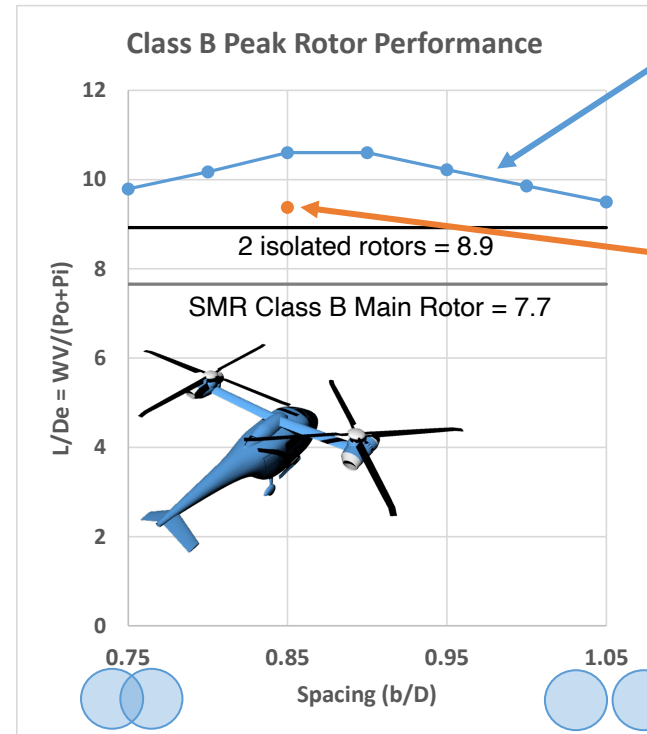
Rendered aircraft for communication and engagement

# What is the composition of a reference vehicle model?

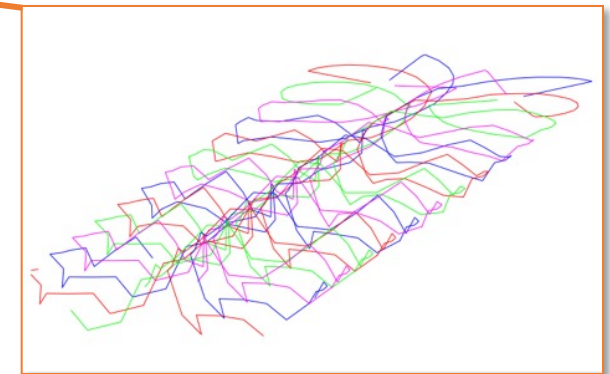
- **Baseline, excursions, and variants**
- **Reports**
- **NDARC design and off-design**
- **Geometry**
- **Comprehensive Analysis**
- **Structures**
- **Flight Dynamics**

Effect captured in NDARC twin rotor interference parameters:

MODEL\_twin='side-by-side',  
Kh\_twin=.92,Kf\_twin=1.0,



Outboard advancing  
Peak L/De ~ 10.5  
Peak b/D ~ 0.85-0.9

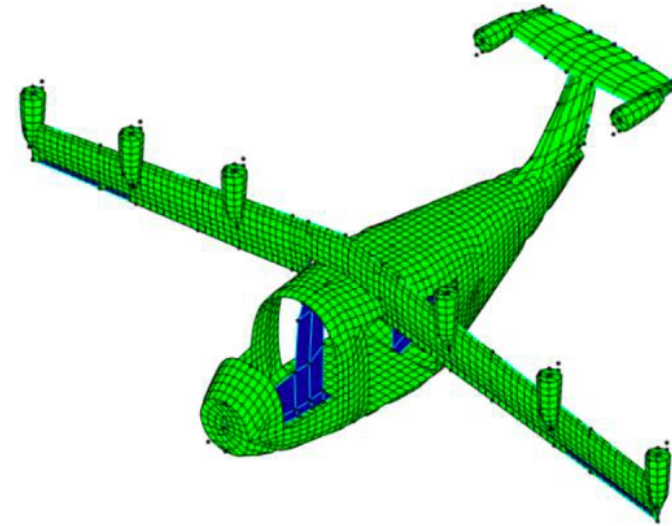


Outboard retreating  
Peak L/De ~ 9.4

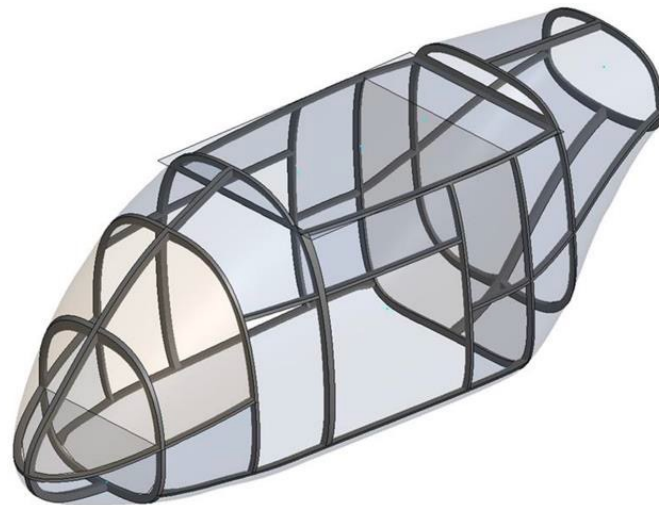


# What is the composition of a reference vehicle model?

- Baseline, excursions, and variants
- Reports
- NDARC design and off-design
- Geometry
- Comprehensive Analysis
- **Structures**
- Flight Dynamics



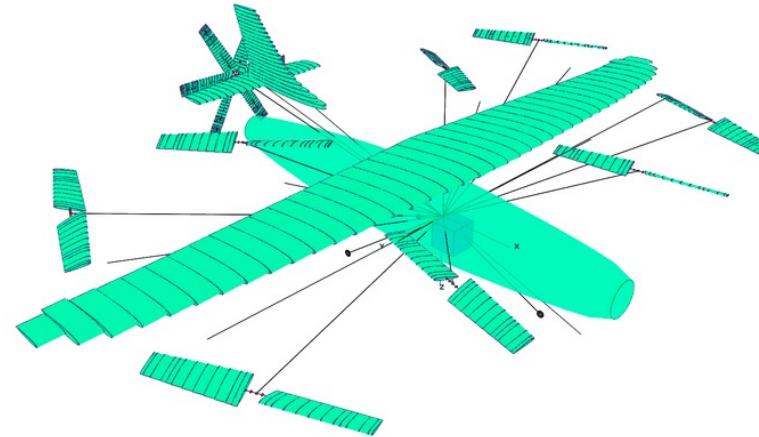
Tiltwing NASTRAN model



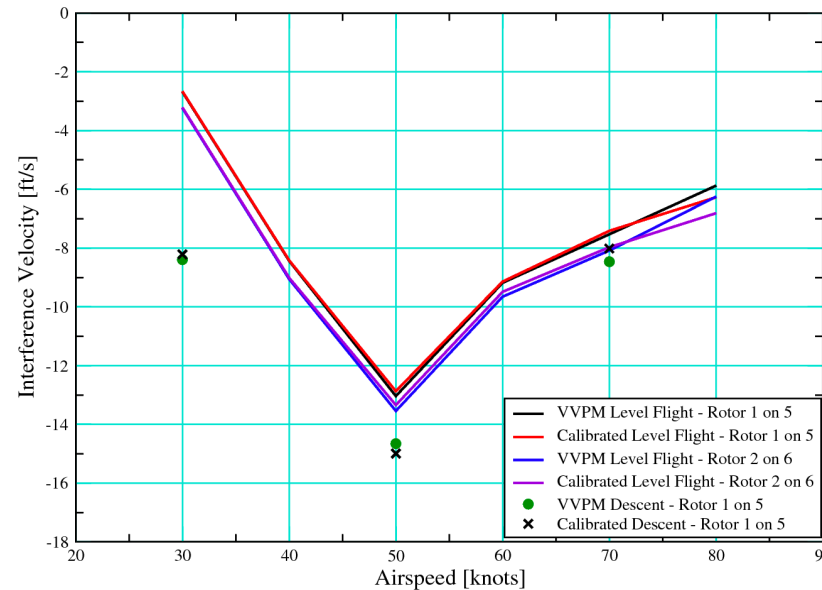
Lift+Cruise cabin crash test article structural components

# What is the composition of a reference vehicle model?

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- **Flight Dynamics**



Lift+Cruise FlightLab geometry






Lift+Cruise FlightLab rotor-to-rotor interference

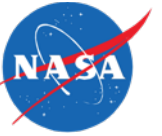


# Reference vehicles for UAM: baselines & some trades we have performed



	Quadrotor	Side-by-side helicopter	Tiltwing			
"Air taxi"				Johnson, "Concept Vehicles for VTOL Air Taxi Operations," January 2018	- Vehicle type	
"UAM"					- Mission range	
Occupants	1, 2, 4, 6	6	6, 15		- Propulsion type and topology	
Rotors	RPM, <b>collective Flapping</b> , stiff Nrotor= 4, 6, 8 Rear rotor z/D	Rotor y/R Rotation	Collective+trim, <b>monocyclic</b>		- Rotor control	
Propulsion	Direct, gear, xshaft Turboshaft, diesel, <b>battery</b>	Turboshaft, <b>parallel hybrid</b> , battery	Direct, gear, <b>xshaft</b> Turboshaft, turboelectric, battery			

# Reference vehicles for UAM: baselines & some trades we have performed



	Quadrotor	Side-by-side helicopter	Tiltwing	Lift+cruise compound	Quiet single main rotor helicopter	Tiltduct
"Air taxi"						
"UAM"						
Occupants	1, 2, 4, 6	6	6, 15	6	6	6
Rotors	RPM, <b>collective Flapping</b> , stiff Nrotor= 4, 6, 8 Rear rotor z/D	Rotor y/R Rotation	Collective+trim, <b>monocyclic</b> , <b>coll+tiltrotor DEP</b>	<b>RPM</b> , collective Rotor z/D	NOTAR	Collective with TBD duct trades
Propulsion	Direct, <b>gear</b> , <b>xshaft</b> <b>Turboshaft</b> , diesel, <b>battery</b>	<b>Turboshaft</b> , <b>parallel hybrid</b> , battery	<b>Direct</b> , gear, <b>xshaft</b> Turboshaft, <b>turboelectric</b> , battery	<b>Turboelectric</b> , battery	<b>Turboshaft</b> , battery	<b>Turboelectric</b> , battery
Noise tech	Tip speed, tip shape, rear rotor z/D	Tip speed, tip shape, rotor y/R, rotation, IBC		Tip speed, rotor z/D, pitch trim	<b>NOTAR</b> <b>Drop</b> <b>Sweep</b> <b>IBC HHC</b>	

Green = Initial exploratory baseline (2018 AHS SF)

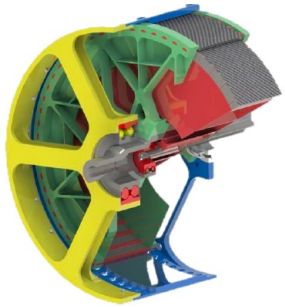
Blue = 6 pax baseline (2018 AVIATION, 2020 VFS SF, 2020 AVIATION)



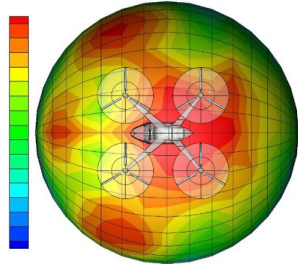
# What have people been (publicly) doing with the models?



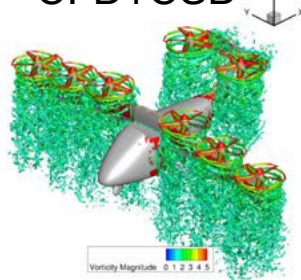
Electric propulsion



Noise



CFD+CSD

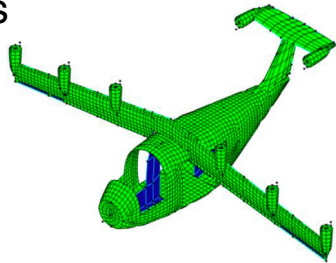


**NASA**

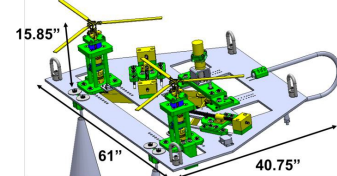
Flight dynamics, handling/ride qualities



Structures



Wind tunnel

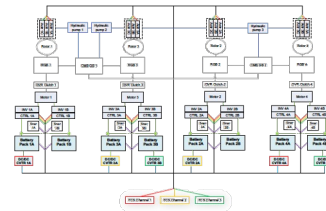


Crashworthiness

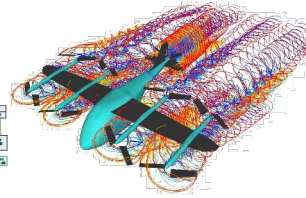


281 papers using the NASA UAM reference vehicles as of 1 January 2024

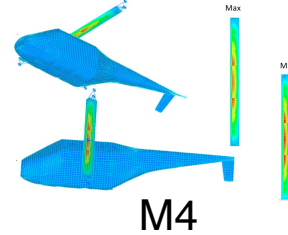
Boeing



CDI

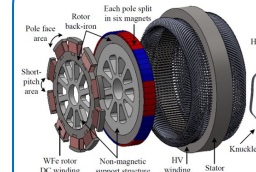


**US Industry**

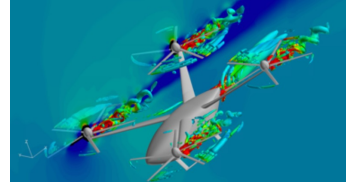


M4

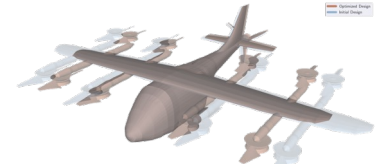
NDSU



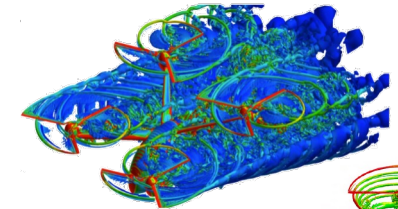
Georgia Tech



**Academia**

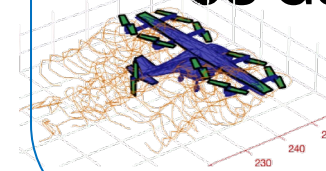


UCSD+SDSU+UCD+BYU+  
Aurora+M4



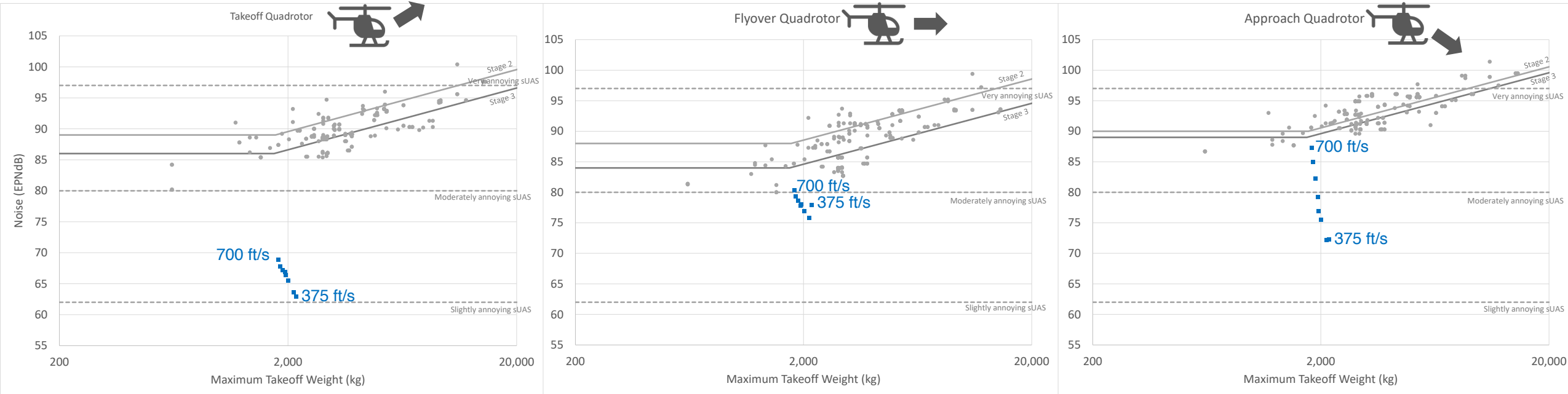
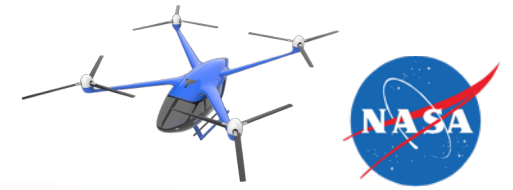
US Army

**US Government**



CDI+MIT+FAA

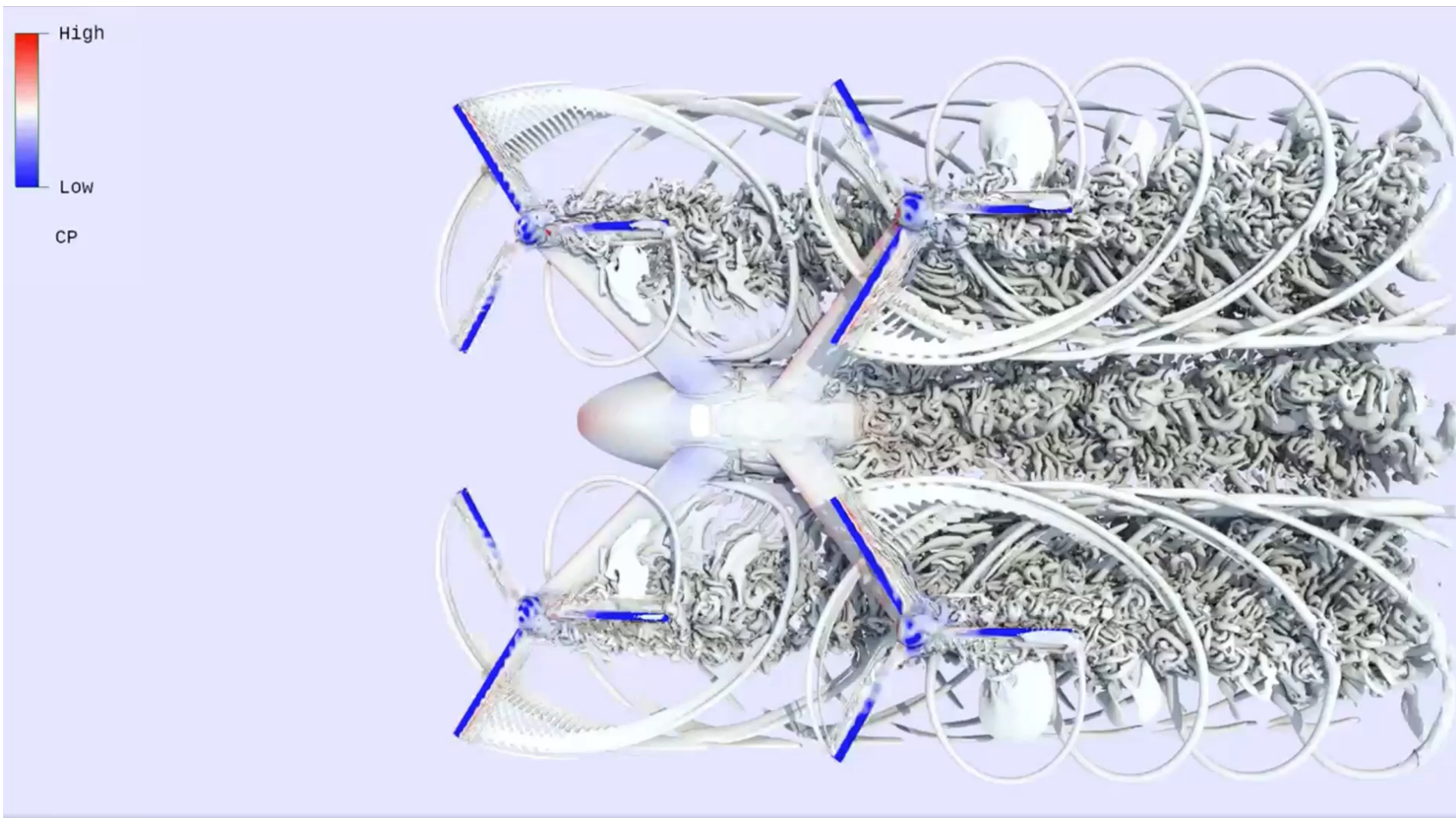
# Tip speed: Quadrotor most (noise) improvement in approach



Tip speed (ft/s)	$\Delta$ Takeoff (EPNdB)	$\Delta$ Flyover (EPNdB)	$\Delta$ Approach (EPNdB)	$\Delta$ Empty Wt (%)	$\Delta$ Block time (%)	$\Delta$ Flyaway (%)
375	-3.9	0.2	-7.0	9%	18%	14%
400	-3.2	-2.0	-7.1	14%	10%	25%
450	-1.3	-0.9	-3.7	5%	7%	10%
500	-0.5	0.2	-2.4	1%	4%	4%
<b>550</b>			Baseline			
600	0.3	0.9	3.0	-3%	-2%	-3%
650	0.9	1.6	5.7	-7%	-3%	-6%
700	2.1	2.5	8.1	-9%	-1%	-9%



# Loose-coupled CFD: Can we minimize bad interactions? (Ventura Diaz)

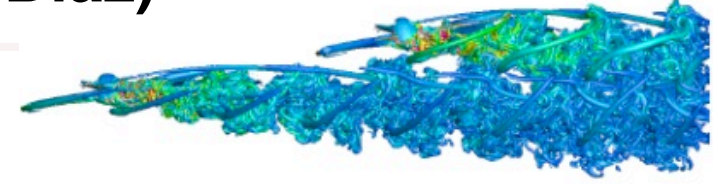


Performance trends similar to comprehensive analysis. Magnitudes vary.

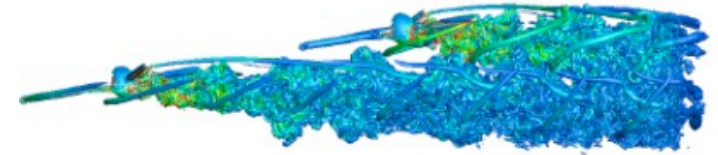
# More detailed look into rotor-rotor interactions (Ventura Diaz)



- Wakes impact downstream rotors
- Mid-fidelity tools predict the cost / benefit
- High-fidelity to confirm effects (or not!)



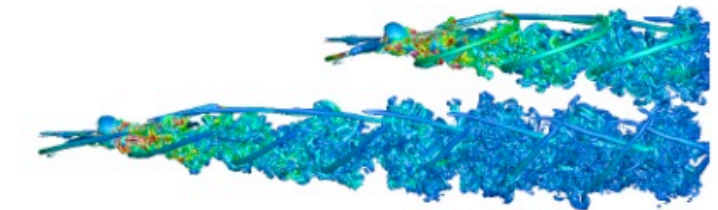
(a) Q-criterion iso-surfaces,  $z_{rear} = 0.25R$



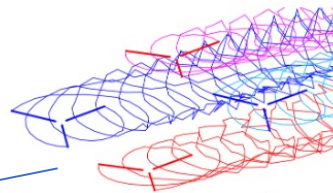
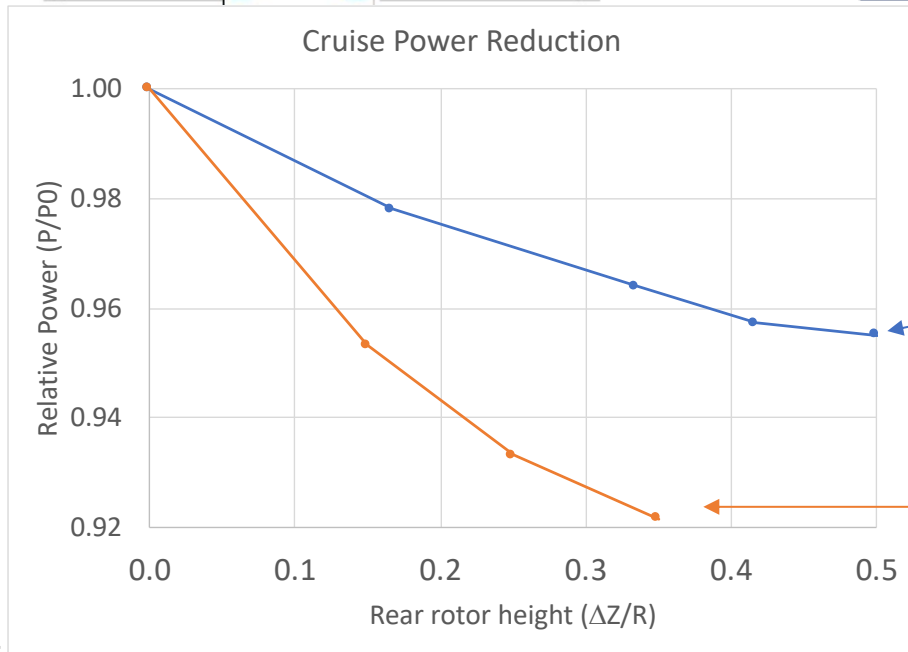
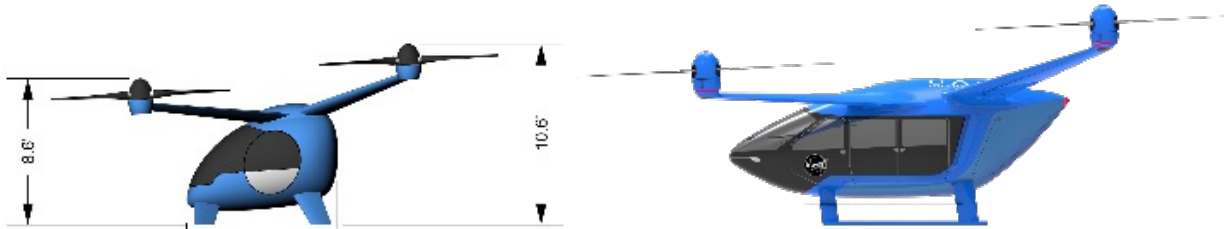
(c) Q-criterion iso-surfaces,  $z_{rear} = 0.4R$



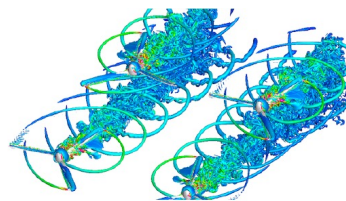
(e) Q-criterion iso-surfaces,  $z_{rear} = 0.5R$



(g) Q-criterion iso-surfaces,  $z_{rear} = 0.6R$



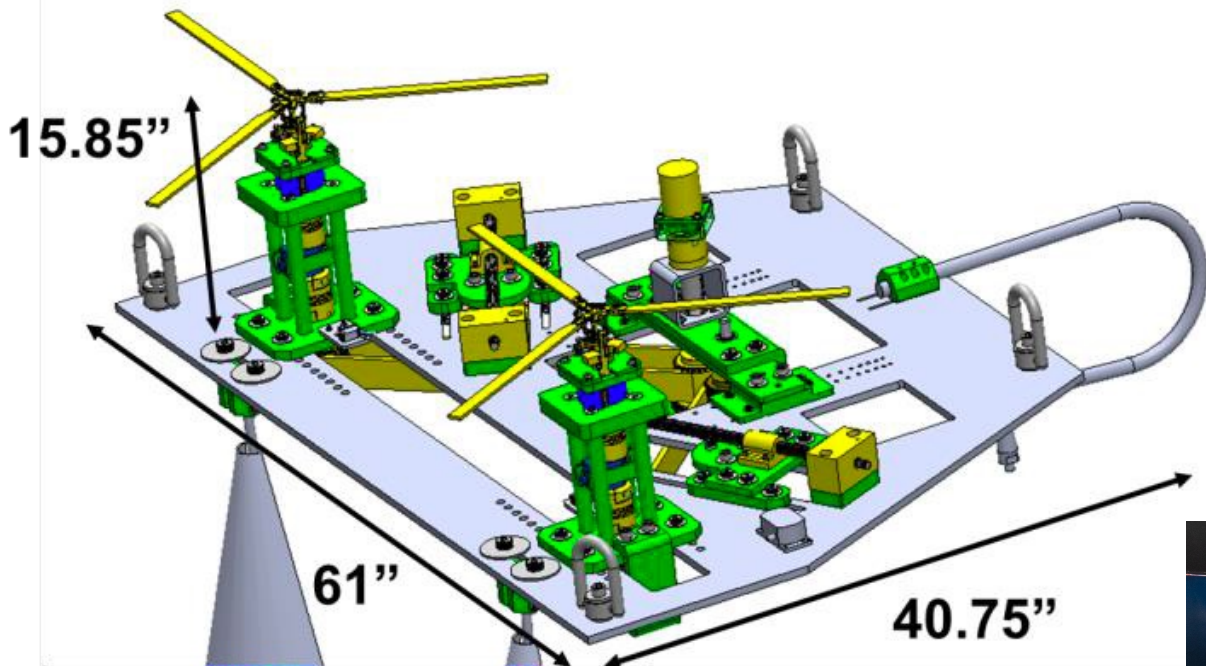
Comprehensive Analysis  
1 pax



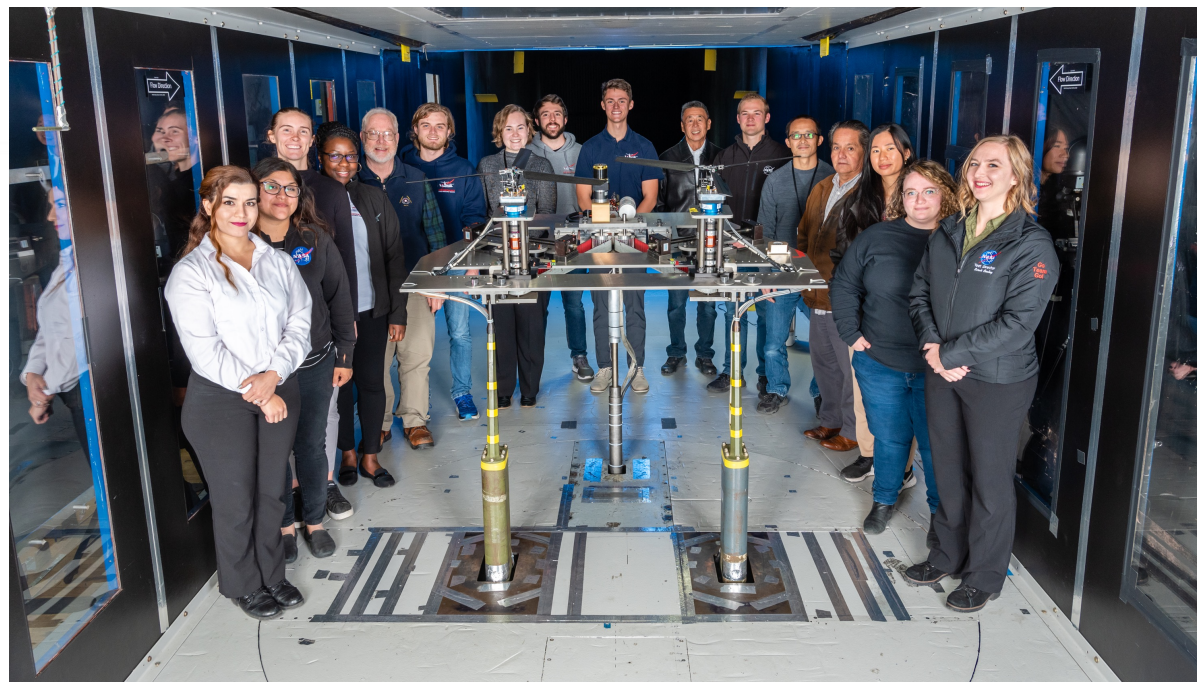
CFD+CA  
6 pax



# If you can predict productive interference, might as well measure it



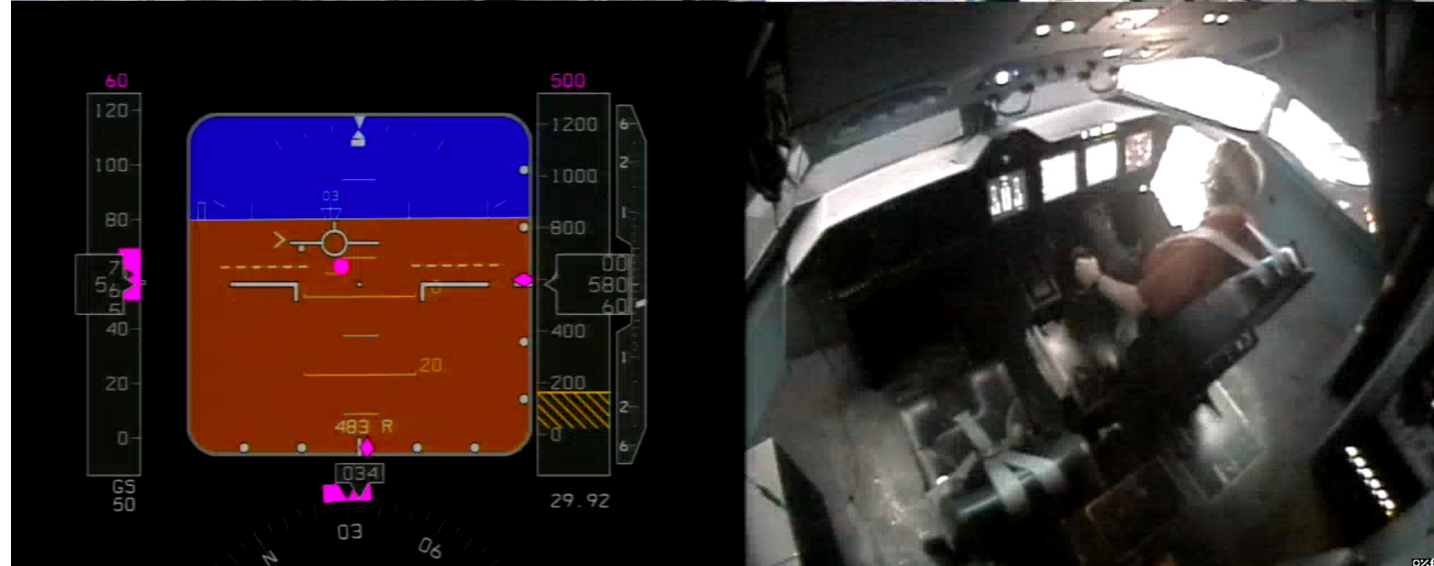
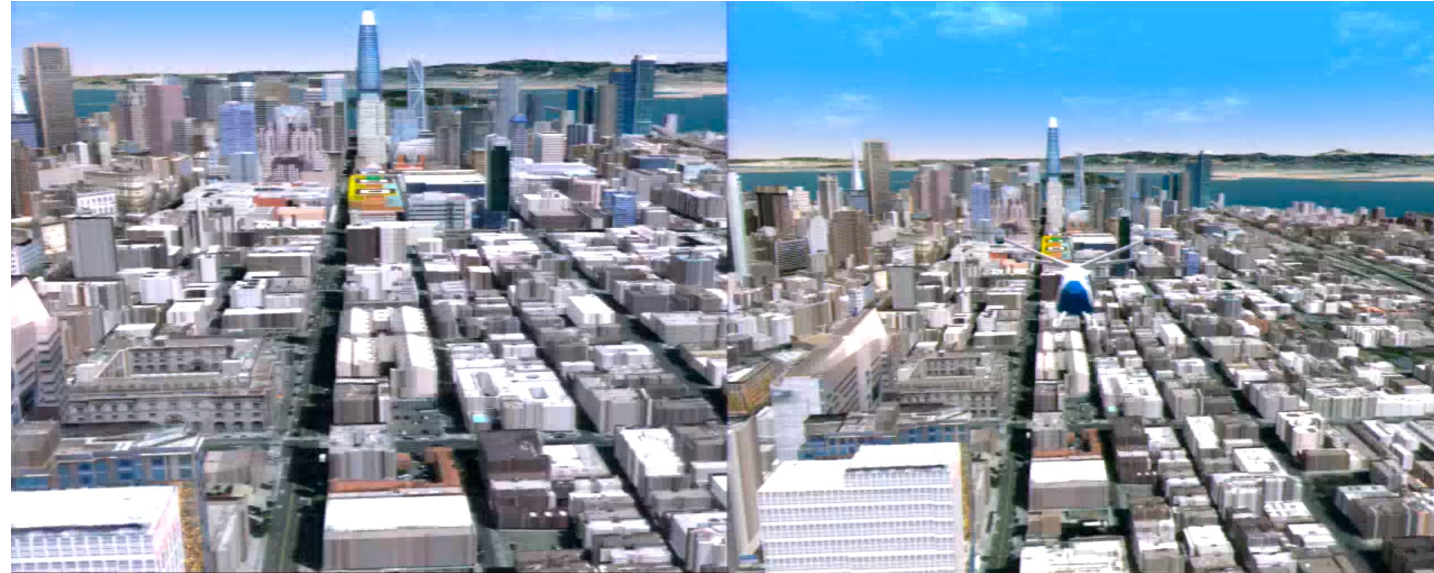
Side-by-Side Test Rig for rotor performance in Army 7x10



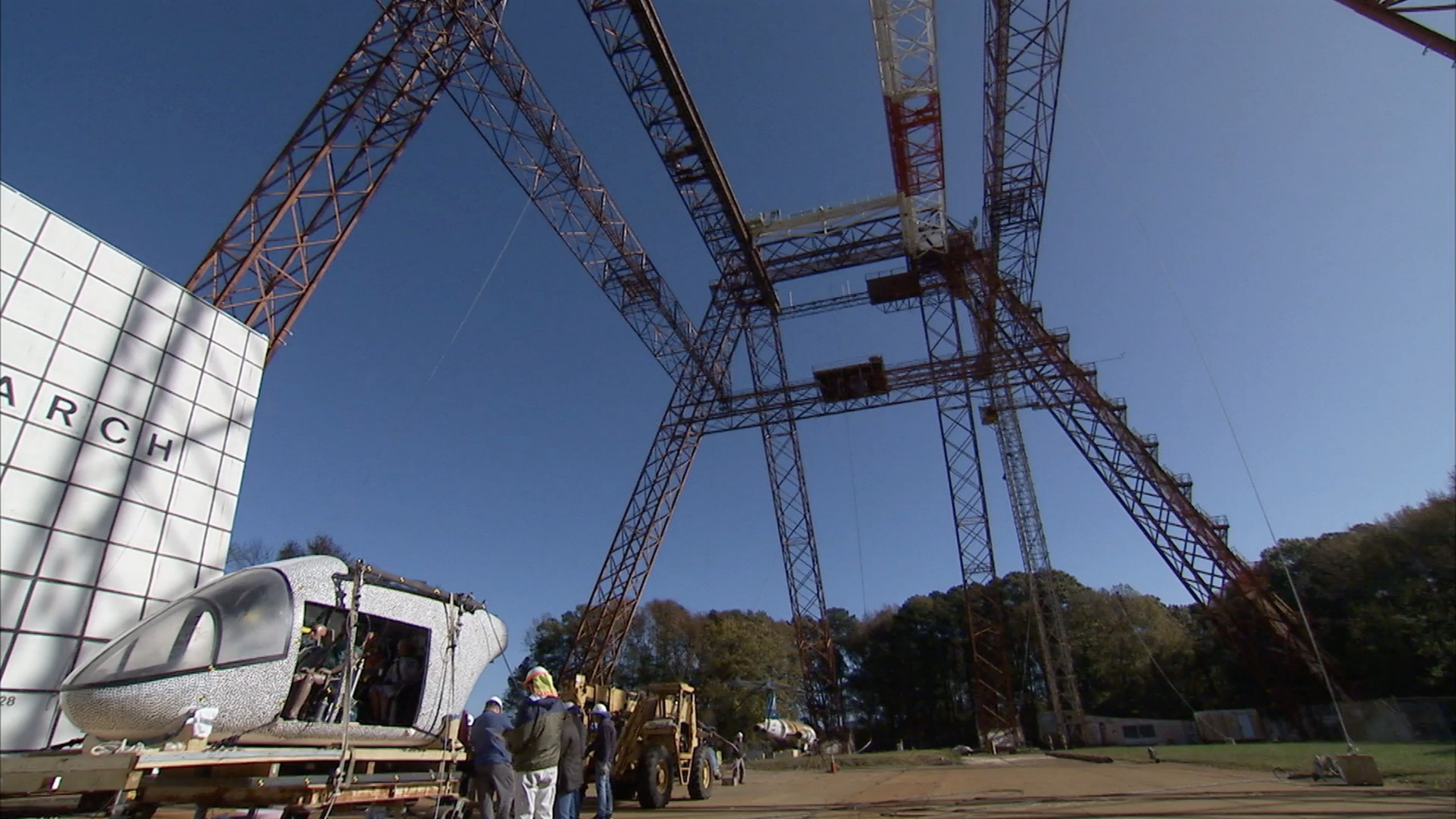


# VMS testing and handling qualities analysis

- **RPM sluggish and sloppy if motors not capable of at least 2x the torque required for steady operation**
  - Pilots really dislike sluggish RPM
  - Passengers dislike sloppy RPM
  - Motor sizing needs to be updated
- **Small rotors for reasonable RPM control**
  - This is one argument for Distributed Electric Propulsion (DEP)
  - Somewhere around 5 ft radius probably the cutoff, but not sure yet

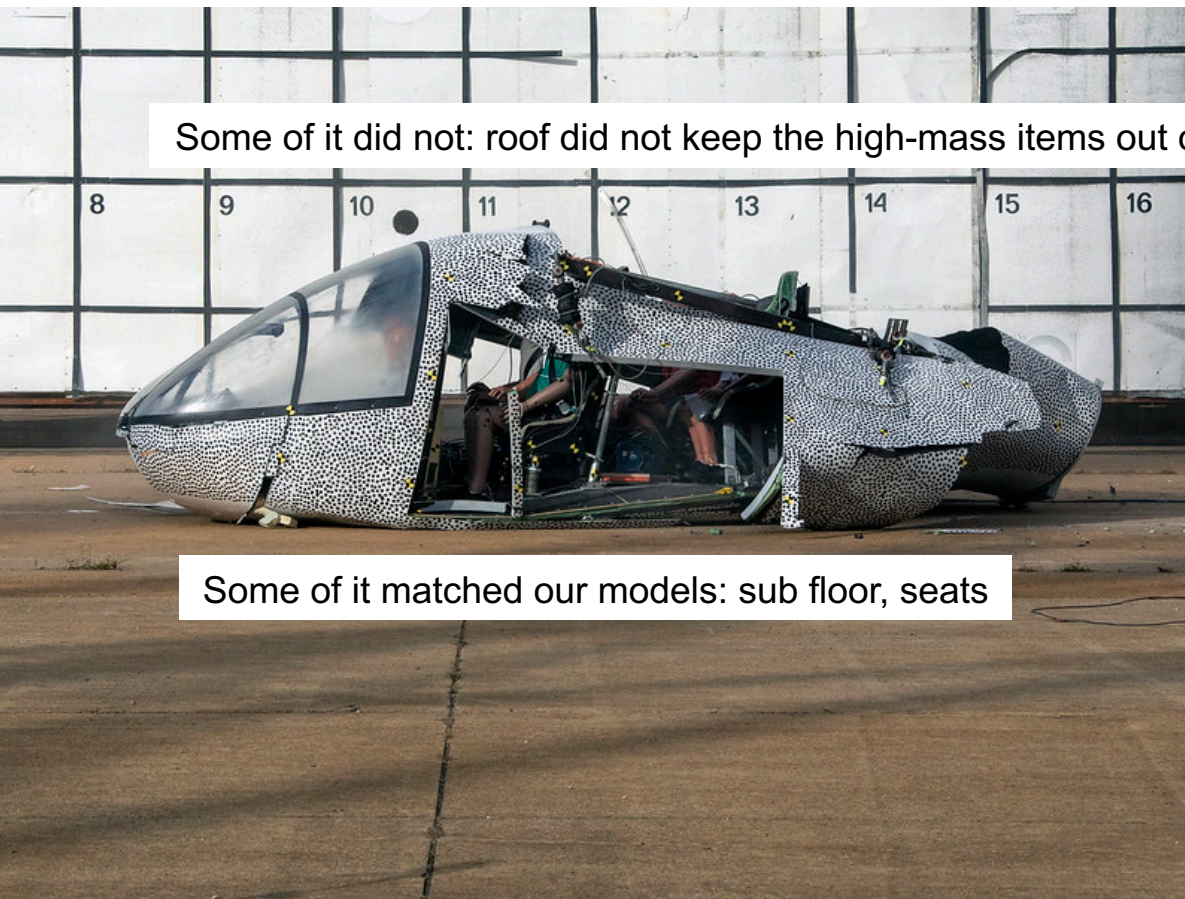






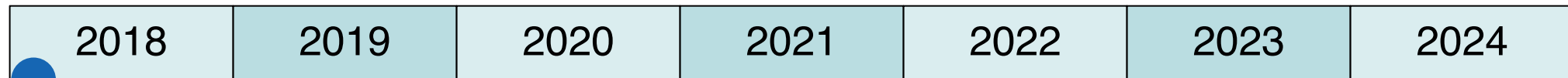


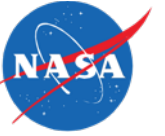
# Fantastic news! We get to do more engineering!





# The UAM landscape was a bit different in early 2018

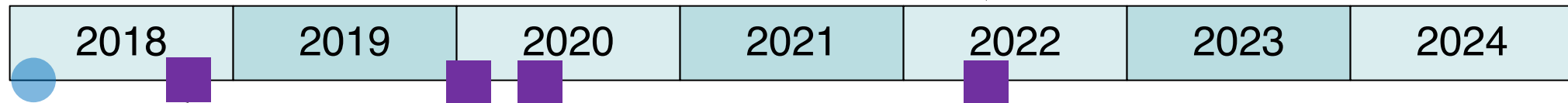




# Some major policy developments in Europe and USA

## FAR Part 23 vs Part 21.17(b) powered lift (May 2022)

- May be similar to tiltrotor requirements
- 2-control trainer aircraft



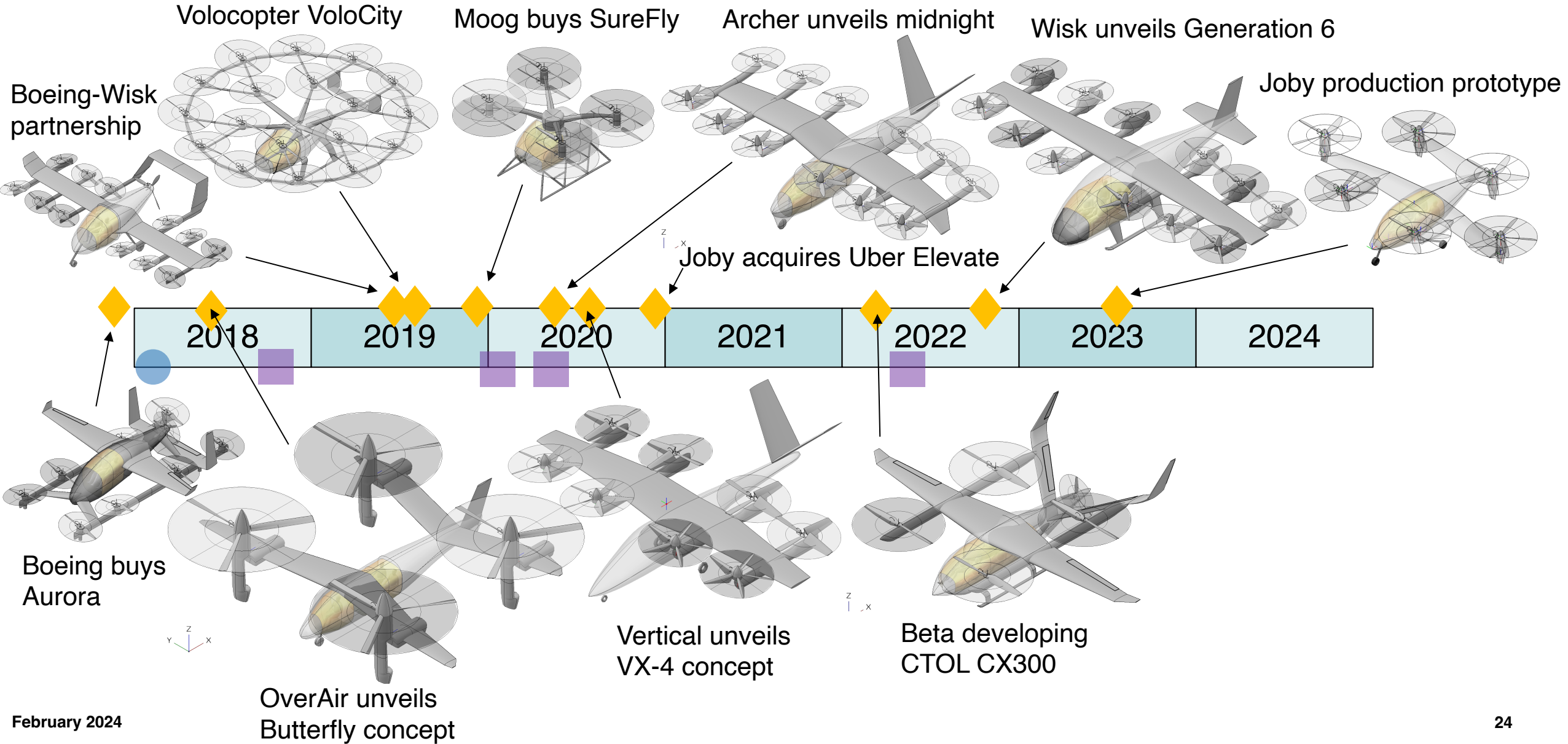
**EASA Means of Compliance published (May 2020; MOC-4 SC-VTOL, December 2023)**

**Proposed Special Condition: Electric/Hybrid Propulsion System (SC-E-19, January 2020)**

**EASA Special Condition for VTOL (SC-VTOL-01, October 2018)**

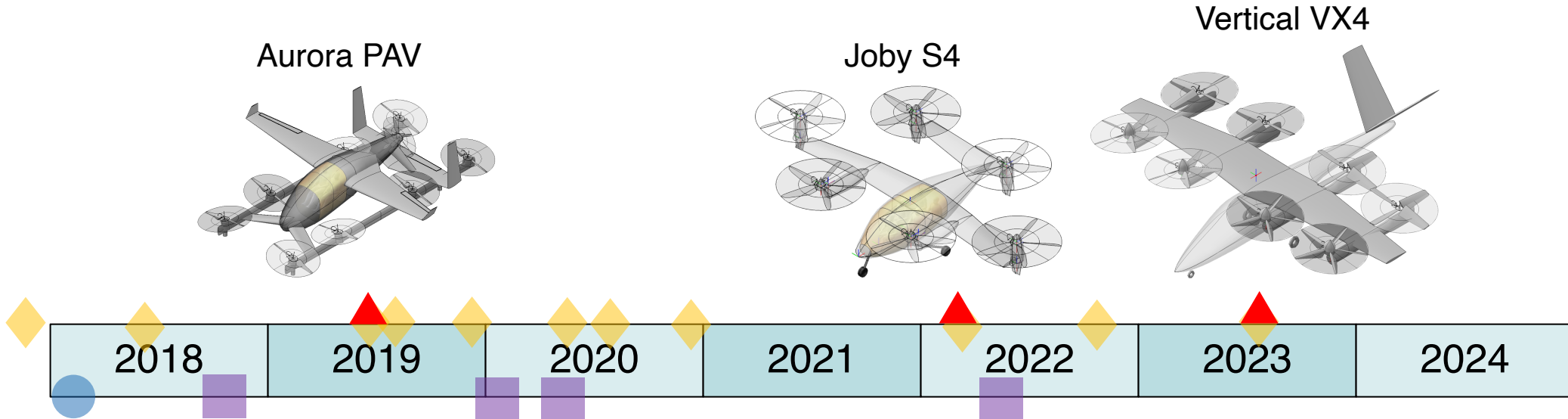
- Allowing operation into and at low altitudes above European urban areas
- No single point of failure in rotating parts
- $10^{-9}$  catastrophic failures per flight hour over populated areas

# (Just a few of the many) New companies and changed vehicles





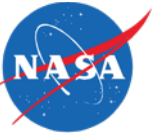
# Crashes in full-scale flight test



# Incremental updates and additions to NASA UAM Reference Vehicles



# Updated NASA UAM Reference vehicles



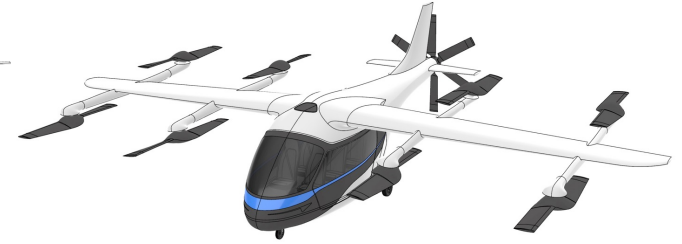
**1+1**



**2**



**4**



**8 + 1**

**Rotors**

Tilting

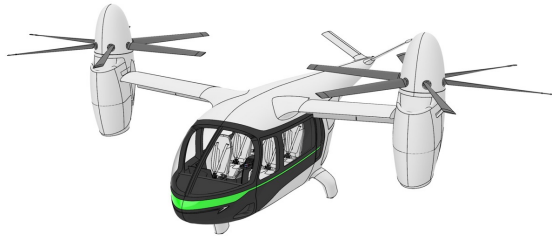
RPM?

Stopping?

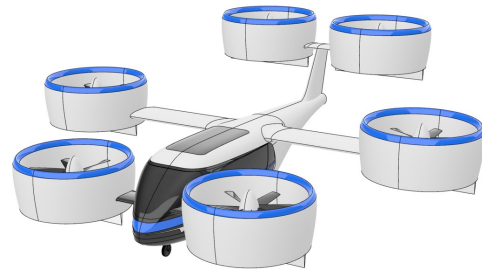
Variant

Yes: 8

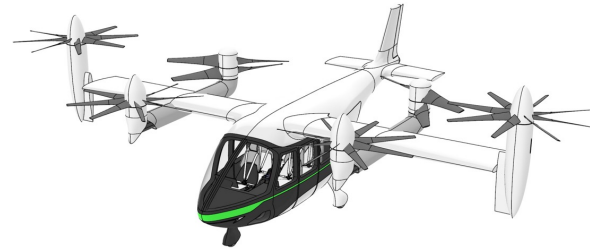
Yes: 8



**2**



**6**



**6**



**8**

**Rotors**

Tilting

RPM?

Stopping?

Yes: 4

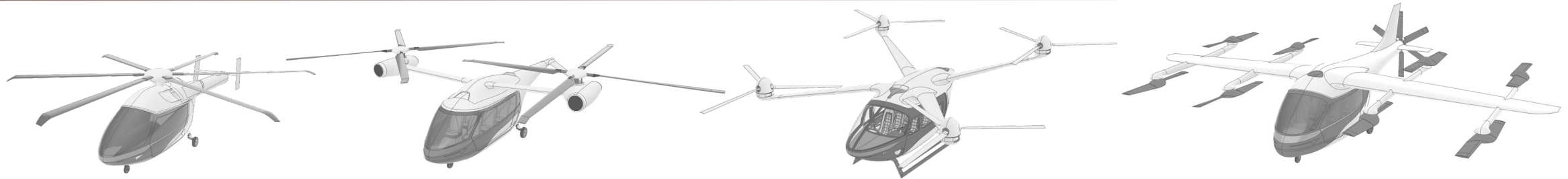
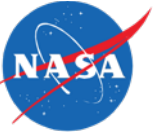
Variant: 2

Yes: 2

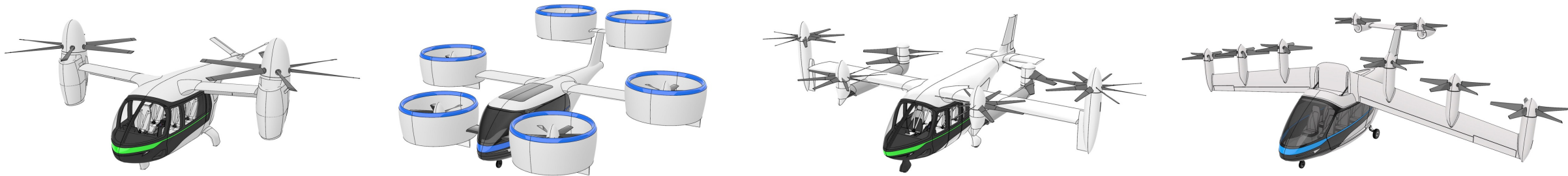
Yes



# Updated NASA UAM Reference vehicles



Rotors	1+1	2	4	8 + 1
Tilting				
RPM?			Variant	Yes: 8
Stopping?				Yes: 8



Rotors	2	6	6	8
<b>Tilting</b>	<b>Yes</b>	<b>Yes</b>	<b>Yes: 4</b>	<b>Yes</b>
RPM?			Variant: 2	
Stopping?			Yes: 2	

# Updated NASA UAM Reference vehicles



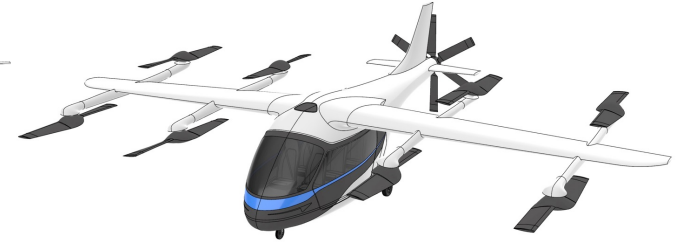
1+1



2



4



8 + 1

Rotors

Tilting

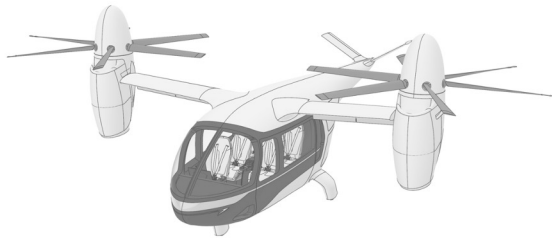
**RPM?**

Stopping?

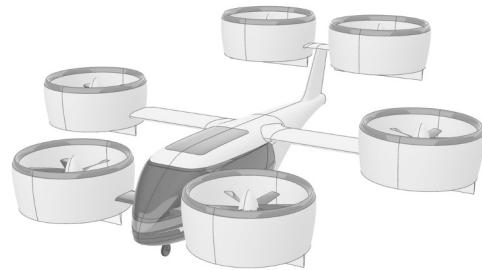
**Variant**

**Yes: 8**

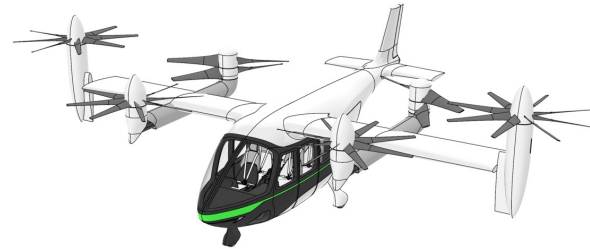
Yes: 8



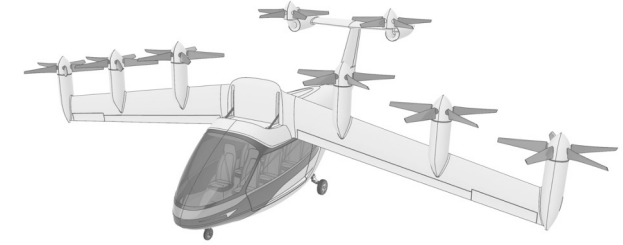
2



6



6



8

Rotors

Tilting

**RPM?**

Stopping?

Yes

Yes

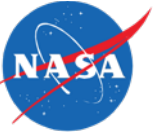
Yes: 4

**Variant: 2**

Yes

Yes: 2

# Updated NASA UAM Reference vehicles



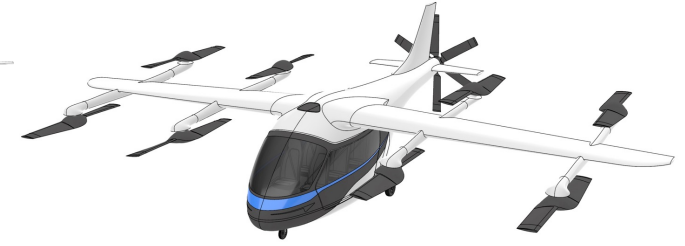
1+1



2



4



8 + 1

Rotors

Tilting

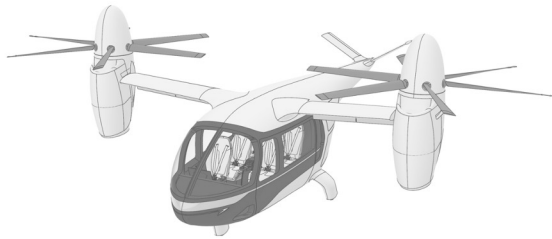
RPM?

**Stopping?**

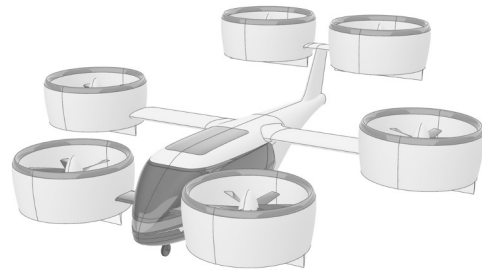
Variant

Yes: 8

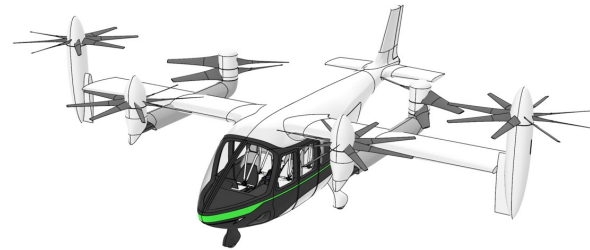
**Yes: 8**



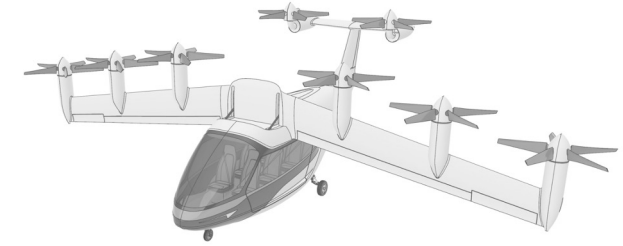
2



6



6



8

Rotors

Tilting

RPM?

**Stopping?**

Yes

Yes

Yes: 4

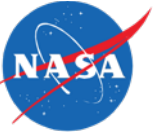
Yes

Variant: 2

**Yes: 2**



# Updated NASA UAM Reference vehicles



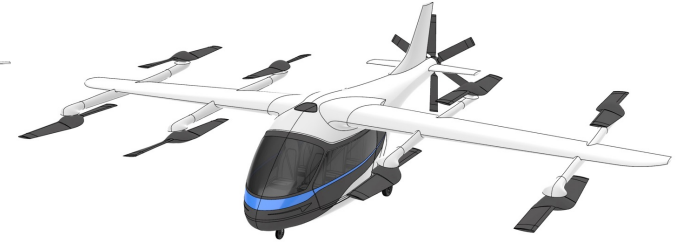
1+1



2



4



8 + 1

Rotors

Tilting

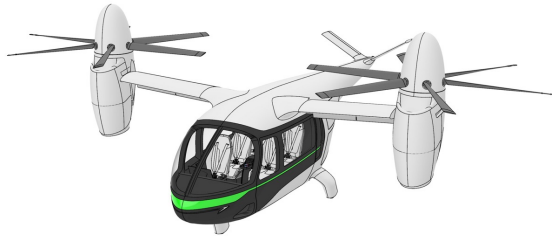
RPM?

Stopping?

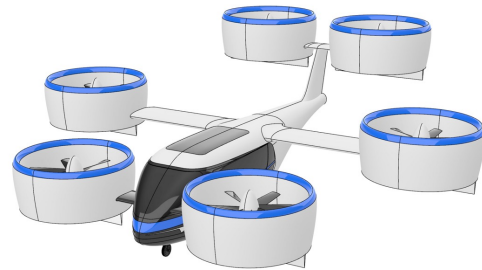
Variant

Yes: 8

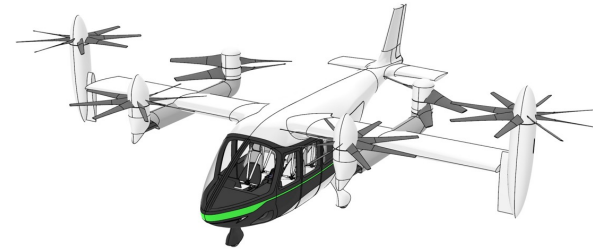
Yes: 8



2



6



6



8

Rotors

Tilting

RPM?

Stopping?

Yes

Yes

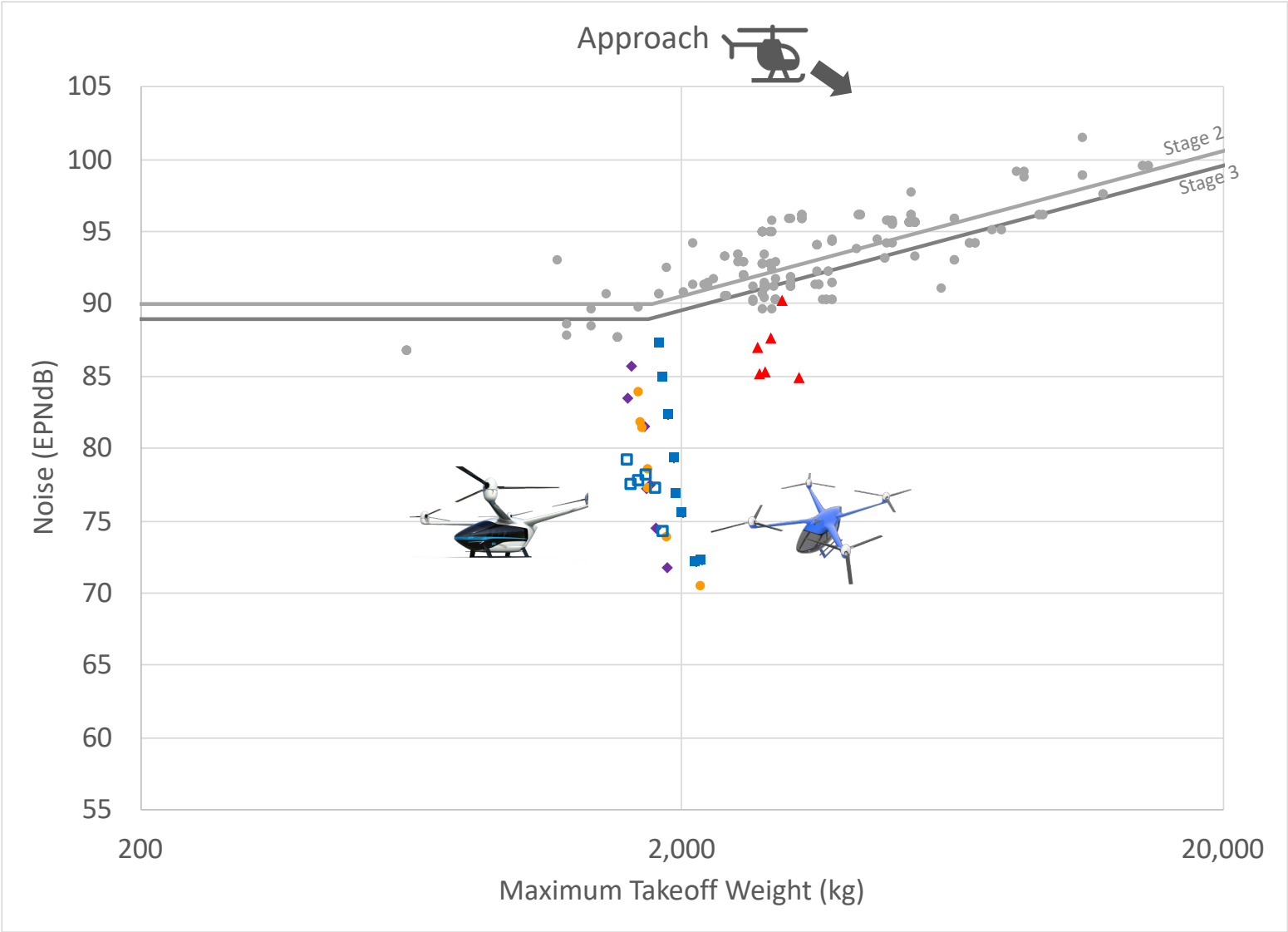
Yes: 4

Yes

Variant: 2

Yes: 2

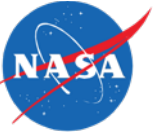
# Part 36H Certification Approach Noise Prediction



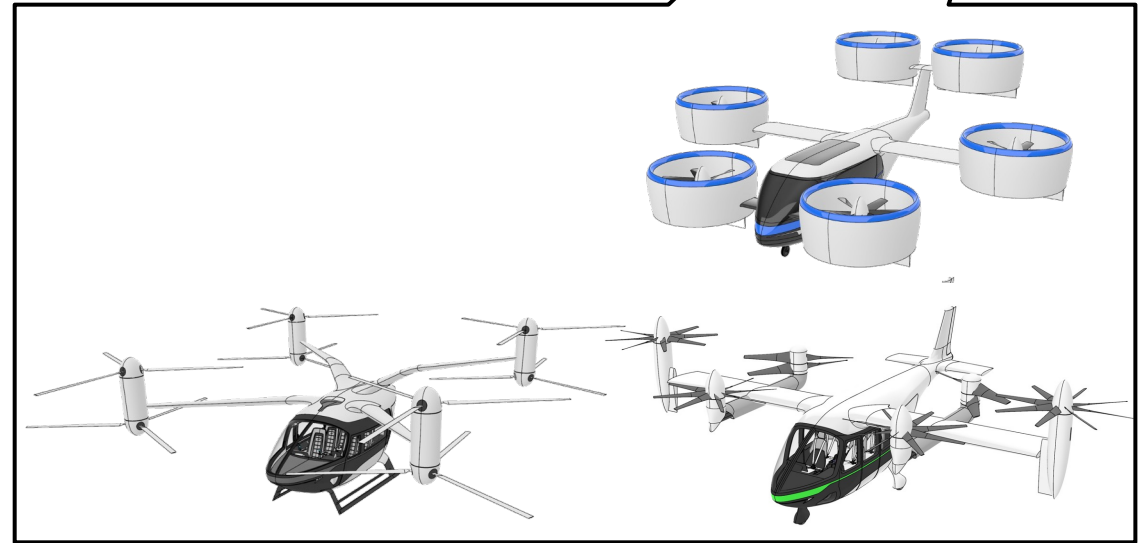
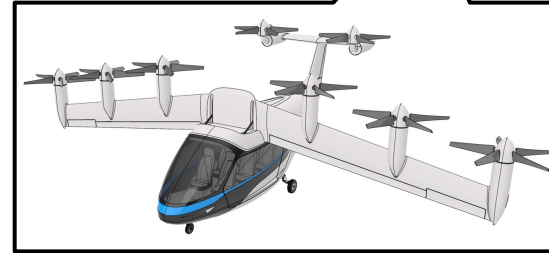
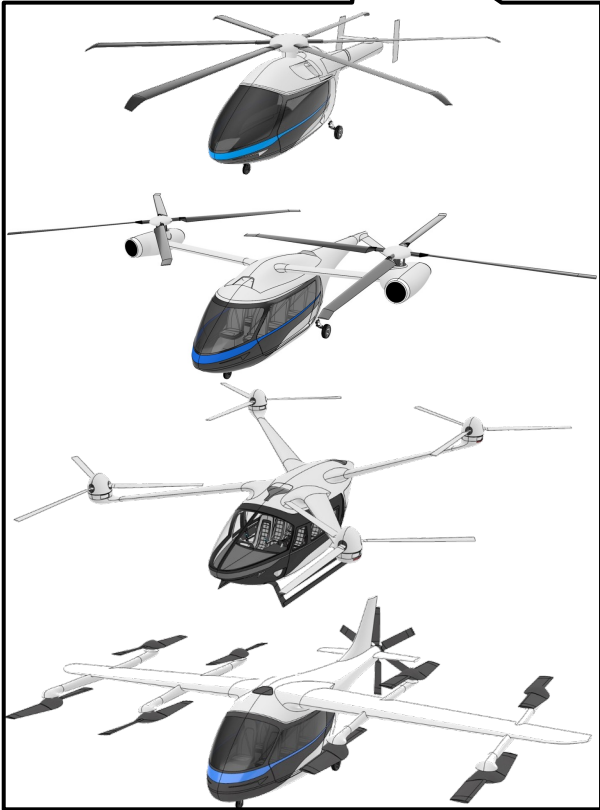




# When will updated UAM Reference Vehicle models be available?



2024				2025				TBD Future
Jan-Mar	Apr-Jun	Jul-Sep	Oct-Dec	Jan-Mar	Apr-Jun	Jul-Sep	Oct-Dec	





# How can you get the models?

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- **NDARC website – NDARC Users**
  - <https://rotorcraft.arc.nasa.gov/ndarc/>
    - Request NDARC: <https://software.nasa.gov/software/ARC-16265-1>
- **Langley Research Center - Systems Analysis and Concepts Directorate- Aeronautics Systems Analysis Branch (LaRC SACD ASAB) – Publicly available**
  - <https://sacd.larc.nasa.gov/uam-refs/>
- **NASA Technical Reports Server – Publicly available (reports only)**
  - <https://ntrs.nasa.gov>



May 2019



